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A NUMERICAL SIMULATION OF THE DISPERSAL
OF AERIAL SPRAYS

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NOMENCLATURE

A	Aircraft aspect ratio
b	Aircraft semispan, m
C	Concentration, volume/semispan ²
C _D	Drag coefficient, D/q _s
C _L	Lift coefficient, L/q _s
D _V	Diffusivity, cm ² /sec
F	Propeller thrust, N
Fr	Froude number, U/\sqrt{bg}
g	Gravitational constant, m/sec ²
h	Aircraft altitude, semispans
K	Inertia parameter, $\sigma \delta^2 U / 18 b \mu$
M _m	Mean molecular weight of gas mixture in transfer path
M _v	Mean molecular weight of evaporating material
m	Mass, Kg
P	Pressure, Kg/m ² (lb/in ²)
P _f	Partial pressure of air, N/m ²
P _v	Vapor pressure of air, N/m ²
PK1 to PK4	Propeller constants
Q	Nozzle flow rate, volume/semispan along flight path
Q _p	Propeller torque, N-m
r	Radial coordinate, in propeller radii
R	Droplet Reynolds number $\rho \delta U \bar{u} - \bar{n} / \mu$
R _U	Droplet free stream Reynolds number, $\rho \delta U / \mu$
RP	Propeller radius, m
S _c	Schmidt number, $\rho D_V / \mu$
T	Temperature, K(°R)
t	Time, secs
U	Free stream velocity, m/sec
u	Local flowfield velocity, dimensionless with U
V	Cummulative volume percent
V _{cw}	Crosswind velocity component, m/sec
x, y, z	Cartesian coordinates
x _o , y _o , z _o	Initial droplet position (nozzle location), semispans
σ	Droplet density, Kg/m ³
σ _s	Standard deviation, microns
Γ	Aircraft circulation strength, m ² /sec
δ	Droplet diameter, microns
δ _m	Volume median droplet diameter, microns
η	Droplet position, dimensionless with b
μ	Air absolute viscosity, Kg/m-sec
ρ	Air density, Kg/m ³
τ	Dimensionless time, Ut/b

Superscripts

•	Derivative with respect to τ
-	Vector quantity

Subscripts

g	Final value at ground plane
wb	Wet bulb condition
db	Dry bulb condition

I. INTRODUCTION

The use of aerospace technology to improve aerial applications promises to yield significant improvements in the efficiency of operation, the environmental acceptability, and many other areas. One area of study is the interaction of the spray droplet with the aircraft wake. This interaction improves the overall efficiency of the operation by increasing the swath width, but also results in the drift of particles suspended in the wake to off target sites. The propeller slipstream and other nonuniform portions the wake also lead to nonuniform deposits of material inside the swath. This research uses an analytical approach to study this wake-particle interaction. Ultimately the goal of such a research effort will be to tailor the aircraft wake and dispersal system to produce a wide uniform distribution of material with minimum drift.

Reed¹ in 1954 presented a two-dimensional analysis of aerial applications. This report, aided by the techniques for trajectory analysis developed in the aircraft icing program, established the fundamentals of the method. This work was hampered by the limited computational capabilities of the time and Reed presents only a limited number of trajectories for his simple model. Bragg² improved this model by performing a three dimensional analysis including propeller slipstream effects. Trayford and Welch³ have made similar improvements including the effect of crosswind on the distribution. The present method is based on reference 2 but includes many improvements which have been developed by the author since the publication of reference 2.

The goal of this research was to document a method, and develop a computer code to simulate the dispersal of liquid sprays from agricultural aircraft. The program was to require small computation times so mathematical models and numerical procedures were kept as simple as possible while still modelling the fundamental physical phenomena. The program was written in such a way that a user could modify the wake model easily, while still using the particle dynamics model and the distribution and drift numerical procedures. The user may provide a flowfield code which can be inserted into this program by use of the USERV subroutine. This allows for the analysis of more complicated flowfields which may contain fuselage vortices, vortices off of flaps due to variable span loading, winglets, or more complicated flowfields due to unconventional aircraft configurations

The computer code consists of three main parts; the droplet dynamics and evaporation models, the aircraft wake models, and the distribution and drift prediction method. The theoretical basis for these models is derived in Section II and the method is compared to other results in Section III. Section IV is the computer program users' manual containing a description of the code and the input, output, and error messages. In the Appendices are sample cases and a complete listing of the code.

II. THEORETICAL DEVELOPMENT

In this section the mathematical models used for the droplet dynamics and the wake flowfield will be described. The numerical procedures used to solve these equation will also be presented.

Droplet Dynamics

By application of Newton's Second Law of Motion to a water droplet the differential equation describing the particle trajectory may be derived. Here particles in the 100 to 500 micron size range are considered. These particles are nearly spherical in shape⁴ permitting the use of standard sphere drag data. In addition the analysis does not include the effect on the trajectory of atmospheric and aircraft induced turbulence, liquid droplet deformation and internal circulation, magnus forces, multiple particle interaction, and electric charge. For the aerial application problem these forces are considered negligible.

The equation of motion for a non-evaporating single particle moving in the wake of an aircraft can then be written as

$$m\left(\frac{d^2\bar{X}}{dt^2}\right) = \bar{D} + \bar{P} + \bar{M}_a + \bar{B} + m\bar{g} \quad (1)$$

The apparent mass, \bar{M}_a , the pressure gradient term, \bar{P} , and the Bassett force, \bar{B} , become important only if the density of the particle is of lower, or similar order of magnitude as that of air⁵. In addition, the apparent mass and Bassett force terms could also become significant if the particle experiences a large acceleration. The particle could experience a sizeable acceleration when initially injected into the flow. This acceleration

is, however, of short duration and in the streamwise direction so as to have little effect on the final spanwise location of the particle. Therefore, noting that the density of water is much greater than that of air, and ignoring the initial acceleration effects; the apparent mass, the pressure gradient, and the Basset force terms may be dropped from equation (1).

Now writing this equation in nondimensional form,

$$K\ddot{\bar{\eta}} = \frac{C_{DR}}{24}(\bar{u} - \dot{\bar{\eta}}) + \frac{1}{F_r^2} \frac{\bar{g}}{g} \quad (2)$$

where the nondimensional numbers K, the inertia parameter, and F_r , the Froude number are

$$K = \frac{\sigma \delta^2 U}{18 b \mu} \quad ; \quad F_r = \frac{U}{\sqrt{bg}} \quad (3) a, b$$

An additional nondimensional number, R_U , the free stream droplet Reynolds number

$$R_U = \frac{\rho \delta U}{\mu} \quad (4)$$

results from the $\frac{C_{DR}}{24}$ term in equation (2). The droplets studied here experience relatively low Reynolds numbers well within the range of zero to one thousand. Several curve fits to the standard sphere drag curve are available in this range and all yield similar results. Here the equation by Langmuir⁶

$$\frac{C_{DR}}{24} = 1. + 0.19R^{0.63} + 2.6 \times 10^{-4}R^{1.38} \quad (5)$$

is used to calculate the sphere drag when computing the particle trajectories. Note that here R is the Reynolds number experienced

by the droplet based on the particle velocity relative to the surrounding fluid

$$R = \frac{\rho \delta U |\vec{u} - \vec{\eta}|}{\mu} \quad (6)$$

The trajectory of a particle is then governed by equation (2) which is a non-linear, second order, ordinary differential equation which must in general be solved numerically. The equation may be reduced to first order and separated into its three components to form a system of six simultaneous differential equations. Given the six initial conditions of particle position and velocity when first injected into the aircraft flowfield, the trajectory may be calculated using a step integration method. The program uses a variable step size, predictor-corrector scheme suitable for stiff systems^{7,8}. When compared against non-stiff methods this scheme shows a considerable decrease in computational time to achieve the same accuracy. The trajectory is terminated when the particle intersects the ground plane or becomes entrained in a vortex.

Evaporation Model

For small liquid droplets under roughly 150 microns in diameter evaporation effects often become significant. As a result of evaporation the ground distribution of material is reduced and the material lost due to drift is greatly increased. To include the effect of evaporation, the analysis of Goering⁹ is used.

Since the mass of the particle is no longer constant the differential equation (1) must be modified. Having already

incorporated the assumptions described early it becomes

$$m\left(\frac{d^2\bar{x}}{dt^2}\right) + \left(\frac{d\bar{x}}{dt}\right) \frac{dm}{dt} = \bar{D} + m\bar{g} \quad (7)$$

Nondimensionalizing, this equation becomes

$$K\left(\frac{d^2\bar{\eta}}{d\tau^2}\right) = \left(\frac{C_{DR}}{24}\right)(\bar{u} - \dot{\bar{\eta}}) - \frac{3\dot{\bar{\eta}}K}{(\delta/b)}\left(\frac{d(\delta/b)}{d\tau}\right) + \frac{1}{Fr^2}\left(\frac{g}{g}\right) \quad (8)$$

Equation (8) is identical to equation (2) except for the addition of the second term on the right hand side which is due to the $\frac{dm}{dt}$ term in equation (7). An additional variable, δ , the particle diameter is introduced in equation (8) and another equation must be added to the system to allow a solution.

The expression for the rate of change of the particle diameter with respect to time is⁹

$$\left(\frac{d(\delta/b)}{d\tau}\right) = -\left(\frac{2}{U}\right)\left(\frac{M_V}{M_m}\right)\left(\frac{D_V}{\delta}\right)\left(\frac{\rho}{\sigma}\right)\left(\frac{\Delta P}{P_f}\right)(2 + 0.6 Sc^{1/3} Re^{1/2}) \quad (9)$$

The terms on the right hand side may be evaluated in the following way.

It is assumed that the water droplet is diffusing into the air, therefore, the molecular weights are $M_m = 29.0$ and $M_V = 18.0$. The diffusion coefficient for water vapor into air is given as a function of temperature⁹

$$D_V = 5.28 \times 10^{-6} T_K^{1.88} \quad (10)$$

The term $\Delta P/P_f$ is shown by Goering⁹ to be

$$\frac{\Delta P}{P_f} = \frac{P_{swb} - P_V}{P_{atm} - P_V} \quad (11)$$

Brooker¹⁰ has developed a mathematical model for the psychrometric chart which is useful in evaluating equation (11). Input to the computer code are the atmospheric pressure, P_{atm} , and the wet and dry bulb temperatures, T_{wb} and T_{db} . The saturation vapor pressure, P_{swb} , at wet bulb temperature, T_{wb} , is given by

$$P_{swb} = e^{54.6329 - \frac{12301.688}{T_{wb}}} - 5.16923 \ln(T_{wb}) \quad (12)$$

and the vapor pressure, P_v , is

$$P_v = P_{swb} + \frac{0.2405(P_{swb} - P_{atm})}{0.62194(1075.8965 - 0.56983(T_{wb} - 491.69))}(T_{db} - T_{wb}) \quad (13)$$

In equation (12) and (13) the temperatures are in °R and the pressures are in psi. Equation (12) is valid only for temperatures above 32°F. Note that by definition $RH = P_v/P_s$, if the relative humidity, RH, is available when T_{wb} is not.

The particle trajectory including the evaporation effect is calculated in the same manner as before, here using the differential equation (8) and adding to the system equation (9). Additional complications due to droplet evaporation which arise in the distribution and drift calculations will be discussed later.

Wake Flowfield

The flowfield model incorporated in the computer code is a horseshoe vortex system in ground effect. This simple model in either its 2-D or 3-D forms provides the wake velocities with a minimum of computer time, allowing for the small execution times characteristic of this program. In addition, a simple propeller, crosswind, and tunnel flowfield models are included in the code.

If these simple models are inadequate, the user may supply an alternate flowfield model by providing the proper subroutine. Here the flowfield models included in the code are discussed.

Basic Wake Model: The wake of an aircraft in ground effect may be modelled in three dimensions using a horse shoe vortex system. This model is reasonably accurate provided that, the wake roll-up occurs in a time much less than that of the particle trajectory, the particle trajectory intersects the ground before this system decays appreciably, and the particle remain outside of the viscous vortex cores and in the potential flow region. A two dimensional model can be formed consisting of only two doubly infinite straight trailing vortices and their reflection through the ground plane, figure 1. The strength of these vortices is calculated assuming an elliptical loading

$$\Gamma = \frac{4C_L U b}{\pi A} \quad (14)$$

The motion of these vortices is given by Bragg² and the velocity can be calculated using the Biot-Savart law¹. This model can be extended to three dimensions by adding a bound vortex of strength Γ on the aircraft quarter chord line, figure 2. The Biot-Savart Law is used to calculate the velocity induced in the x direction² and the y and z velocities are the same as in the two dimensional model.

It should be noted that the two dimensional model uses Reed's¹ correction for the bound vortex. If the user inputs zero for the initial droplet velocity, the program assigns to the droplet an initial velocity equal to the velocity the particle would have if

falling in free air in a constant flowfield with the velocity components equal to that existing in the 2-D model at the point of particle injection. This correction is ignored if instead non-zero initial droplet velocities are input to the computer program.

Tunnel Model. The two dimensional model of reference 2 has been included in this program. This permits the calculation of droplet trajectories from an aircraft model in a wind tunnel to evaluate the influence of the tunnel walls and the floor. The tunnel ceiling is assumed to have a negligible effect on the droplet trajectories since a model in ground effect will be in close proximity to the tunnel floor. The simple geometry of this model is shown in figure 3. A conformal mapping method is used to determine the velocity at a point in the tunnel due to the two dimensional vortex system. The complex function describing the velocity potential and stream function in free air is transformed to the plane with infinite reflections of this vortex system in the y direction by using a complex sine mapping function. The vortex position is determined by step integration using a truncated series to express the induced velocities on a vortex filament. The derivative of this complex function with respect to the complex particle position, yields the y and z velocities at any point in the tunnel.

Propeller Model: A propeller slipstream is characterized by a rotation in the y-z plane due to the torque and an increased axial velocity due to the generated thrust. Functional forms for the axial and rotational velocities as a function of radial

distance from the propeller hub have been assumed. These equations are of a form compatible with existing data on the thrust and torque loadings on general aviation propeller blades. The equation for the axial velocity is

$$V_{axial} = PK1(1. - r) + PK2(r) \sin(\pi r) \quad (15)$$

and the rotational velocity is given by

$$V_{rot} = PK3(1. - r) + PK4(r) \sin(\pi r) \quad (16)$$

Figure 4 shows the velocity distributions for a typical propeller model. PK1 and PK3 govern the velocities at the centerline while PK2 and PK4 control primarily the maximum velocities. For most applications PK3 = 0 and PK1 is set equal to some small fraction of PK2. Using equations (15) and (16) the slipstream can be integrated to determine the thrust, F, and the torque, Q_p . The thrust is

$$\begin{aligned} F = (RP)^2 \pi \rho \Big[& 2U(0.166667PK1 + 0.189304PK2) \\ & + 0.0833333(PK1)^2 + 0.129006(PK1)(PK2) \\ & + 0.0870040(PK2)^2 \Big] \end{aligned} \quad (17)$$

and for the torque,

$$\begin{aligned} Q_p = 2\pi \rho (RP)^3 \Big[& 0.033333(PK1)(PK3) + 0.036657 [(PK1)(PK4) \\ & + (PK2)(PK3)] + 0.0507039(PK2)(PK4) + 0.083333U(PK3) \\ & + 0.124801U(PK4) \Big] \end{aligned} \quad (18)$$

Equations (17) and (18) are then used to determine the constants PK1, PK2, PK3, and PK4 knowing the propeller thrust and torque. This method assumes that the propeller slipstream extends downstream from the propeller disc plane to infinity with no dissipation.

In reality the rotational velocities are noticeably reduced due to viscous dissipation and the straightening effects of the main wing, fuselage, landing gear, horizontal, and vertical stabilizers. This reduction in rotational velocity is strongly dependent on the particular aircraft being modelled. It is recommended therefore that V_{rot} , that is PK3 and Pk4, be reduced from those calculated from equation (16). Although little experimental data is available on which to base this, a reduction to 75 percent of the calculated values seems reasonable from preliminary studies of conventional low wing monoplanes.

Crosswind Model: The component of the wind perpendicular to the flight path and parallel to the ground plane is modelled by the equation¹¹

$$V_{cw} = VCW(z)^{0.25} \quad (19)$$

Here VCW is the crosswind component in m/sec 1 meter above the ground and z is the height above the ground plane in meters. The crosswind not only contributes a velocity in the y direction, but also causes a lateral translation of the entire aircraft wake system. This is modelled in the program by including the first order differential equations for the positions of the trailing vortices and propeller slipstream with the system of equations for the droplet trajectory and solving this enlarged system by the same step integration technique.

Distribution and Drift Model

By combining several droplet trajectories from a single nozzle location the ground deposition from the nozzle can be calculated

and the drift estimated. This procedure is straight forward for the case of no crosswind, evaporation, or propeller and was first developed for the aerial spray problem by Reed¹. Here an improved form of Reed's analysis is used for the simple case and major changes are made to handle the more difficult case of the unsymmetric and often double valued distributions.

The concentration, C, from a single nozzle can be written as

$$C = Q \frac{dV}{d\delta} \frac{d\delta}{dy_g} \quad (20)$$

By superposition the concentration from any number of nozzles can be found. The derivative $d\delta/dy_g$ can be found from the plot of the particle diameter as a function of the final lateral position where the particle intersects the ground plane. These points are cubic splined to obtain $d\delta/dy_g$. This value must then be multiplied by $dV/d\delta$, a function of δ and the particular nozzle used, which represents the volume of material released by the nozzle as a function of particle diameter. Here a normal distribution of material about some volume median diameter, δ_m , is assumed which gives

$$\frac{dV}{d\delta} = \frac{1}{\sqrt{2\pi} \sigma_s} e^{-\frac{1}{2} \left(\frac{\delta - \delta_m}{\sigma_s} \right)^2} \quad (21)$$

where σ_s is the standard deviation. The Q in equation (20) is the volume flow rate of the nozzle. The concentration of material on the ground is then given by equation (20) and all that remains is to determine the drift.

The drift is estimated by an extrapolation procedure which finds the smallest particle, within some maximum allowable error, which reaches the ground. The amount of material in the distribution from particles smaller than this drifted particle is found by integration of the normal distribution function. The allowable error is a function of the number of standard deviations the particle size is from δ_m to control the error in terms of volume of material. Particles which encircle the wing-tip vortices are considered as drift and their trajectory calculations are terminated. In addition, off target distribution is also considered drift. The user can select a lateral semispan distance beyond which the material is considered to be drift.

This procedure must be modified for unsymmetrical distributions or cases including droplet evaporation. As a result of the propeller slipstream effects or the crosswind model the calculation of $d\delta/dy_g$ can become more difficult. Here it is not uncommon for several particle sizes to intersect the ground at the same y_g location, thus making the function double valued. The double valued and often erratic behavior of the curve makes the use of the cubic spline very difficult. Many points would be required to obtain a reasonable fit. When these complications arise the program abandons the cubic spline and instead uses linear interpolation between the calculated trajectory points. In regions of large changes in slope the program automatically inserts additional particle trajectories using again an error parameter weighted from the volume median diameter. This method provides good results in a minimum of computer time. Here the drift is estimated in the same manner as before.

When the droplet evaporation is considered the mass lost through evaporation must be accounted for. This results in an increase in material lost due to drift and a decrease in the ground deposition. The final droplet diameter, δ_f , is determined as a function of the initial diameter, δ_o , by using a cubic spline on the results of the trajectory calculations. The distribution is then reduced by the ratio of δ_f^3/δ_o^3 . In addition to the material which does not reach the ground or lies outside the target area, the amount of material which evaporates is determined by integration and included in the total drift. By using this procedure accurate drift and distributions are determined from the calculated particle trajectories.

III. CODE VALIDATION

Initial validation of the computer code was conducted by comparing the results of this method to the early analytical work of Reed¹. Reed's method uses a two dimensional wake model without propeller, evaporation, or crosswind effects. The distribution is calculated using a least-squares fit of a second order polynomial to the trajectory data. A comparison of the two methods is shown in figure 5. Here the circulation, Γ , has been matched since the early work assumed a rectangular lift distribution. The comparison is quite good considering that modern digital computers were not available for Reed to make his trajectory calculations.

Comparisons of the code to experimental data is difficult because of the number of variables involved which are difficult to control or measure. Recent small scale model tests of agricultural aircraft dispensing scaled particles have generated some distribution data which may be used for comparison^{12,13}. Figure 6 shows the measured distributions from three nozzles and the corresponding calculated distributions. The model used was unpowered with a wing semispan of 0.914m moving at 20.6 m/sec at an altitude of 0.51 semispans. The particles used were glass microbeads with a mean diameter of 125 microns and a density of 2.42 g/m³. Since the test program was designed only to validate the scaling laws, no measurements of particle size distribution or nozzle volumetric flow rate was made. These parameters were estimated for this comparison. Note also that the concentration reported in reference 12 is in terms of the numbers of particles per unit area where this

method calculates concentration in terms of volume of material per unit area. This accounts for the difference in the areas under the curve and also for the larger measured concentrations to the right of the peak concentration where the particles are smaller in diameter. The difference in the distributions at the $y_0 = 0.30$ nozzle location is due in part to the fuselage vortex not accounted for in the present flowfield model. The comparison in figure 5 is however, quite good overall and supports the analysis.

In figure 7, the present method is again compared to data from reference 12. Here the same model is used as in figure 5 but the altitude is 0.35 semispans. The lateral position of the nozzle is at 0.40 semispans and the particles have a mean diameter of 140 microns and a density of 0.58 g/m^3 . These particles scaled to a 200 micron water droplet for the full scale aircraft. Here again the comparison is quite good considering the differences in the definition of concentration. The amount of lateral transport of the material and the general shape and magnitude of the concentration curves compare quite well.

The present method was also compared qualitatively to the powered test conducted in reference 12. Again considering the influence of the fuselage vortex the comparisons were encouraging. The amount of lateral transport due to the propeller was approximately the same if the calculated swirl velocity was reduced by 25% as discussed earlier to account for straightening effects. Sufficient data were not available to make a detailed comparison of the distributions.

Detailed comparisons of this method have been conducted with both analytical and experimental data. The method has

performed well over the range of parameters tested. The code appears to do a good job of predicting the lateral transport of droplets in the wake of aircraft in ground effect.

IV. COMPUTER PROGRAM MANUAL

A computer program has been developed to predict the ground deposition resulting from the dispersal of water droplets into the wake of an aircraft in ground effect. A complete listing of this program has been provided in Appendix B. After a general description of the program, the input, output, error messages, and test cases will be presented in this section.

General Description

The computer code has been written in FORTRAN and developed on the AARL Harris Computer System. The code is in single precision and should execute on CDC systems with only minor changes. Conversion of the code to execute on IBM type machines should involve little more than a conversion to double precision and modifying the Hollerith codes. The code is approximately 2500 card images in length and requires 15 to 45 seconds of execution time per single nozzle deposition on the Harris SLASH 6 Computer. The Harris /6 is slightly slower than a CDC Cyber 73 and about 6 times slower than an IBM 370/168. Additional storage must be supplied to the program on octal file 10.

A plot package has been provided which is completely contained in subroutine PLOT1. A single call to this subroutine occurs as the last executable statement in the main program. Since the plot subroutine used here may not be compatible to those available on other systems, the program was written to allow for easy modification of the plot package.

Certain limitations on the size of the analysis that can be conducted considering the present DO loop parameters and array

dimensions should be discussed. The maximum swath width that can be used is ± 10 semispans and the number of nozzles input can not exceed 100. The program is limited to 15 particle trajectories per nozzle including those determined internally to estimate the drift and improve accuracy in a region of a large change in lateral transport. Therefore no more than 10 particle diameters should ever be input and in general 5 to 7 is usually sufficient. No more than 1000 steps can be taken in the calculation of a single particle trajectory. The number of particles calculated and the number of steps can be controlled by use of the user input error parameters.

A flow chart including the major subroutines is shown in figure 8.

Input Data

The program reads the input from device 5 in the manner described below. Several options are available to the user with the acceptable combination of options shown in figure 9. In addition the user may supply the wake flowfield to be used by the inclusion of a subroutine USERV. This allows the present particle dynamics, including evaporation effects distribution, and drift models to be used with a different or more sophisticated wake model. USERV is described by comment statements in the listing. It may use by inclusion of the proper COMMON statements or subroutine calls use other variables and sections of the code as desired.

The inputs to the program are primarily physical variables in SI units. Some variables have been nondimensionalized where it

seemed the most logical approach. It is hoped that this type of input will make the code more useful. It should be noted however, that the problem can be nondimensionalized as in reference 11. The number of independent parameters is much less than the number of physical variables input to this code. The inputs could have been in terms of completely nondimensional numbers such as $R\bar{U}$, K , and F_r as discussed earlier. The use of this code for sensitivity analysis can, therefore, be greatly simplified by considering the nondimensional parameters.

Card 1: (Format 13A6) Title

Card 2: (Format I5)
 NDROPS Number of trajectories to be run.
 Set = 1 if NDIST = 1

Cards 3-8 repeated NDROPS times

Card 3: (Format 9I2, 2X, E15.6)

N2D	= 0	For a 3D run
	1	For a 2D run
	2	For a 2D run using the user supplied USERV subroutine
N3D	= 0	For a 2D run
	1	For a 3D run
	2	For a 3D run using the user supplied USERV subroutine
NEVAP	= 0	No droplet evaporation
	1	Droplet evaporation included
NPROP2	= 0	No propeller effects
	1	Propeller slipstream model included
NCW	= 0	No crosswind effects (set VCW = 0)
	1	Crosswind effects included
NTUN	= 0	No tunnel model
	1	Tunnel model included
NDIST	= 0	Single particle trajectory run
	1	Ground depositions calculated

NPRINT = 0 Short output option
 1 Long output option

 NPLOT = 0 No plotting
 1 Plot distribution of trajectories

 EPS Step integration error control parameter,
 usually set equal to 1.E-5.

Card 4 included only if NEVAP = 1

Card 4: (Format 3F10.6)

PA Atomspheric pressure, N/m^2
 TDB Dry bulb temperature, $^{\circ}\text{C}$
 TWB Wet bulb temperature, $^{\circ}\text{C}$

Card 5 included only if NPROP2=1

Card 5: (Format 6F10.4)

ZPROP2 Z coordinate of propeller hub, semispans
 PK1 to 4 Propeller slipstream constants as defined
 in Section II
 RP Slipstream radius (usually equal to pro-
 peller radius), semispans

Card 6 (Format 7F10.4)

A Aircraft aspect ratio
 CL Aircraft lift coefficient
 B Aircraft semispan, m
 U Aircraft speed, m/sec
 G Gravitational constant, m/sec^2
 ZY0 Initial Y coordinate of trailing vortex,
 semispans
 ZZ0 Initial Z coordinate of trailing vortex,
 semispans

Card 7: (Format 2E20.6, 2F10.4)

DA	Air Density (ignored if NEVAP = 1), Kg/m^3
VIS	Absolute air viscosity (ignored if NEVAP = 1), Kg/m-sec
VCW	Crosswind velocity (=0. if NCW = 0), m/sec
D	Tunnel width (ignored if NTUN = 0), m

Card 8 included only if NDIST = 0

Card 8: (Format 8F10.5)

DIA	Particle diameter, microns
DD	Particle density, Kg/m^3
X, Y, Z	Initial particle position, semispans
UD, VD, WD	Initial particle velocity relative to the aircraft fixed coordinate system, nondimensional with respect to U

Cards 9 to 16 included if NDIST = 1

Card 9: (Format 6I5)

NCOL	Number of nozzles
NROW	Number of initial particle sizes input
NDIAMN =	0 All nozzles the same DIAMN 1 DIAMN input for each nozzle
NSTDEV =	0 All nozzle have the same STDEV 1 STDEV input for each nozzle
NQ =	0 All nozzles have the same Q 1 Q input for each nozzle
NZNOZ =	0 All nozzles have same Z coordinates 1 Z input for each nozzle

Card 10: (Format 8F10 5)

DD	Particle density, Kg/m^3
XO	Nozzle X coordinate (ignored if N2D = 1), semispans
UDO, VDO, WDO	Initial particle velocity relative to the aircraft fixed coordinate system, nondimensional with respect to U
SWIDTH	One-half the swadth width calculated, semispans

DERR Constant by which default error limits
 are multiplied, normally set equal to one

DWIDTH Material landing beyond \pm this value are
 considered drift, semispans

Card 11: (Format 8F10.5)

YNOZ Y coordinate of each nozzle (NCOL values),
 semispans

Card 12. (Format 8F10.5)

SIZE Particle sizes input to represent distri-
 bution (NROW values), microns

Card 13: (Format 8F10.5)

DIAMN Median particle diameter, microns
 Input one value if NDIAMN = 0
 Input NCOL values if NDIAMN = 1

Card 14: (Format 8F10 5)

STDEV Standard deviation of particle size
 distribution, microns
 Input one value if NSTDEV = 0
 Input NCOL values if NSTDEV = 1

Card 15. (Format 8F10.5)

Q Nozzle flow rate, unit volume/semispan
 along flight path
 Input one value if NQ = 0
 Input NCOL values if NQ = 1

Card 16 (Format 8F10 5)

ZNOZ Z coordinate of nozzle, semispan
 Input one value if NZNOZ = 0
 Input NCOL values if NZNOZ = 1

Cards 1-16 may be repeated to run several cases simultaneously.

Output

The program first outputs all the basic input data. The variables output are all in the same units as the corresponding inputs. After this initial listing, the output depends primarily on only the choice of NDIST and NPRINT. The program's calculated output is all dimensionless unless indicated otherwise.

For the $\text{NDIST} = 1$ case and $\text{NPRINT} = 1$, the distribution from each individual nozzle is listed. The output is self explanatory, note that concentration has units of volume per semispan squared. If more than one particle size lands at the same lateral position, YG, the diameter, DIA, column is omitted. The rest of the output is the same for both the $\text{NPRINT} = 0$ and 1 options. A table including the input values for each nozzle is output along with the drift in percent of total for each nozzle. Next a trajectory summary listing all the trajectories calculated is included before the final distribution. The final distribution is a compilation of the distributions from all nozzles and is output in 0.01 semi-span increments. The total drift given is in percent of total material and is weighted by the volume flow rate of each nozzle. As a check, the total material reaching the target area is also given. This is determined by a simple trapezoidal integration of the final integration and is meant only as a check. The accuracy of this estimate will be low for complicated distributions.

For the single particle trajectory calculations of $\text{NDIST} = 0$ the output is simple for $\text{NPRINT} = 0$. Here the initial inputs are output as before, followed by the final value of the variables in their normal units. The final values include particle position, velocity, droplet diameter for $\text{NEVAP} = 1$, and the nondimensional time, τ , required for the trajectory. This output is repeated for each particle trajectory.

If the user selects $\text{NPRINT} = 1$, all the dependent variables and the independent variable, τ , are output at each time step in the integration process. This can result in a great many lines

in some cases. Since so many combinations of variables are possible depending of the user options selected, no header is provided for this output. The variables are output in the following manner. Column 1 is always the step number and 2 the nondimensional time, τ . Columns 3-6 always contain, the lateral position, Y, the lateral velocity, VD, the vertical position, Z, and the vertical velocity, WD. If the run is 3-D, columns 7 and 8 contain the streamwise position, X, and the corresponding velocity, UD. If NEVAP = 1 the last column always contains the current droplet diameter. The columns between WD or UD and the droplet diameter contain the vortex and propeller slipstream position. If options are selected making the entire flowfield transport laterally, the columns will be Y and Z coordinate of the left, then right trailing vortex and then the Y and Z position of the propeller slipstream. If all these variables are not present, the initial values make it clear which one is being output.

Error Messages

The program provides several internal error messages and warnings. These messages and the most likely user actions needed to correct the problem are given here

MESSAGE	COMMENT
Max Number of Runs Exceeded Before Drift Calculation Completed (E)	Number of Particle Trajectories Exceeds 15, Reduce NROW or Increase DERR
Trajectory Terminated, Propeller Slipstream Entrainment (W)	Particle Considered Drift

MESSAGE	COMMENT
DO Loop Parameter Exceeded in SUB2D (E)	Number of Steps Exceeds 1000, Increase EPS
Trajectory Completed, Z less Than Zero	Normal Trajectory Termination
Trajectory Terminated, Vortex Entrainment (W)	Particle Considered Drift
DO Loop Parameter Exceeded in SUB3D(E)	Number of Steps Exceeds 1000, Reduce EPS
NVAR Incorrect(E)	Illegal Combination of Options
Cubic Spline Problem Nozzle No DIA = (E)	Cubic Spline Provides Poor Fit to Trajectory Data, Change Error Parameters or Input Particle Diameters
Insufficient Data To Calculate Distribution (E)	Not Enough Trajectories to Cubic Spline Particle Diameters for NEVAP = 1 Case, Change Input Particle Diameter
Insufficient Data To Estimate Drift, Nozzle No. Drift Set Equal To Zero (E)	(Same as Above)
WDR Exceeded DO Loop in INCON (E)	Error in 2D Initial Droplet Velocity Calculation, Input Non Zero Initial Velocity, Often Due to Illegal Input
Right Endpoint in INCON Incorrect (E)	(Same as Above)
DO Loop Parameter in INCON Exceeded (E)	(Same as Above)

(E) ERROR (W) WARNING

Sample Cases

Five sample cases have been provided in Appendix A. These cases are intended both as examples and as check cases after the program has been installed on a new system. The complete input, output, and plots are given for each of the cases. With the use of the input and output descriptions in this section, the cases should be self explanatory.

V SUMMARY

A method has been developed and a computer program written to predict the dispersal of aerial sprays from agricultural aircraft. The method has been compared to early analytical results and recent experimental studies and found to compare well with these results. A user's manual for the program including sample cases and FORTRAN program listing have been provided. The code was written to allow the flexibility to install a more sophisticated wake model to increase the usefulness of the program. The computer program provides a useful tool to aid in agricultural aircraft and dispersal system design as well as for aerial applications research.

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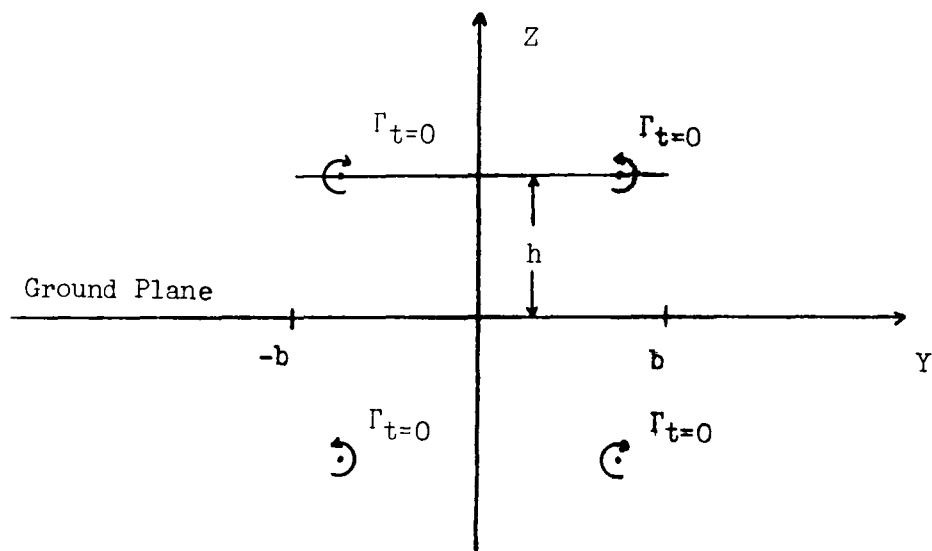


Figure 1. Two Dimensional Wake Vortex Model

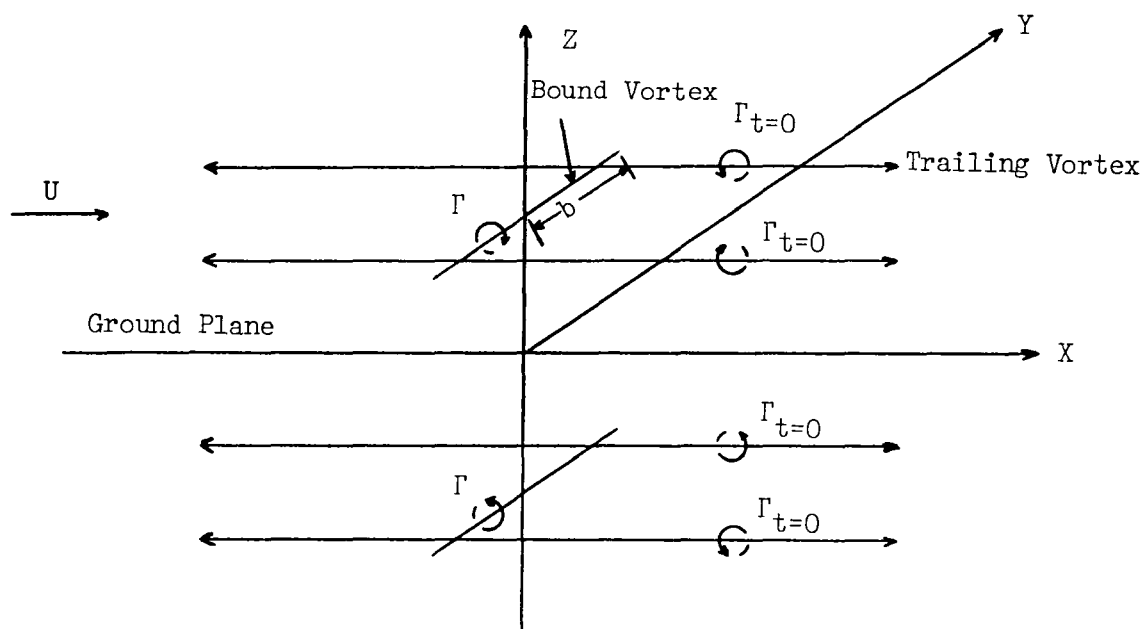


Figure 2. Three Dimensional Wake Vortex Model

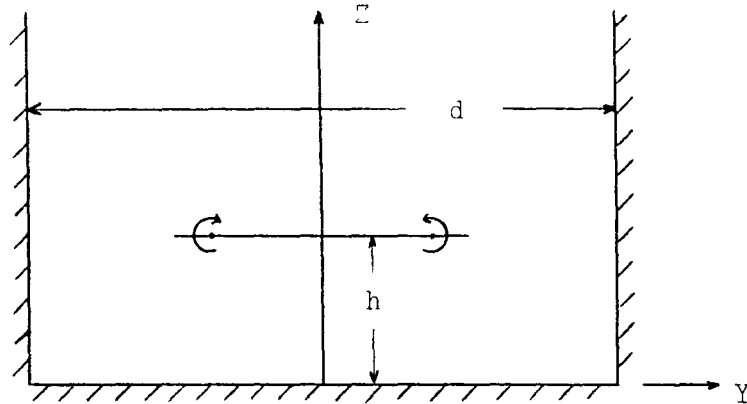


Figure 3 Two Dimensional Tunnel Model

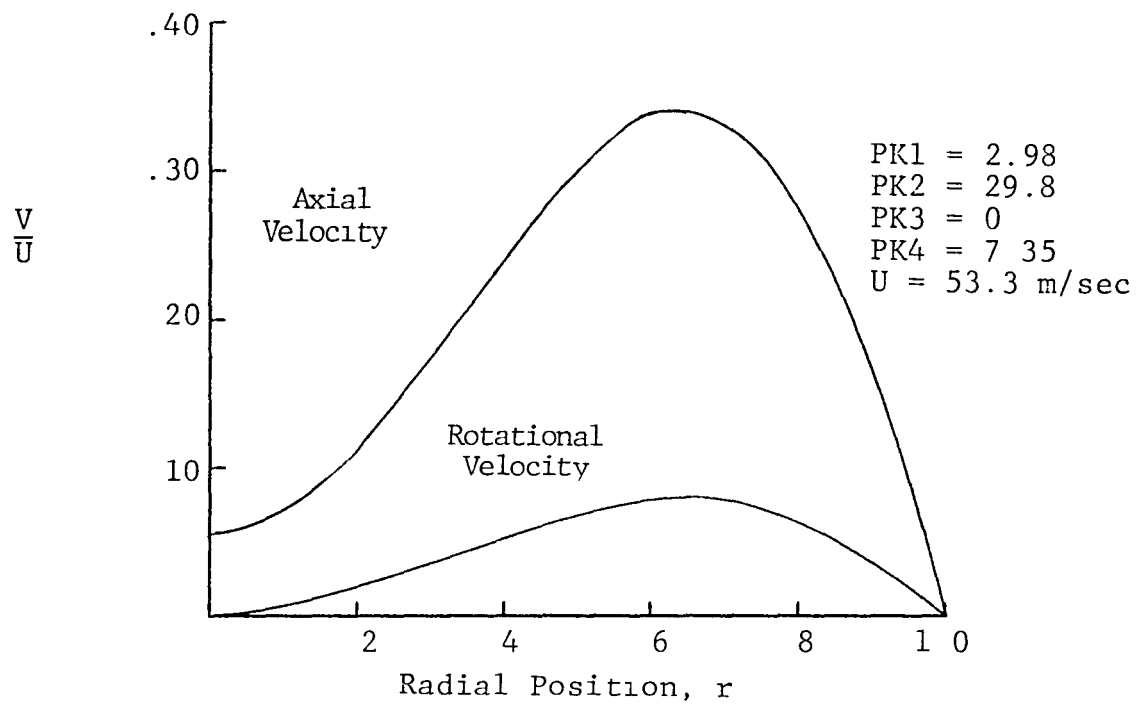


Figure 4. Typical Velocities for the Propeller Model

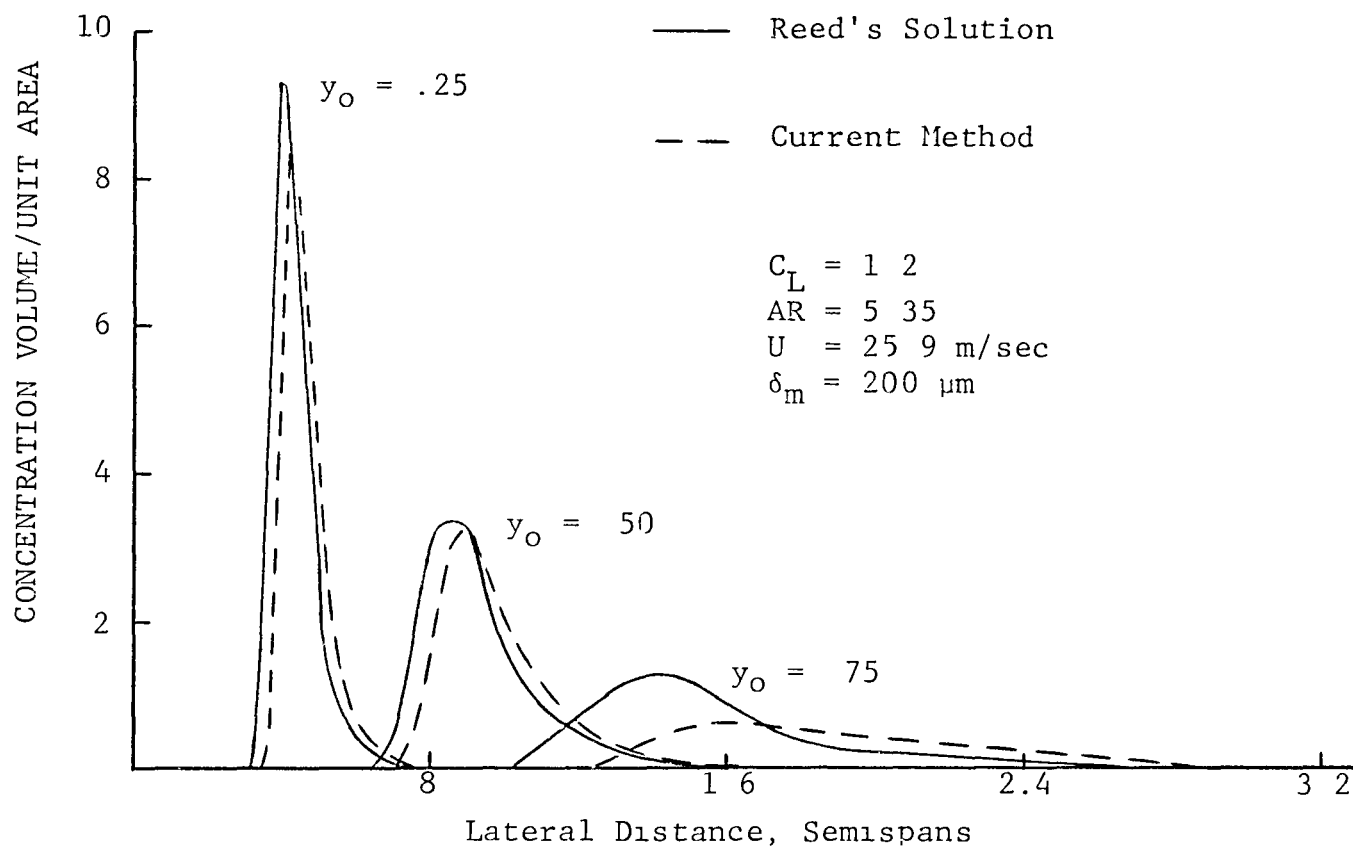


Figure 5. Present Method Compared to that of Reed

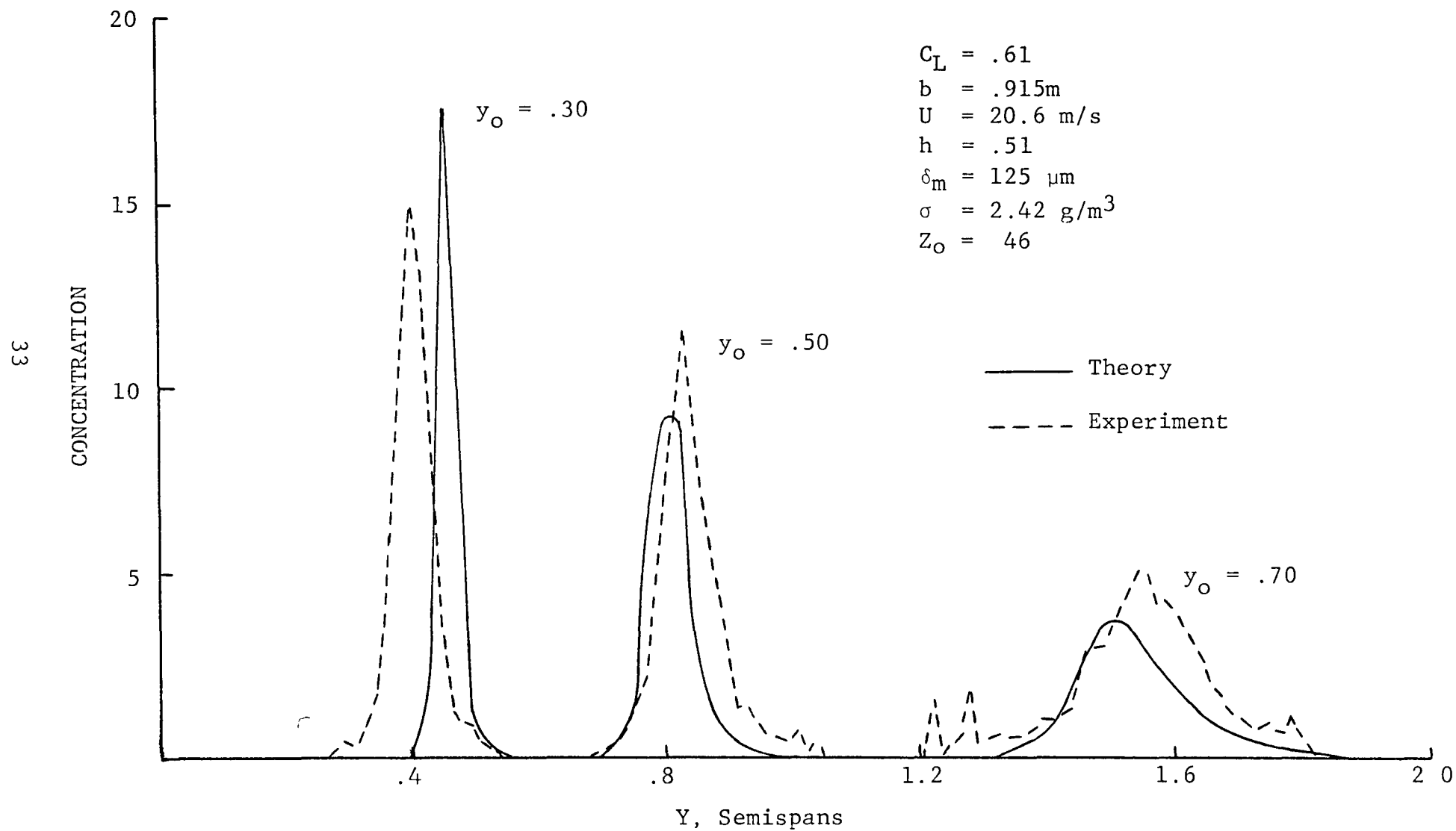


Figure 6. Comparison of Theory to Experiment for Three Nozzle Locations

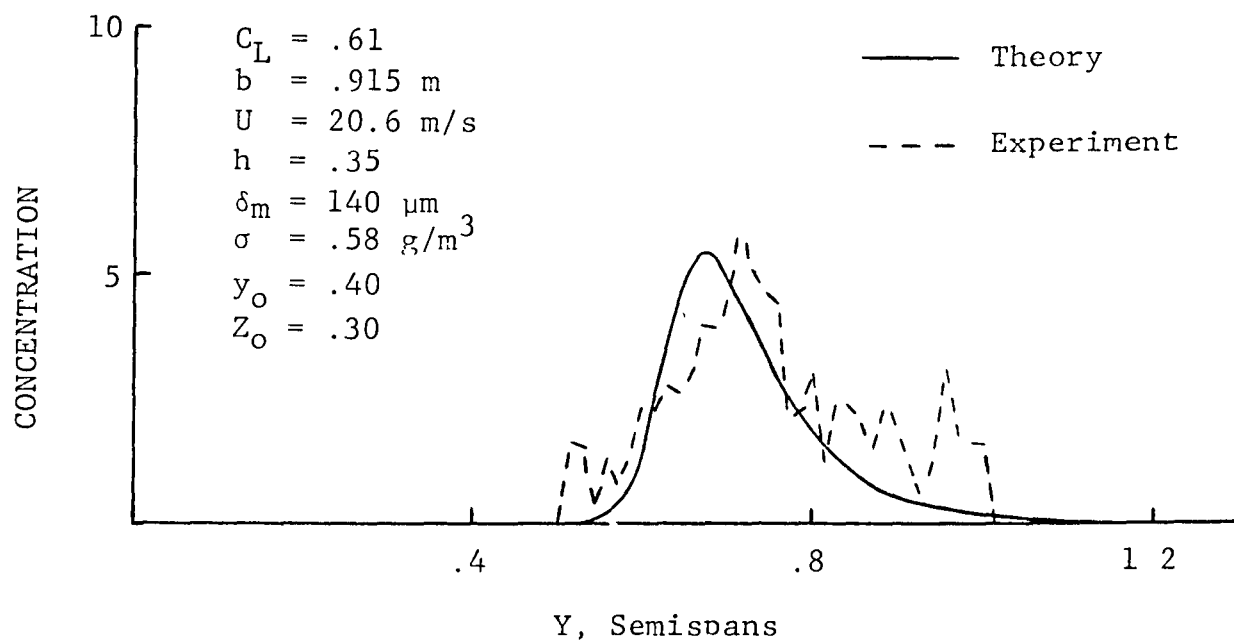


Figure 7 Comparison of Theory to Experiment

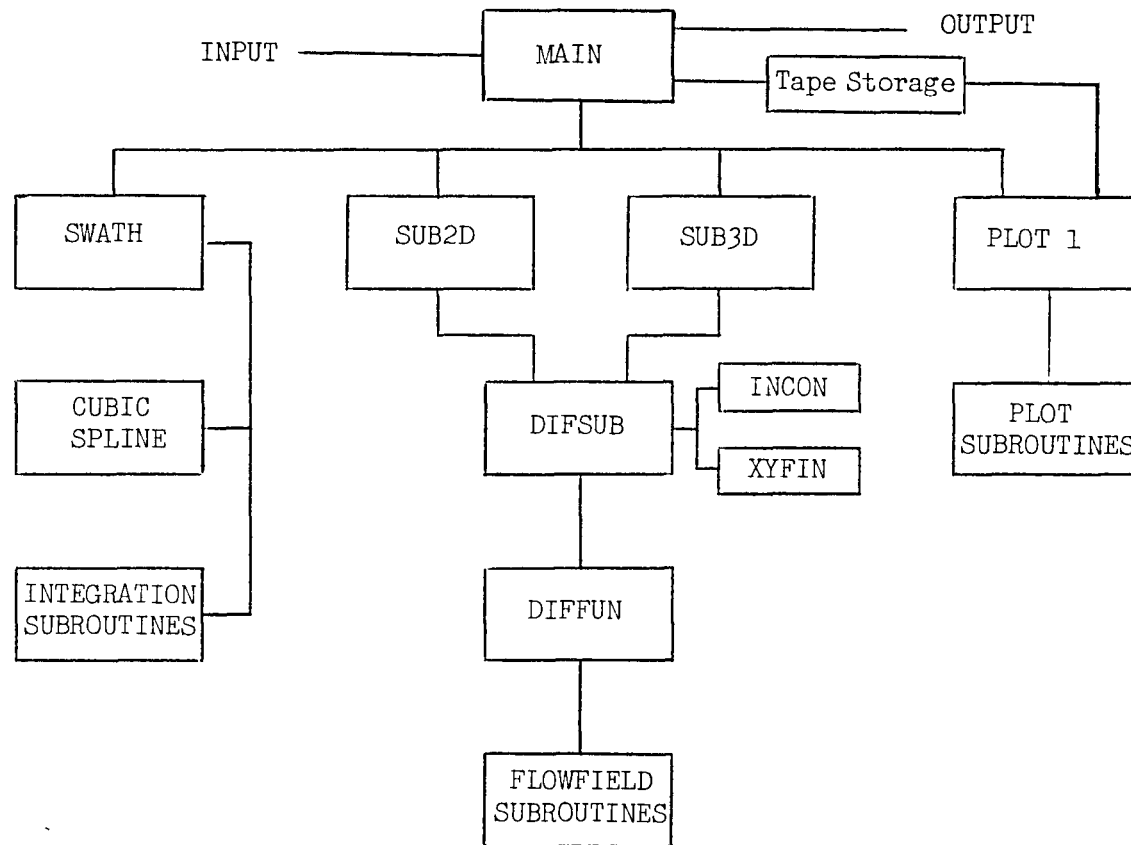


Figure 8. Computer Program Flow Chart

	N2D=1	N2D=2	N3D=1	N3D=2	NEVAP	NPROP2	NCW	NTUN	NDIST	NPLOT	NPRINT
N2D=1	X	-	-	-	X	X	X	X	X	X	X
N2D=2	-	X	-	-	X	-	-	-	X	X	X
N3D=1	-	-	X	-	X	X	X	-	X	X	X
N3D=2	-	-	-	X	X	-	-	-	X	X	X
NEVAP	X	X	X	X	X	X	X	X	X	X	X
NPROP2	X	-	X	-	X	X	X	X	X	X	X
NCW	X	-	X	-	X	X	X	-	X	X	X
NTUN	X	-	-	-	X	X	-	X	X	X	X
NDIST	X	X	X	X	X	X	X	X	X	X	X
NPRINT	X	X	X	X	X	X	X	X	X	X	X
NPLOT	X	X	X	X	X	X	X	X	X	X	X

X Allowable Option
 - Unacceptable Option

Figure 9. Program Option Available

APPENDIX A

FIVE SAMPLE CASES

All five sets of input data are listed first, then each sample output including the plot.

ASP TEST CASE 1

```

1
0 1 0 1 1 0 0 1 1      1.E-5
.50      1.793      17.93      0.      6.029      0.20
6.      .6      6.096      53.035      9.8066      .8      .5
      1.725F00      1.7932E-5 2.5      0.
200.      1000.      0.      .10      .40      0.      0.      0.
$EOJ

```

ASP TEST CASE 2

```

1
1 0 0 0 0 0 1 1 1      1.E-5
6.      .6      6.096      53.035      9.8066      .8      .5
      1.725F00      1.7932E-5 0.      0.
1 6      1      1      1
1000.      0.      0.      0.      0.      4.0      1.      5.
.40
600.      300.      250.      200.      150.      100.
200.
50.
1.
.4

```

ASP TEST CASE 3

```

1
1 0 1 0 0 0 1 0 1      1.E-5
101352.1 23.8889      18.3333
6.      .6      6.096      53.035      9.8066      .8      .5
      1.725F00      1.7932E-5 0.      0.
1 6      1      1      1      1
1000.      0.      0.      0.      0.      4.0      1.      5.
.40
600.      300.      250.      200.      150.      100.
200.
50.
1.
.4

```

ASP TEST CASE 4

```

1
1 0 0 0 0 1 1 0 1      1.E-5
6.      .6      6.096      53.035      9.8066      .8      .5
      1.225F00      1.7932E-5 0.      36.576
1      6      1      1      1      1
1000.      0.      0.      0.      0.      4.0      1.      5.
.40
600.      300.      250.      200.      150.      100.
200.
50.
1.
.4

```

ASP TEST CASE 5

```

1
0 1 1 1 1 0 1 0 1      1.E-5
101352.1      23.8889      18.3333
.50      1.793      17.93      0.      6.029      .20
6.      .6      6.096      53.035      9.8066      .8      .5
      1.225F00      1.7932E-5 1.0      0.
3      6      1      1      1      1
1000.      0.      0.      0.      0.      4.0      1.      5.
-.30      0.      .30
600.      300.      250.      200.      150.      100.
200.      200.      200.
50.      50.      50.
1.      1.      1.
.4      .4      .4

```

OSU / AARL

AERIAL SPRAY PROGRAM

LAST UPDATE 2/1/81

ASP TEST CASE 1

PROGRAM CONTROL

N2D= 0 N3D= 1 NEVAP= 0 NPROP2= 1 NCW= 1
NTUN= 0 NDIST= 0 NPRINT= 1 NPLOT= 1 EPS= 0.100000E-04

PARAMETER VALUES

A= 6.0000 CL= 0.6000 B= 6.0960 U= 53.0350 G= 9.8066
ZY0= 0.8000 ZZO= 0.5000 DA= 0.122500E+01 VIS= 0.179320E-04 VCW= 2.5000
D= 0.0000 DD= 1000.0000

PROP2 INPUTS

ZPROP2= 0.5000 PK1= 1.7930 PK2= 17.9300 PK3= 0.0000 PK4= 6.0290
RP= 0.2000

INITIAL VALUES

X= 0.000000 Y= 0.100000 Z= 0.400000 UD= 0.000000 VD= 0.000000
WD= 0.000000 DIA= 200.0000

1	0.00010	0.10000	0.00003	0.40000	-0.00011	0.00000	0.00142	0.80001	0.50000	-0.80000	0.50000	0.00001	0.50000
2	0.00020	0.10000	0.00005	0.40000	-0.00022	0.00000	0.00283	0.80002	0.50000	-0.79999	0.50000	0.00001	0.49999
3	0.00116	0.10000	0.00030	0.40000	-0.00124	0.00001	0.01625	0.80009	0.50000	-0.79994	0.50000	0.00007	0.49996
4	0.00212	0.10000	0.00054	0.40000	-0.00225	0.00003	0.02937	0.80016	0.49999	-0.79990	0.49999	0.00013	0.49993
5	0.00308	0.10000	0.00077	0.39999	-0.00323	0.00007	0.04219	0.80024	0.49999	-0.79985	0.49999	0.00019	0.49990
6	0.00472	0.10000	0.00116	0.39999	-0.00487	0.00015	0.06347	0.80036	0.49998	-0.79977	0.49998	0.00029	0.49985
7	0.00636	0.10001	0.00153	0.39998	-0.00644	0.00027	0.08395	0.80049	0.49998	-0.79970	0.49998	0.00040	0.49980
8	0.00801	0.10001	0.00189	0.39997	-0.00777	0.00043	0.10368	0.80062	0.49997	-0.79962	0.49997	0.00050	0.49975
9	0.00965	0.10001	0.00224	0.39995	-0.00945	0.00061	0.12269	0.80074	0.49997	-0.79954	0.49997	0.00060	0.49970
10	0.01185	0.10002	0.00268	0.39993	-0.01136	0.00091	0.14711	0.80091	0.49996	-0.79943	0.49996	0.00074	0.49963
11	0.01405	0.10002	0.00310	0.39990	-0.01321	0.00126	0.17040	0.80108	0.49995	-0.79933	0.49995	0.00088	0.49957
12	0.01625	0.10003	0.00350	0.39987	-0.01499	0.00166	0.19263	0.80125	0.49994	-0.79922	0.49994	0.00101	0.49950
13	0.01845	0.10004	0.00388	0.39984	-0.01671	0.00211	0.21387	0.80142	0.49993	-0.79912	0.49993	0.00115	0.49943
14	0.02225	0.10005	0.00449	0.39977	-0.01957	0.00294	0.24838	0.80171	0.49992	-0.79894	0.49992	0.00139	0.49931
15	0.02605	0.10007	0.00505	0.39969	-0.02231	0.00399	0.28043	0.80200	0.49991	-0.79876	0.49991	0.00162	0.49920
16	0.02986	0.10009	0.00557	0.39960	-0.02494	0.00512	0.31025	0.80229	0.49989	-0.79858	0.49989	0.00186	0.49908
17	0.03366	0.10011	0.00606	0.39950	-0.02749	0.00635	0.33808	0.80259	0.49988	-0.79839	0.49988	0.00210	0.49896
18	0.03746	0.10014	0.00650	0.39939	-0.02998	0.00768	0.36411	0.80288	0.49987	-0.79821	0.49987	0.00233	0.49884
19	0.04210	0.10017	0.00701	0.39924	-0.03294	0.00944	0.39372	0.80324	0.49985	-0.79799	0.49985	0.00262	0.49870
20	0.04674	0.10020	0.00747	0.39908	-0.03584	0.01134	0.42117	0.80359	0.49983	-0.79777	0.49983	0.00291	0.49856
21	0.05139	0.10024	0.00789	0.39891	-0.03870	0.01335	0.44670	0.80395	0.49982	-0.79755	0.49982	0.00320	0.49842
22	0.05603	0.10028	0.00828	0.39872	-0.04152	0.01548	0.47052	0.80431	0.49980	-0.79733	0.49980	0.00349	0.49827
23	0.06067	0.10032	0.00864	0.39853	-0.04432	0.01772	0.49281	0.80466	0.49978	-0.79711	0.49978	0.00378	0.49813

25	0.07287	U.10043	0.00945	0.39794	-0.05157	0.02406	0.54516	0.80560	0.49974	-0.79652	0.49974	0.00454	0.49775
26	0.07897	U.10049	0.00940	0.39761	-0.05516	0.02745	0.56851	0.80607	0.49972	-0.79623	0.49972	0.00492	0.49757
27	0.08506	U.10055	0.01011	0.39727	-0.05873	0.03049	0.59031	0.80654	0.49970	-0.79594	0.49970	0.00529	0.49738
28	0.09116	U.10061	0.01040	0.39690	-0.06228	0.03465	0.61073	0.80701	0.49968	-0.79565	0.49968	0.00567	0.49719
29	0.09726	U.10067	0.01065	0.39651	-0.06579	0.03843	0.62994	0.80748	0.49965	-0.79536	0.49965	0.00605	0.49700
30	0.10709	U.10078	0.01102	0.39583	-0.07136	0.04477	0.65870	0.80823	0.49962	-0.79489	0.49962	0.00666	0.49670
31	0.11692	U.10089	0.01133	0.39511	-0.07679	0.05137	0.68515	0.80899	0.49959	-0.79442	0.49959	0.00728	0.49640
32	0.12675	U.10100	0.01159	0.39433	-0.08203	0.05823	0.70963	0.80974	0.49955	-0.79396	0.49955	0.00789	0.49610
33	0.13658	U.10112	0.01181	0.39349	-0.08705	0.06532	0.73244	0.81050	0.49952	-0.79349	0.49952	0.00850	0.49580
34	0.14641	U.10123	0.01199	0.39261	-0.09179	0.07262	0.75378	0.81125	0.49948	-0.79302	0.49948	0.00911	0.49550
35	0.15623	U.10135	0.01214	0.39169	-0.09624	0.08013	0.77381	0.81201	0.49945	-0.79255	0.49945	0.00972	0.49520
36	0.17090	U.10153	0.01231	0.39023	-0.10229	0.09169	0.80153	0.81314	0.49939	-0.79185	0.49939	0.01063	0.49475
37	0.18556	U.10171	0.01242	0.38869	-0.10758	0.10363	0.82692	0.81427	0.49934	-0.79115	0.49934	0.01154	0.49430
38	0.20023	U.10190	0.01249	0.38708	-0.11213	0.11593	0.85024	0.81539	0.49929	-0.79045	0.49929	0.01245	0.49385
39	0.21489	U.10208	0.01251	0.38541	-0.11596	0.12856	0.87167	0.81652	0.49924	-0.78976	0.49924	0.01336	0.49340
40	0.22956	U.10226	0.01251	0.38368	-0.11910	0.14149	0.89137	0.81765	0.49919	-0.78906	0.49919	0.01427	0.49296
41	0.24422	U.10245	0.01247	0.38192	-0.12163	0.15470	0.90948	0.81878	0.49914	-0.78836	0.49914	0.01518	0.49251
42	0.26215	U.10267	0.01240	0.37972	-0.12396	0.17119	0.92765	0.82016	0.49907	-0.78751	0.49907	0.01630	0.49197
43	0.28008	U.10289	0.01231	0.37748	-0.12557	0.18802	0.94785	0.82153	0.49901	-0.78665	0.49901	0.01741	0.49142
44	0.29801	U.10311	0.01220	0.37522	-0.12656	0.20516	0.96427	0.82291	0.49895	-0.78580	0.49895	0.01852	0.49088
45	0.31594	U.10333	0.01207	0.37294	-0.12704	0.22259	0.97911	0.82429	0.49889	-0.78495	0.49889	0.01963	0.49033
46	0.33387	U.10354	0.01194	0.37066	-0.12702	0.24026	0.99252	0.82567	0.49882	-0.78409	0.49882	0.02074	0.48979
47	0.35180	U.10376	0.01181	0.36839	-0.12679	0.25817	1.00464	0.82705	0.49876	-0.78324	0.49876	0.02185	0.48925
48	0.37972	U.10408	0.01160	0.36486	-0.12575	0.28646	1.02126	0.82920	0.49866	-0.78191	0.49866	0.02359	0.48840
49	0.40764	U.10440	0.01141	0.36137	-0.12420	0.31517	1.03552	0.83135	0.49857	-0.78059	0.49857	0.02531	0.48755
50	0.43556	U.10472	0.01123	0.35793	-0.12229	0.34426	1.04777	0.83349	0.49847	-0.77926	0.49847	0.02704	0.48672
51	0.46348	U.10503	0.01109	0.35454	-0.12012	0.37366	1.05830	0.83564	0.49837	-0.77793	0.49837	0.02876	0.48588
52	0.49140	U.10534	0.01097	0.35122	-0.11778	0.40334	1.06737	0.83779	0.49828	-0.77660	0.49828	0.03049	0.48504
53	0.51932	U.10564	0.01080	0.34797	-0.11534	0.43325	1.07518	0.83994	0.49818	-0.77528	0.49818	0.03222	0.48421
54	0.55757	U.10606	0.01062	0.34362	-0.11190	0.47457	1.08415	0.84288	0.49805	-0.77346	0.49805	0.03458	0.48306
55	0.59583	U.10647	0.01042	0.33941	-0.10843	0.51619	1.09145	0.84583	0.49791	-0.77165	0.49791	0.03694	0.48192
56	0.63409	U.10689	0.01049	0.33532	-0.10500	0.55806	1.09736	0.84877	0.49778	-0.76983	0.49778	0.03930	0.48078
57	0.67235	U.10731	0.01102	0.33137	-0.10162	0.60014	1.10212	0.85172	0.49765	-0.76802	0.49765	0.04166	0.47965
58	0.71060	U.10773	0.01121	0.32755	-0.09832	0.64238	1.10590	0.85466	0.49752	-0.76620	0.49752	0.04402	0.47852
59	0.74886	U.10817	0.01147	0.32385	-0.09512	0.68474	1.10885	0.85761	0.49739	-0.76439	0.49739	0.04637	0.47739
60	0.80074	U.10877	0.01190	0.31902	-0.09093	0.74235	1.11175	0.86161	0.49721	-0.76193	0.49721	0.04956	0.47580
61	0.85263	U.10940	0.01243	0.31441	-0.08690	0.80009	1.11358	0.86560	0.49704	-0.75947	0.49704	0.05275	0.47433
62	0.90451	U.11006	0.01306	0.31000	-0.08302	0.85789	1.11452	0.86960	0.49686	-0.75701	0.49686	0.05594	0.47282
63	0.95639	U.11076	0.01377	0.30579	-0.07929	0.91573	1.11469	0.87360	0.49668	-0.75456	0.49668	0.05913	0.47130
64	1.00828	U.11149	0.01457	0.30177	-0.07570	0.97355	1.11420	0.87759	0.49651	-0.75210	0.49651	0.06231	0.46980
65	1.06016	U.11227	0.01545	0.29793	-0.07226	1.03133	1.11315	0.88159	0.49634	-0.74965	0.49634	0.06549	0.46829
66	1.13080	U.11341	0.01675	0.29299	-0.06780	1.10990	1.11092	0.88704	0.49610	-0.74631	0.49610	0.06981	0.46625
67	1.20144	U.11464	0.01814	0.28835	-0.06363	1.18827	1.10797	0.89240	0.49586	-0.74298	0.49586	0.07413	0.46423
68	1.27209	U.11597	0.01959	0.28399	-0.05978	1.26642	1.10446	0.89793	0.49563	-0.73964	0.49563	0.07845	0.46221
69	1.34273	U.11741	0.02108	0.27990	-0.05624	1.34431	1.10055	0.90338	0.49539	-0.73631	0.49539	0.08276	0.46020
70	1.41337	U.11895	0.02257	0.27604	-0.05303	1.42191	1.09639	0.90883	0.49516	-0.73298	0.49516	0.08707	0.45820
71	1.48401	U.12060	0.02404	0.27240	-0.05013	1.49921	1.09209	0.91428	0.49493	-0.72965	0.49493	0.09137	0.45621
72	1.57953	U.12299	0.02536	0.26778	-0.04669	1.60324	1.08624	0.92165	0.49467	-0.72515	0.49467	0.09571	0.45383
73	1.67504	U.12555	0.02777	0.26346	-0.04374	1.70671	1.08048	0.92902	0.49431	-0.72066	0.49431	0.10297	0.45057
74	1.77055	U.12829	0.02946	0.25941	-0.04123	1.80964	1.07496	0.93639	0.49400	-0.71617	0.49400	0.10876	0.44823
75	1.86606	U.13118	0.03102	0.25557	-0.03911	1.91206	1.06973	0.94377	0.49369	-0.71168	0.49369	0.11454	0.44560
76	1.96158	U.13421	0.03244	0.25193	-0.03730	2.01399	1.06482	0.95114	0.49338	-0.70720	0.49338	0.12031	0.44299
77	2.08133	U.13819	0.03404	0.24757	-0.03542	2.14116	1.05915	0.96039	0.49300	-0.70159	0.49300	0.12753	0.43975
78	2.20108	U.14235	0.03545	0.24343	-0.03389	2.26768	1.05397	0.96965	0.49267	-0.69598	0.49267	0.13474	0.43653
79	2.32083	U.14668	0.03671	0.23945	-0.03263	2.39361	1.04925	0.97890	0.49225	-0.69037	0.49225	0.14194	0.43334
80	2.44058	U.15114	0.03792	0.23560	-0.03159	2.51900	1.04494	0.98816	0.49187	-0.68478	0.49187	0.14912	0.43017
81	2.56033	U.15573	0.03881	0.23187	-0.03072	2.64389	1.04098	0.99743	0.49150	-0.67918	0.49150	0.15628	0.42703
82	2.68008	U.16043	0.03970	0.22824	-0.02999	2.76832	1.03732	1.00669	0.49113	-0.67360	0.49113	0.16343	0.42392
83	2.83746	U.16676	0.04074	0.22358	-0.02919	2.93123	1.03288	1.01887	0.49065	-0.66626	0.49065	0.17281	0.41987
84	2.99485	U.17325	0.04167	0.21904	-0.02852	3.09346	1.02877	1.03106	0.49017	-0.65894	0.49017	0.18216	0.41587
85	3.15224	U.17987	0.04252	0.21460	-0.02796	3.25507	1.02491	1.04325	0.48970	-0.65162	0.48970	0.19149	0.41192
86	3.30962	U.18663	0.04331	0.21024	-0.02748	3.41609	1.02123	1.05545	0.48923	-0.64432	0.48923	0.20079	0.40800
87	3.46701	U.19350	0.04405	0.20595	-0.02705	3.57653	1.01769	1.06765	0.48877	-0.63702	0.48877	0.21007	0.40414
88	3.62439	U.20049	0.04476	0.20172	-0.02667	3.73643	1.01424	1.07985	0.48831	-0.62973	0.48831	0.21932	0.40032
89	3.75188	U.20823	0.04533	0.19834	-0.02640	3.89556	1.01149	1.08974	0.48793	-0.62384	0.48793	0.22679	0.39726

90	3.79186	0.20805	0.04549	0.19729	-0.02632	3.90597	1.01067	1.09284	0.48782	-0.62199	0.48782	0.22913	0.39630
91	3.81903	0.20929	0.04559	0.19657	-0.02627	3.93342	1.01014	1.09495	0.48774	-0.62073	0.48774	0.23072	0.39566
92	3.84620	0.21053	0.04568	0.19586	-0.02623	3.96086	1.00964	1.09706	0.48766	-0.61948	0.48766	0.23231	0.39501
93	3.87337	0.21177	0.04577	0.19515	-0.02619	3.98828	1.00917	1.09917	0.48758	-0.61822	0.48758	0.23390	0.39437
94	3.95217	0.21539	0.04597	0.19309	-0.02607	4.06776	1.00793	1.10528	0.48736	-0.61459	0.48736	0.23850	0.39251
95	4.03097	0.21901	0.04612	0.19104	-0.02598	4.14714	1.00687	1.11140	0.48713	-0.61095	0.48713	0.24310	0.39066
96	4.10977	0.22265	0.04622	0.18899	-0.02588	4.22644	1.00596	1.11752	0.48691	-0.60732	0.48691	0.24769	0.38882
97	4.18857	0.22630	0.04628	0.18696	-0.02579	4.30568	1.00517	1.12364	0.48669	-0.60369	0.48669	0.25227	0.38699
98	4.29082	0.23103	0.04630	0.18433	-0.02567	4.40842	1.00430	1.13158	0.48640	-0.59898	0.48640	0.25821	0.38464
99	4.39308	0.23577	0.04628	0.18171	-0.02554	4.51108	1.00358	1.13953	0.48611	-0.59427	0.48611	0.26413	0.38230
100	4.49533	0.24049	0.04622	0.17910	-0.02541	4.61366	1.00297	1.14747	0.48583	-0.58957	0.48583	0.27005	0.37998
101	4.59759	0.24522	0.04613	0.17651	-0.02526	4.71620	1.00246	1.15542	0.48554	-0.58488	0.48554	0.27596	0.37768
102	4.69984	0.24993	0.04600	0.17394	-0.02511	4.81868	1.00204	1.16337	0.48526	-0.58018	0.48526	0.28185	0.37540
103	4.87481	0.25795	0.04574	0.16957	-0.02483	4.99395	1.00147	1.17698	0.48478	-0.57216	0.48478	0.29191	0.37154
104	5.04977	0.26593	0.04542	0.16525	-0.02453	5.16913	1.00105	1.19059	0.48431	-0.56615	0.48431	0.30194	0.36773
105	5.22473	0.27384	0.04506	0.16098	-0.02422	5.34425	1.00075	1.20421	0.48384	-0.55615	0.48384	0.31193	0.36397
106	5.39970	0.28169	0.04466	0.15677	-0.02391	5.51933	1.00052	1.21784	0.48337	-0.54816	0.48337	0.32190	0.36026
107	5.57466	0.28947	0.04424	0.15262	-0.02359	5.69437	1.00036	1.23147	0.48291	-0.54018	0.48291	0.33183	0.35661
108	5.84230	0.30122	0.04356	0.14637	-0.02311	5.96208	1.00019	1.25233	0.48222	-0.52800	0.48222	0.34695	0.35111
109	6.10994	0.31279	0.04285	0.14025	-0.02263	6.22976	1.00009	1.27320	0.48153	-0.51584	0.48153	0.36201	0.34573
110	6.37758	0.32416	0.04211	0.13425	-0.02217	6.49741	1.00003	1.29409	0.48085	-0.50370	0.48085	0.37698	0.34047
111	6.64523	0.33533	0.04135	0.12838	-0.02171	6.76506	1.00000	1.31499	0.48019	-0.49159	0.48019	0.39188	0.33531
112	6.91287	0.34629	0.04057	0.12263	-0.02127	7.03269	0.99998	1.33590	0.47953	-0.47950	0.47953	0.40670	0.33027
113	7.18051	0.35704	0.03977	0.11699	-0.02084	7.30033	0.99996	1.35682	0.47888	-0.46744	0.47888	0.42145	0.32534
114	7.55511	0.37173	0.03863	0.10930	-0.02026	7.67491	0.99996	1.38813	0.47799	-0.45059	0.47799	0.44196	0.31861
115	7.92970	0.38598	0.03745	0.10181	-0.01971	8.04949	0.99996	1.41546	0.47712	-0.43379	0.47712	0.46231	0.31208
116	8.30430	0.39978	0.03624	0.09453	-0.01918	8.42407	0.99996	1.44481	0.47627	-0.41704	0.47627	0.48251	0.30574
117	8.67890	0.41313	0.03500	0.08744	-0.01867	8.79866	0.99997	1.47419	0.47543	-0.40033	0.47543	0.50256	0.29960
118	9.05349	0.42600	0.03371	0.08053	-0.01819	9.17324	0.99997	1.50359	0.47461	-0.38366	0.47461	0.52246	0.29364
119	9.42809	0.43838	0.03238	0.07381	-0.01773	9.54783	0.99997	1.53302	0.47380	-0.36703	0.47380	0.54220	0.28786
120	10.05584	0.45798	0.03005	0.06291	-0.01700	10.17556	0.99998	1.58237	0.47249	-0.33926	0.47249	0.57494	0.27856
121	10.68359	0.47607	0.02756	0.05246	-0.01631	10.80329	0.99998	1.63180	0.47123	-0.31161	0.47123	0.60725	0.26972
122	11.31133	0.49254	0.02486	0.04242	-0.01567	11.43103	0.99998	1.68128	0.47000	-0.28407	0.47000	0.63913	0.26131
123	11.93908	0.50723	0.02190	0.03277	-0.01506	12.05877	0.99999	1.73082	0.46882	-0.25664	0.46882	0.67058	0.25331
124	12.56683	0.51995	0.01857	0.02350	-0.01448	12.68651	0.99999	1.78042	0.46767	-0.22931	0.46767	0.70160	0.24570
125	13.19458	0.53041	0.01464	0.01459	-0.01390	13.31425	0.99999	1.83008	0.46656	-0.20209	0.46656	0.73219	0.23845
126	13.53726	0.53500	0.01209	0.00988	-0.01359	13.65693	0.99999	1.85721	0.46597	-0.18728	0.46597	0.74872	0.23463
127	13.79823	0.53786	0.00982	0.00637	-0.01333	13.91789	0.99999	1.87788	0.46553	-0.17601	0.46553	0.76122	0.23180
128	13.98649	0.53953	0.00788	0.00388	-0.01313	14.10615	0.99999	1.89280	0.46521	-0.16790	0.46521	0.77019	0.22978
129	14.09471	0.54032	0.00657	0.00247	-0.01301	14.21438	0.99999	1.90138	0.46503	-0.16324	0.46503	0.77533	0.22864
130	14.17140	0.54078	0.00548	0.00147	-0.01291	14.29106	0.99999	1.90746	0.46491	-0.15994	0.46491	0.77896	0.22783
131	14.22379	0.54104	0.00462	0.00080	-0.01282	14.34345	0.99999	1.91162	0.46482	-0.15768	0.46482	0.78144	0.22729
132	14.25067	0.54116	0.00410	0.00045	-0.01277	14.37034	0.99999	1.91375	0.46478	-0.15653	0.46478	0.78271	0.22701
133	14.26808	0.54123	0.00372	0.00023	-0.01274	14.38775	0.99999	1.91513	0.46475	-0.15578	0.46475	0.78354	0.22682
134	14.27811	0.54126	0.00347	0.00010	-0.01271	14.39778	0.99999	1.91593	0.46473	-0.15535	0.46473	0.78401	0.22672
135	14.28217	0.54128	0.00335	0.00005	-0.01270	14.40184	0.99999	1.91625	0.46473	-0.15517	0.46473	0.78420	0.22668
136	14.28461	0.54128	0.00328	0.00002	-0.01269	14.40428	0.99999	1.91644	0.46472	-0.15507	0.46472	0.78432	0.22665
137	14.28627	0.54129	0.00321	0.00000	-0.01268	14.40594	0.99999	1.91657	0.46472	-0.15500	0.46472	0.78440	0.22664
138	14.28793	0.54129	0.00314	-0.00002	-0.01268	14.40760	0.99999	1.91670	0.46472	-0.15492	0.46472	0.78447	0.22662

*** TRAJECTORY COMPLETED, 7 LESS THAN ZERO ***

FINAL VALUES

X= 14.405961
WD= -0.012684

Y= 0.541290
DIA= 200.0000

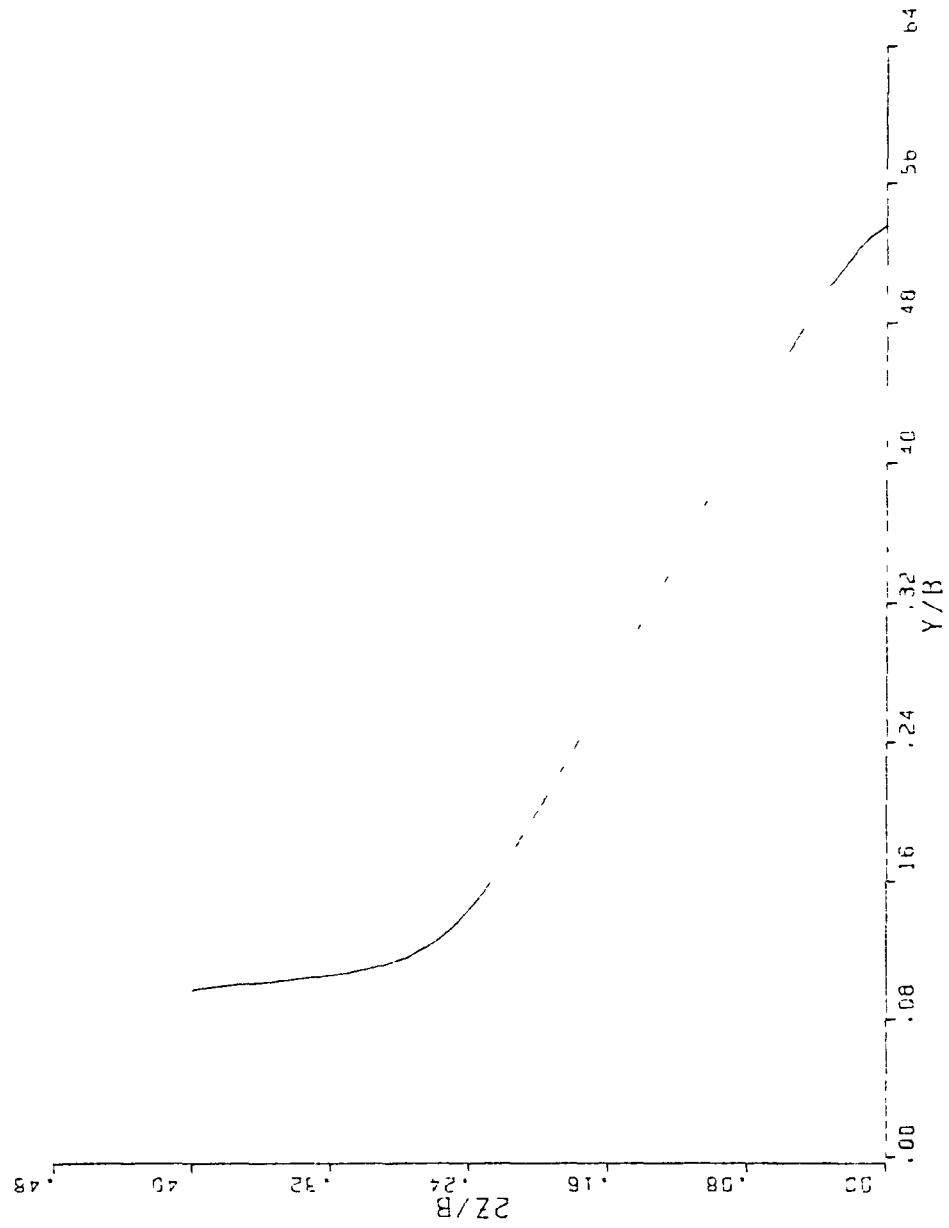
Z= 0.000000
T= 14.286297

UD= 0.999992

VD= 0.003211

HAKGND: STOP

ASP TEST CASE 1



OSU / AARL

AERIAL SPRAY PROGRAM

LAST UPDATE 2/1/81

ASP TEST CASE 2

PROGRAM CONTROL

N2D= 1	N3D= 0	NEVAP= 0	NPROP2= 0	NCW= 0
NTUN= 0	NDIST= 1	NPRINT= 1	NPLOT= 1	EPS= 0.100000E-04

PARAMETER VALUES

A= 6.0000	CL= 0.6000	B= 6.0960	U= 53.0350	G= 9.8066
ZY0= 0.8000	Z70= 0.5000	DA= 0.122500E+01	VIS= 0.179320E-04	VCW= 0.0000
D= 0.0000	DD= 1000.0000			

DISTRIBUTION INPUTS

NCOL= 1	NROW= 6	NDIAMN= 1	NSTDEV= 1	NQ= 1
NZNOZ= 1	DD=1000.00000	X= 0.00000	UD= 0.00000	VD= 0.00000
WD= 0.00000	SWIDTH= 4.00000	DFRR= 1.00000	DWIDTH= 5.00000	

YNOZ=0.400

YG	DIA	CONCENTRATION	YG	DIA	CONCENTRATION	YG	DIA	CONCENTRATION
0.640000E+00	0.441041E+03	0.110564E-03	0.650000E+00	0.426213E+03	0.408700E-03	0.660000E+00	0.412488E+03	0.126168E-02
0.670000E+00	0.399762E+03	0.334558E-02	0.680000E+00	0.387928E+03	0.779777E-02	0.690000E+00	0.376896E+03	0.162809E-01
0.700000E+00	0.366585E+03	0.309327E-01	0.710000E+00	0.356927E+03	0.541859E-01	0.720000E+00	0.347861E+03	0.884828E-01
0.730000E+00	0.339334E+03	0.135946E+00	0.740000E+00	0.331299E+03	0.198073E+00	0.750000E+00	0.323714E+03	0.275513E+00
0.760000E+00	0.316543E+03	0.367965E+00	0.770000E+00	0.309752E+03	0.474192E+00	0.780000E+00	0.303311E+03	0.592145E+00
0.790000E+00	0.297195E+03	0.719134E+00	0.800000E+00	0.291380E+03	0.851925E+00	0.810000E+00	0.285844E+03	0.987170E+00
0.820000E+00	0.280570E+03	0.112161E+01	0.830000E+00	0.275540E+03	0.125221E+01	0.840000E+00	0.270738E+03	0.137634E+01
0.850000E+00	0.266150E+03	0.149180E+01	0.860000E+00	0.261762E+03	0.159689E+01	0.870000E+00	0.257562E+03	0.169039E+01
0.880000E+00	0.253538E+03	0.177154E+01	0.890000E+00	0.249681E+03	0.184000E+01	0.900000E+00	0.245979E+03	0.189559E+01
0.910000E+00	0.242426E+03	0.193848E+01	0.920000E+00	0.239012E+03	0.196920E+01	0.930000E+00	0.235731E+03	0.198850E+01
0.940000E+00	0.232576E+03	0.199723E+01	0.950000E+00	0.229539E+03	0.199634E+01	0.960000E+00	0.226616E+03	0.198681E+01
0.970000E+00	0.223799E+03	0.196964E+01	0.980000E+00	0.221084E+03	0.194580E+01	0.990000E+00	0.218467E+03	0.191622E+01
0.100000E+01	0.215941E+03	0.188178E+01	0.101000E+01	0.213502E+03	0.184330E+01	0.102000E+01	0.211147E+03	0.180153E+01
0.103000E+01	0.208872E+03	0.175716E+01	0.104000E+01	0.206672E+03	0.171079E+01	0.105000E+01	0.204544E+03	0.166297E+01
0.106000E+01	0.202485E+03	0.161418E+01	0.107000E+01	0.200492E+03	0.156483E+01	0.108000E+01	0.198561E+03	0.151536E+01
0.109000E+01	0.196691E+03	0.146626E+01	0.110000E+01	0.194877E+03	0.141774E+01	0.111000E+01	0.193117E+03	0.136998E+01
0.112000E+01	0.191409E+03	0.132313E+01	0.113000E+01	0.189751E+03	0.127730E+01	0.114000E+01	0.188139E+03	0.123259E+01
0.115000E+01	0.186572E+03	0.118905E+01	0.116000E+01	0.185048E+03	0.114675E+01	0.117000E+01	0.183566E+03	0.110572E+01
0.118000E+01	0.182122E+03	0.106597E+01	0.119000E+01	0.180717E+03	0.102752E+01	0.120000E+01	0.179347E+03	0.990355E+00
0.121000E+01	0.178013E+03	0.954498E+00	0.122000E+01	0.176711E+03	0.919905E+00	0.123000E+01	0.175442E+03	0.886566E+00
0.124000E+01	0.174204E+03	0.854457E+00	0.125000E+01	0.172995E+03	0.823551E+00	0.126000E+01	0.171815E+03	0.793816E+00
0.127000E+01	0.170662E+03	0.765222E+00	0.128000E+01	0.169536E+03	0.737734E+00	0.129000E+01	0.168436E+03	0.711317E+00
0.130000E+01	0.167360E+03	0.685937E+00	0.131000E+01	0.166308E+03	0.661557E+00	0.132000E+01	0.165279E+03	0.638143E+00
0.133000E+01	0.164272E+03	0.615658E+00	0.134000E+01	0.163286E+03	0.594069E+00	0.135000E+01	0.162322E+03	0.573341E+00
0.136000E+01	0.161377E+03	0.553441E+00	0.137000E+01	0.160452E+03	0.534334E+00	0.138000E+01	0.159546E+03	0.515991E+00
0.139000E+01	0.158657E+03	0.498379E+00	0.140000E+01	0.157787E+03	0.481469E+00	0.141000E+01	0.156934E+03	0.465232E+00
0.142000E+01	0.156097E+03	0.449639E+00	0.143000E+01	0.155276E+03	0.434664E+00	0.144000E+01	0.154471E+03	0.420280E+00
0.145000E+01	0.153681E+03	0.406463E+00	0.146000E+01	0.152906E+03	0.393188E+00	0.147000E+01	0.152146E+03	0.380433E+00
0.148000E+01	0.151399E+03	0.368174E+00	0.149000E+01	0.150665E+03	0.356392E+00	0.150000E+01	0.149945E+03	0.345064E+00
0.151000E+01	0.149238E+03	0.334066E+00	0.152000E+01	0.148543E+03	0.323321E+00	0.153000E+01	0.147861E+03	0.317838E+00
0.154000E+01	0.147192E+03	0.302624E+00	0.155000E+01	0.146536E+03	0.292683E+00	0.156000E+01	0.145893E+03	0.283019E+00
0.157000E+01	0.145262E+03	0.273632E+00	0.158000E+01	0.144644E+03	0.264523E+00	0.159000E+01	0.144038E+03	0.255689E+00
0.160000E+01	0.143445E+03	0.247128E+00	0.161000E+01	0.142864E+03	0.238838E+00	0.162000E+01	0.142295E+03	0.230813E+00
0.163000E+01	0.141738E+03	0.223049E+00	0.164000E+01	0.141192E+03	0.215542E+00	0.165000E+01	0.140659E+03	0.208284E+00
0.166000E+01	0.140136E+03	0.201271E+00	0.167000E+01	0.139626E+03	0.194496E+00	0.168000E+01	0.139126E+03	0.187953E+00
0.169000E+01	0.138637E+03	0.181636E+00	0.170000E+01	0.138159E+03	0.175537E+00	0.171000E+01	0.137691E+03	0.169650E+00
0.172000E+01	0.137234E+03	0.163969E+00	0.173000E+01	0.136788E+03	0.158487E+00	0.174000E+01	0.136351E+03	0.153197E+00
0.175000E+01	0.135924E+03	0.148094E+00	0.176000E+01	0.135507E+03	0.143170E+00	0.177000E+01	0.135100E+03	0.138419E+00
0.178000E+01	0.134701E+03	0.133837E+00	0.179000E+01	0.134312E+03	0.129415E+00	0.180000E+01	0.133932E+03	0.125150E+00
0.181000E+01	0.133561E+03	0.121034E+00	0.182000E+01	0.133199E+03	0.117063E+00	0.183000E+01	0.132845E+03	0.113232E+00
0.184000E+01	0.132499E+03	0.109534E+00	0.185000E+01	0.132162E+03	0.105966E+00	0.186000E+01	0.131833E+03	0.102522E+00
0.187000E+01	0.131511E+03	0.991974E-01	0.188000E+01	0.131197E+03	0.959882E-01	0.189000E+01	0.130891E+03	0.928897E-01
0.190000E+01	0.130592E+03	0.898979E-01	0.191000E+01	0.130300E+03	0.870086E-01	0.192000E+01	0.130016E+03	0.842382E-01
0.193000E+01	0.129738E+03	0.816066E-01	0.194000E+01	0.129466E+03	0.790880E-01	0.195000E+01	0.129201E+03	0.766933E-01
0.196000E+01	0.128942E+03	0.744098E-01	0.197000E+01	0.128689E+03	0.722311E-01	0.198000E+01	0.128442E+03	0.701514E-01
0.199000E+01	0.128200E+03	0.681652E-01	0.200000E+01	0.127963E+03	0.662675E-01	0.201000E+01	0.127731E+03	0.644532E-01
0.202000E+01	0.127503E+03	0.627181E-01	0.203000E+01	0.127281E+03	0.610577E-01	0.204000E+01	0.127063E+03	0.594682E-01
0.205000E+01	0.126849E+03	0.579458E-01	0.206000E+01	0.126639E+03	0.564870E-01	0.207000E+01	0.126433E+03	0.550886E-01
0.208000E+01	0.126231E+03	0.537473E-01	0.209000E+01	0.126033E+03	0.524604E-01	0.210000E+01	0.125839E+03	0.512251E-01
0.211000E+01	0.125647E+03	0.500387E-01	0.212000E+01	0.125460E+03	0.488990E-01	0.213000E+01	0.125275E+03	0.478034E-01
0.214000E+01	0.125093E+03	0.467500E-01	0.215000E+01	0.124915E+03	0.457366E-01	0.216000E+01	0.124739E+03	0.447614E-01
0.217000E+01	0.124567E+03	0.438225E-01	0.218000E+01	0.124397E+03	0.429181E-01	0.219000E+01	0.124229E+03	0.420668E-01
0.220000E+01	0.124064E+03	0.412068E-01	0.221000E+01	0.123902E+03	0.403969E-01	0.222000E+01	0.123742E+03	0.396155E-01
0.223000E+01	0.123594E+03	0.388614E-01	0.224000E+01	0.123429E+03	0.381334E-01	0.225000E+01	0.123275E+03	0.374303E-01
0.226000E+01	0.123174E+03	0.367510E-01	0.227000E+01	0.122975E+03	0.360944E-01	0.228000E+01	0.122828E+03	0.354595E-01
0.229000E+01	0.122682E+03	0.348454E-01	0.230000E+01	0.122539E+03	0.342512E-01	0.231000E+01	0.122397E+03	0.336761E-01
0.232000E+01	0.122257E+03	0.331191E-01	0.233000E+01	0.122119E+03	0.325796E-01	0.234000E+01	0.121983E+03	0.320569E-01
0.235000E+01	0.121848E+03	0.315501E-01	0.236000E+01	0.121714E+03	0.310587E-01	0.237000E+01	0.121582E+03	0.305821E-01

0.241000E+01 0.121069E+03 0.288111E-01 0.242000E+01 0.120945E+03 0.283996E-01 0.243000E+01 0.120821E+03 0.279996E-01
 0.244000E+01 0.120699E+03 0.276107E-01 0.245000E+01 0.120577E+03 0.27325E-01 0.246000E+01 0.120458E+03 0.268645E-01

DRIFT= 0.558032E+01

NOZZLE NO.	YNOZ	ZNOZ	DIAMN	STDEV	Q	DRIFT
1	0.4000	0.4000	200.0000	50.0000	1.0000	5.5803

SINGLE DROPLET LATERAL DISPLACEMENT

NOZZLE:	YG	1	DIA
0.567768E+00	0.600000E+03		
0.785351E+00	0.300000E+03		
0.889156E+00	0.250000E+03		
0.107252E+01	0.200000E+03		
0.149923E+01	0.150000E+03		
0.191002E+01	0.100300E+03		
0.246069E+01	0.120449E+03		
0.000000E+01	0.000000E+01		
0.000000E+01	0.000000E+01		
0.000000E+01	0.000000E+01		
0.000000E+01	0.000000E+01		
0.000000E+01	0.000000E+01		
0.000000E+01	0.000000E+01		
0.000000E+01	0.000000E+01		
0.000000E+01	0.000000E+01		

FINAL DISTRIBUTION

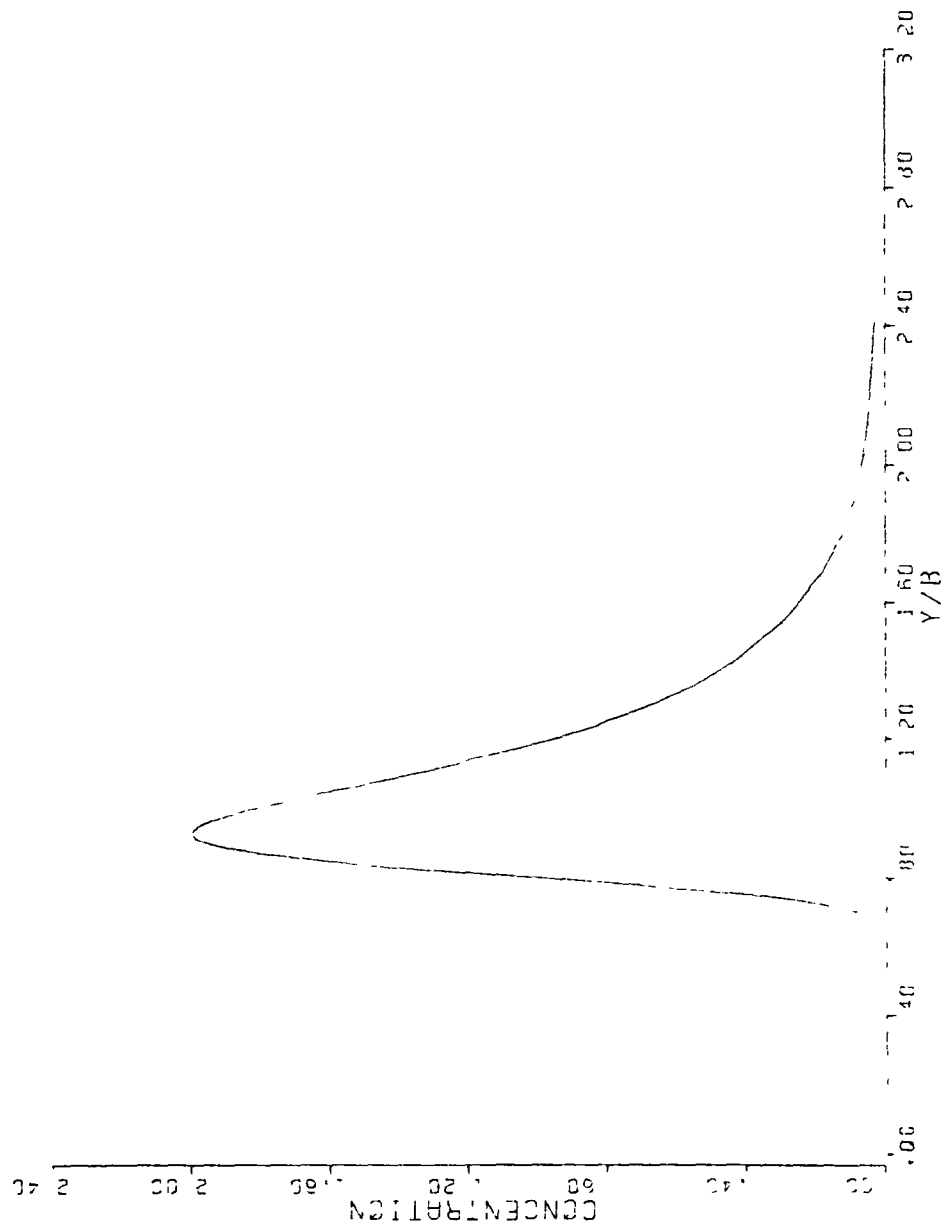
YG	CONCENTRATION	YG	CONCENTRATION	YG	CONCENTRATION	YG	CONCENTRATION	YG	CONCENTRATION
0.64	0.110564E-03	0.65	0.408700E-03	0.66	0.126168E-02	0.67	0.334558E-02	0.68	0.779777E-02
0.69	0.162809E-01	0.70	0.309327E-01	0.71	0.541859E-01	0.72	0.884828E-01	0.73	0.135946E+00
0.74	0.198073E+00	0.75	0.275513E+00	0.76	0.367965E+00	0.77	0.474192E+00	0.78	0.592145E+00
0.79	0.719134E+00	0.80	0.851925E+00	0.81	0.987170E+00	0.82	0.112161E+01	0.83	0.125221E+01
0.84	0.137634E+01	0.85	0.149180E+01	0.86	0.159689E+01	0.87	0.169039E+01	0.88	0.177154E+01
0.89	0.184000E+01	0.90	0.189559E+01	0.91	0.193848E+01	0.92	0.196920E+01	0.93	0.198850E+01
0.94	0.199723E+01	0.95	0.199634E+01	0.96	0.198681E+01	0.97	0.196964E+01	0.98	0.194580E+01
0.99	0.191622E+01	1.00	0.188178E+01	1.01	0.184330E+01	1.02	0.180153E+01	1.03	0.175716E+01
1.04	0.171079E+01	1.05	0.166297E+01	1.06	0.161418E+01	1.07	0.156483E+01	1.08	0.151536E+01
1.09	0.146626E+01	1.10	0.141774E+01	1.11	0.136998E+01	1.12	0.132313E+01	1.13	0.127730E+01
1.14	0.123259E+01	1.15	0.118905E+01	1.16	0.114675E+01	1.17	0.110572E+01	1.18	0.106597E+01
1.19	0.102757E+01	1.20	0.990345E+00	1.21	0.954498E+00	1.22	0.919905E+00	1.23	0.886566E+00
1.24	0.854457E+00	1.25	0.823551E+00	1.26	0.793816E+00	1.27	0.765222E+00	1.28	0.737734E+00
1.29	0.711317E+00	1.30	0.685937E+00	1.31	0.661557E+00	1.32	0.638143E+00	1.33	0.615658E+00
1.34	0.594069E+00	1.35	0.573341E+00	1.36	0.553441E+00	1.37	0.534334E+00	1.38	0.515991E+00
1.39	0.448379E+00	1.40	0.448146E+00	1.41	0.465232E+00	1.42	0.449639E+00	1.43	0.434664E+00
1.44	0.420280E+00	1.45	0.406463E+00	1.46	0.393188E+00	1.47	0.380433E+00	1.48	0.368174E+00
1.49	0.356392E+00	1.50	0.345064E+00	1.51	0.334066E+00	1.52	0.323321E+00	1.53	0.312838E+00
1.54	0.302624E+00	1.55	0.292683E+00	1.56	0.283019E+00	1.57	0.273632E+00	1.58	0.264523E+00
1.59	0.255689E+00	1.60	0.247128E+00	1.61	0.238838E+00	1.62	0.230813E+00	1.63	0.223049E+00
1.64	0.215542E+00	1.65	0.208284E+00	1.66	0.201271E+00	1.67	0.194496E+00	1.68	0.187953E+00
1.69	0.181636E+00	1.70	0.175537E+00	1.71	0.169650E+00	1.72	0.163969E+00	1.73	0.158487E+00
1.74	0.153197E+00	1.75	0.148094E+00	1.76	0.143170E+00	1.77	0.138419E+00	1.78	0.133837E+00
1.79	0.129415E+00	1.80	0.125150E+00	1.81	0.121034E+00	1.82	0.117063E+00	1.83	0.113232E+00
1.84	0.109534E+00	1.85	0.105946E+00	1.86	0.102522E+00	1.87	0.991974E-01	1.88	0.959882E-01
1.89	0.928897E-01	1.90	0.898979E-01	1.91	0.870086E-01	1.92	0.842382E-01	1.93	0.816006E-01
1.94	0.790880E-01	1.95	0.766933E-01	1.96	0.744098E-01	1.97	0.722311E-01	1.98	0.701514E-01
1.99	0.681652E-01	2.00	0.662675E-01	2.01	0.644532E-01	2.02	0.627181E-01	2.03	0.610577E-01
2.04	0.594682E-01	2.05	0.579458E-01	2.06	0.564870E-01	2.07	0.550886E-01	2.08	0.537473E-01
2.09	0.524604E-01	2.10	0.512251E-01	2.11	0.500387E-01	2.12	0.488990E-01	2.13	0.478034E-01
2.14	0.467500E-01	2.15	0.457366E-01	2.16	0.447614E-01	2.17	0.438225E-01	2.18	0.429181E-01
2.19	0.420468E-01	2.20	0.412068E-01	2.21	0.403969E-01	2.22	0.396155E-01	2.23	0.388614E-01
2.24	0.381334E-01	2.25	0.374303E-01	2.26	0.367510E-01	2.27	0.360944E-01	2.28	0.354595E-01
2.29	0.348454E-01	2.30	0.342512E-01	2.31	0.336761E-01	2.32	0.331191E-01	2.33	0.325796E-01
2.34	0.320569E-01	2.35	0.315501E-01	2.36	0.310587E-01	2.37	0.305821E-01	2.38	0.301195E-01
2.39	0.296705E-01	2.40	0.292346E-01	2.41	0.288111E-01	2.42	0.283996E-01	2.43	0.279996E-01
2.44	0.276107E-01	2.45	0.272375E-01	2.46	0.268645E-01				

TOTAL DRIFT= 0.558032E+01

TOTAL= 0.944312E+02

BACKGND: STOP

ASP TEST CASE 2



OSU / AARL

AERIAL SPRAY PROGRAM

LAST UPDATE 2/1/81

ASP TEST CASE 3

PROGRAM CONTROL

N2D= 1	N3D= 0	NEVAP= 1	NPROP2= 0	NCW= 0
NTUN= 0	NOIST= 1	NPRINT= 0	NPLOT= 1	EPS= 0.100000E-04

PARAMETER VALUFS

A= 6.0000	CL= 0.6000	B= 6.0960	U= 53.0350	G= 9.8066
ZY0= 0.8000	Z70= 0.5000	DA= 0.118901E+01	VIS= 0.181330E-04	VCW= 0.0000
D= 0.0000	DD= 1000.0000			

EVAPORATION INPUTS

PA= 101352.10	TDR= 23.888900	TWR= 18.333300
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DISTRIBUTION INPUTS

NCOL= 1	NROW= 6	NDIAMN= 1	NSTDEV= 1	NQ= 1
NZNOZ= 1	DD=1000.00000	X= 0.00000	UD= 0.00000	VD= 0.00000
WD= 0.00000	SWIDTH= 4.00000	DEFF= 1.00000	DWIDTH= 5.00000	

NOZZLE NO.	YNOZ	ZNOZ	DIAMN	STDEV	Q	DRIFT
1	0.4000	0.4000	200.0000	50.0000	1.0000	21.2474

SINGLE DROPLET LATERAL DISPLACEMENT

NOZZLE:	1	
YG	UIA	DIAG
0.566534E+00	0.600000E+03	0.598786E+03
0.786095E+00	0.300000E+03	0.297565E+03
0.892304E+00	0.250000E+03	0.247031E+03
0.108410E+01	0.200000E+03	0.196103E+03
0.155559E+01	0.150000E+03	0.143102E+03
0.204485E+01	0.131844E+03	0.119342E+03
0.235994E+01	0.127329E+03	0.109949E+03
0.000000E+01	0.000000E+01	0.000000E+01
0.000000E+01	0.000000E+01	0.000000E+01
0.000000E+01	0.000000E+01	0.000000E+01
0.000000E+01	0.000000E+01	0.000000E+01
0.000000E+01	0.000000E+01	0.000000E+01
0.000000E+01	0.000000E+01	0.000000E+01
0.000000E+01	0.000000E+01	0.000000E+01
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FINAL DISTRIBUTION

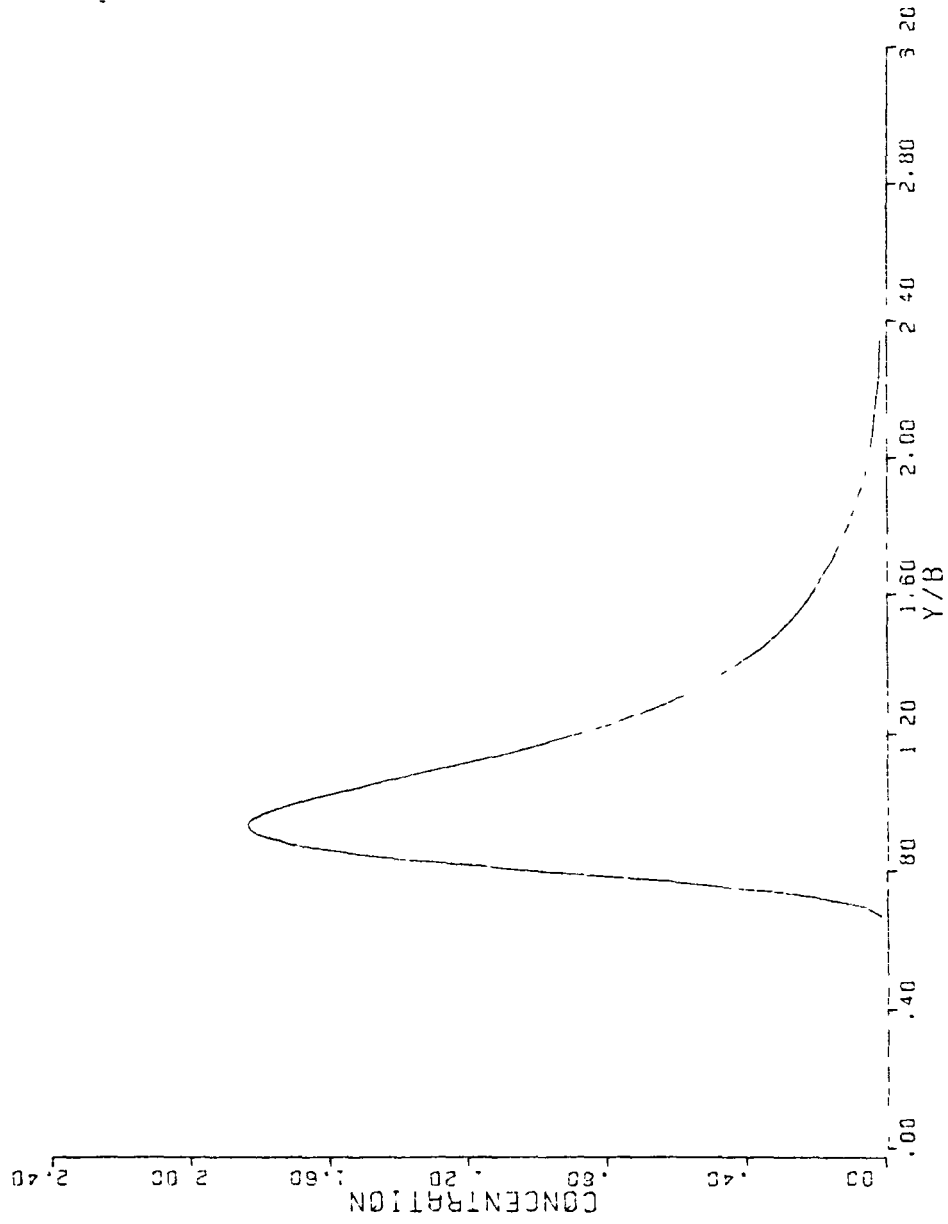
YG	CONCENTRATION	YG	CONCENTRATION	YG	CONCENTRATION	YG	CONCENTRATION	YG	CONCENTRATION
0.64	0.124109E-03	0.65	0.445961E-03	0.66	0.134425E-02	0.67	0.349339E-02	0.68	0.800428E-02
0.69	0.164709E-01	0.70	0.309088E-01	0.71	0.535747E-01	0.72	0.866979E-01	0.73	0.132177E+00
0.74	0.191308E+00	0.75	0.264595E+00	0.76	0.351662E+00	0.77	0.451289E+00	0.78	0.561524E+00
0.79	0.679842E+00	0.80	0.803113E+00	0.81	0.928154E+00	0.82	0.105193E+01	0.83	0.117164E+01
0.84	0.128490E+01	0.85	0.138974E+01	0.86	0.148466E+01	0.87	0.156864E+01	0.88	0.164107E+01
0.89	0.170174E+01	0.90	0.175078E+01	0.91	0.178863E+01	0.92	0.181580E+01	0.93	0.183294E+01
0.94	0.184078E+01	0.95	0.184014E+01	0.96	0.183185E+01	0.97	0.181677E+01	0.98	0.179573E+01
0.99	0.176951E+01	1.00	0.173887E+01	1.01	0.170452E+01	1.02	0.166711E+01	1.03	0.162724E+01
1.04	0.158543E+01	1.05	0.154217E+01	1.06	0.149790E+01	1.07	0.145297E+01	1.08	0.140773E+01
1.09	0.136244E+01	1.10	0.131730E+01	1.11	0.127254E+01	1.12	0.122836E+01	1.13	0.118492E+01
1.14	0.114237E+01	1.15	0.110081E+01	1.16	0.106032E+01	1.17	0.102097E+01	1.18	0.982809E+00
1.19	0.945863E+00	1.20	0.910151E+00	1.21	0.875679E+00	1.22	0.842445E+00	1.23	0.810440E+00
1.24	0.779645E+00	1.25	0.750041E+00	1.26	0.721601E+00	1.27	0.694295E+00	1.28	0.668093E+00
1.29	0.642961E+00	1.30	0.618865E+00	1.31	0.595770E+00	1.32	0.573638E+00	1.33	0.552436E+00
1.34	0.532126E+00	1.35	0.512674E+00	1.36	0.494045E+00	1.37	0.476205E+00	1.38	0.459121E+00
1.39	0.442760E+00	1.40	0.427091E+00	1.41	0.412084E+00	1.42	0.397710E+00	1.43	0.383941E+00
1.44	0.370749E+00	1.45	0.358108E+00	1.46	0.345994E+00	1.47	0.334383E+00	1.48	0.323252E+00
1.49	0.312579E+00	1.50	0.302342E+00	1.51	0.292523E+00	1.52	0.283103E+00	1.53	0.274062E+00
1.54	0.265384E+00	1.55	0.257052E+00	1.56	0.249039E+00	1.57	0.241244E+00	1.58	0.233645E+00
1.59	0.226239E+00	1.60	0.219023E+00	1.61	0.211994E+00	1.62	0.205151E+00	1.63	0.198490E+00
1.64	0.192008E+00	1.65	0.185704E+00	1.66	0.179573E+00	1.67	0.173614E+00	1.68	0.167824E+00
1.69	0.162200E+00	1.70	0.156739E+00	1.71	0.151439E+00	1.72	0.146296E+00	1.73	0.141308E+00
1.74	0.136471E+00	1.75	0.131783E+00	1.76	0.127240E+00	1.77	0.122840E+00	1.78	0.118579E+00
1.79	0.114455E+00	1.80	0.110463E+00	1.81	0.106601E+00	1.82	0.102866E+00	1.83	0.992536E-01
1.84	0.957618E-01	1.85	0.923869E-01	1.86	0.891259E-01	1.87	0.859754E-01	1.88	0.829325E-01
1.89	0.799939E-01	1.90	0.771566E-01	1.91	0.744176E-01	1.92	0.717737E-01	1.93	0.692221E-01
1.94	0.667599E-01	1.95	0.643842E-01	1.96	0.620922E-01	1.97	0.598811E-01	1.98	0.577482E-01
1.99	0.556909E-01	2.00	0.537067E-01	2.01	0.517930E-01	2.02	0.499474E-01	2.03	0.481675E-01
2.04	0.464509E-01	2.05	0.447997E-01	2.06	0.432341E-01	2.07	0.417527E-01	2.08	0.403504E-01
2.09	0.390227E-01	2.10	0.377650E-01	2.11	0.365733E-01	2.12	0.354437E-01	2.13	0.343724E-01
2.14	0.333562E-01	2.15	0.323918E-01	2.16	0.314761E-01	2.17	0.306064E-01	2.18	0.297799E-01
2.19	0.289942E-01	2.20	0.282470E-01	2.21	0.275361E-01	2.22	0.268593E-01	2.23	0.262148E-01
2.24	0.256007E-01	2.25	0.250153E-01	2.26	0.244571E-01	2.27	0.239245E-01	2.28	0.234160E-01
2.29	0.229304E-01	2.30	0.224664E-01	2.31	0.220229E-01	2.32	0.215986E-01	2.33	0.211926E-01
2.34	0.208038E-01	2.35	0.204314E-01						

TOTAL DRIFT= 0.212474E+02

TOTAL= 0.860481E+02

BACKGND: STOP

ASP TEST CASE 3



LAST UPDATE 2/1/81

PROGRAM CONTROL

PARAMETER VALUES

DISTRIBUTION INPUTS

NOZZLE NO.	YNOZ	ZNOZ	DIAMN	STDEV	Q	DRIFT
1	0.4000	0.4000	200.0000	50.0000	1.0000	5.5777

SINGLE DROPLET LATERAL DISPLACEMENT

[illegible]

FINAL DISTRIBUTION

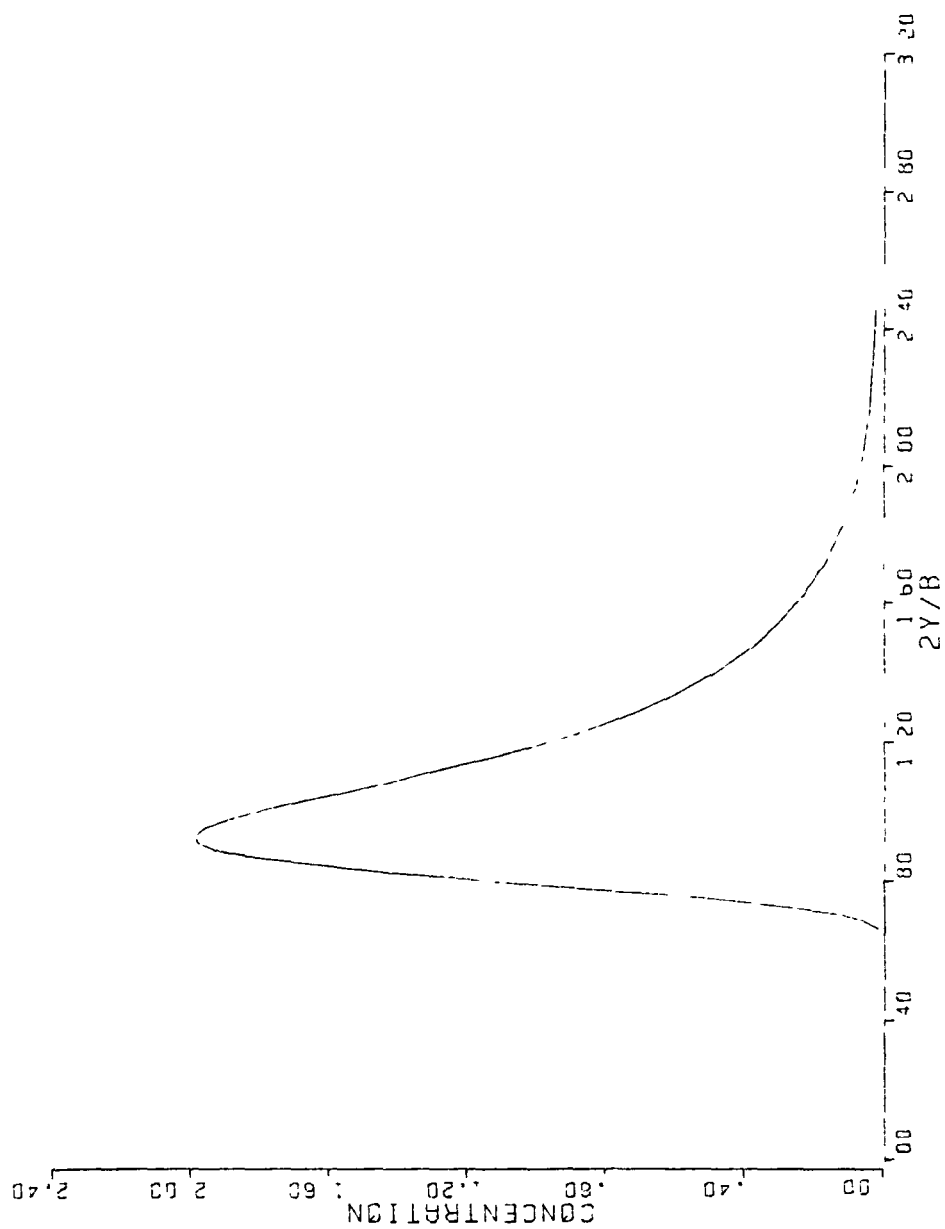
YG	CONCENTRATION	YG	CONCENTRATION	YG	CONCENTRATION	YG	CONCENTRATION	YG	CONCENTRATION
0.64	0.117847E-03	0.65	0.431420E-03	0.66	0.132101E-02	0.67	0.347890E-02	0.68	0.806143E-02
0.69	0.167483E-01	0.70	0.316871E-01	0.71	0.553080E-01	0.72	0.900375E-01	0.73	0.137970E+00
0.74	0.200565E+00	0.75	0.278435E+00	0.76	0.371242E+00	0.77	0.477720E+00	0.78	0.595802E+00
0.79	0.722783E+00	0.80	0.855396E+00	0.81	0.990285E+00	0.82	0.112419E+01	0.83	0.125410E+01
0.84	0.137738E+01	0.85	0.149188E+01	0.86	0.159591E+01	0.87	0.168829E+01	0.88	0.176829E+01
0.89	0.183559E+01	0.90	0.189011E+01	0.91	0.193209E+01	0.92	0.196208E+01	0.93	0.198083E+01
0.94	0.198918E+01	0.95	0.198809E+01	0.96	0.197851E+01	0.97	0.196144E+01	0.98	0.193781E+01
0.99	0.190854E+01	1.00	0.187451E+01	1.01	0.183650E+01	1.02	0.179526E+01	1.03	0.175145E+01
1.04	0.170567E+01	1.05	0.165845E+01	1.06	0.161025E+01	1.07	0.156150E+01	1.08	0.151260E+01
1.09	0.146403E+01	1.10	0.141601E+01	1.11	0.136872E+01	1.12	0.132230E+01	1.13	0.127688E+01
1.14	0.123253E+01	1.15	0.118933E+01	1.16	0.114733E+01	1.17	0.110657E+01	1.18	0.106707E+01
1.19	0.102884E+01	1.20	0.991886E+00	1.21	0.956193E+00	1.22	0.921753E+00	1.23	0.888547E+00
1.24	0.856552E+00	1.25	0.825743E+00	1.26	0.796091E+00	1.27	0.767564E+00	1.28	0.740131E+00
1.29	0.713757E+00	1.30	0.688409E+00	1.31	0.664052E+00	1.32	0.640652E+00	1.33	0.618174E+00
1.34	0.596504E+00	1.35	0.575848E+00	1.36	0.555935E+00	1.37	0.536811E+00	1.38	0.518445E+00
1.39	0.500807E+00	1.40	0.483868E+00	1.41	0.467598E+00	1.42	0.451970E+00	1.43	0.436958E+00
1.44	0.422535E+00	1.45	0.408677E+00	1.46	0.395361E+00	1.47	0.382563E+00	1.48	0.370261E+00
1.49	0.358435E+00	1.50	0.347063E+00	1.51	0.336030E+00	1.52	0.325237E+00	1.53	0.314697E+00
1.54	0.304418E+00	1.55	0.294405E+00	1.56	0.284664E+00	1.57	0.275197E+00	1.58	0.266004E+00
1.59	0.257084E+00	1.60	0.248437E+00	1.61	0.240058E+00	1.62	0.231946E+00	1.63	0.224095E+00
1.64	0.216501E+00	1.65	0.209158E+00	1.66	0.202061E+00	1.67	0.195204E+00	1.68	0.188580E+00
1.69	0.182184E+00	1.70	0.176009E+00	1.71	0.170048E+00	1.72	0.164295E+00	1.73	0.158743E+00
1.74	0.153386E+00	1.75	0.148218E+00	1.76	0.143231E+00	1.77	0.138420E+00	1.78	0.133779E+00
1.79	0.129302E+00	1.80	0.124982E+00	1.81	0.120815E+00	1.82	0.116794E+00	1.83	0.112915E+00
1.84	0.109171E+00	1.85	0.105559E+00	1.86	0.102073E+00	1.87	0.987078E-01	1.88	0.954596E-01
1.89	0.923239E-01	1.90	0.892963E-01	1.91	0.863727E-01	1.92	0.835671E-01	1.93	0.808973E-01
1.94	0.783554E-01	1.95	0.759341E-01	1.96	0.736265E-01	1.97	0.714261E-01	1.98	0.693269E-01
1.99	0.673232E-01	2.00	0.654097E-01	2.01	0.635815E-01	2.02	0.618341E-01	2.03	0.601629E-01
2.04	0.585640E-01	2.05	0.570334E-01	2.06	0.555678E-01	2.07	0.541635E-01	2.08	0.528176E-01
2.09	0.515269E-01	2.10	0.502887E-01	2.11	0.491003E-01	2.12	0.479593E-01	2.13	0.468633E-01
2.14	0.458100E-01	2.15	0.447975E-01	2.16	0.438237E-01	2.17	0.428867E-01	2.18	0.419848E-01
2.19	0.411164E-01	2.20	0.402798E-01	2.21	0.394737E-01	2.22	0.386965E-01	2.23	0.379469E-01
2.24	0.372237E-01	2.25	0.365258E-01	2.26	0.358519E-01	2.27	0.352009E-01	2.28	0.345720E-01
2.29	0.339640E-01	2.30	0.333761E-01	2.31	0.328075E-01	2.32	0.322572E-01	2.33	0.317246E-01
2.34	0.312088E-01	2.35	0.307091E-01	2.36	0.302249E-01	2.37	0.297556E-01	2.38	0.293004E-01
2.39	0.288590E-01	2.40	0.284305E-01	2.41	0.280147E-01	2.42	0.276109E-01	2.43	0.272187E-01
2.44	0.268375E-01	2.45	0.264671E-01	2.46	0.261069E-01	2.47	0.257566E-01	2.48	0.254158E-01

TOTAL DRIFT= 0.557706E+01

TOTAL= 0.944302E+02

BACKGND: STOP

ASP TEST CASE 4



OSU / AARL
AERIAL SPRAY PROGRAM
LAST UPDATE 2/1/81

ASP TEST CASE 5

PROGRAM CONTROL

N2D= 0 N3D= 1 NEVAP= 1 NPROP2= 1 NCW= 1
NTUN= 0 NDIST= 1 NPRINT= 0 NPLOT= 1 EPS= 0.100000E-04

PARAMETER VALUES

A= 6.0000 CL= 0.6000 B= 6.0960 U= 53.0350 G= 9.8066
ZY0= 0.8000 ZZO= 0.5000 OA= 0.118901E+01 VIS= 0.181330E-04 VCW= 1.0000
D= 0.0000 DD= 1000.0000

EVAPORATION INPUTS

PA= 101352.10 TOR= 23.888900 TWR= 18.333300

PROP2 INPUTS

ZPROP2= 0.5000 PK1= 1.7930 PK2= 17.9300 PK3= 0.0000 PK4= 6.0290
RP= 0.2000

DISTRIBUTION INPUTS

NCOL= 3 NROW= 6 NDIAMN= 1 NSTDEV= 1 NQ= 1
NZNOZ= 1 ND=1000.00000 X= 0.00000 UD= 0.00000 VD= 0.00000
WD= 0.00000 SWIDTH= 4.00000 UFRR= 1.00000 DWIDTH= 5.00000

NOZZLE NO.	YNOZ	ZNOZ	DIAMN	STDEV	Q	DRIFT
1	-0.3000	0.4000	200.0000	50.0000	1.0000	7.1442
2	0.0000	0.4000	200.0000	50.0000	1.0000	13.0321
3	0.3000	0.4000	200.0000	50.0000	1.0000	9.2220

SINGLE DROPLET LATERAL DISPLACEMENT

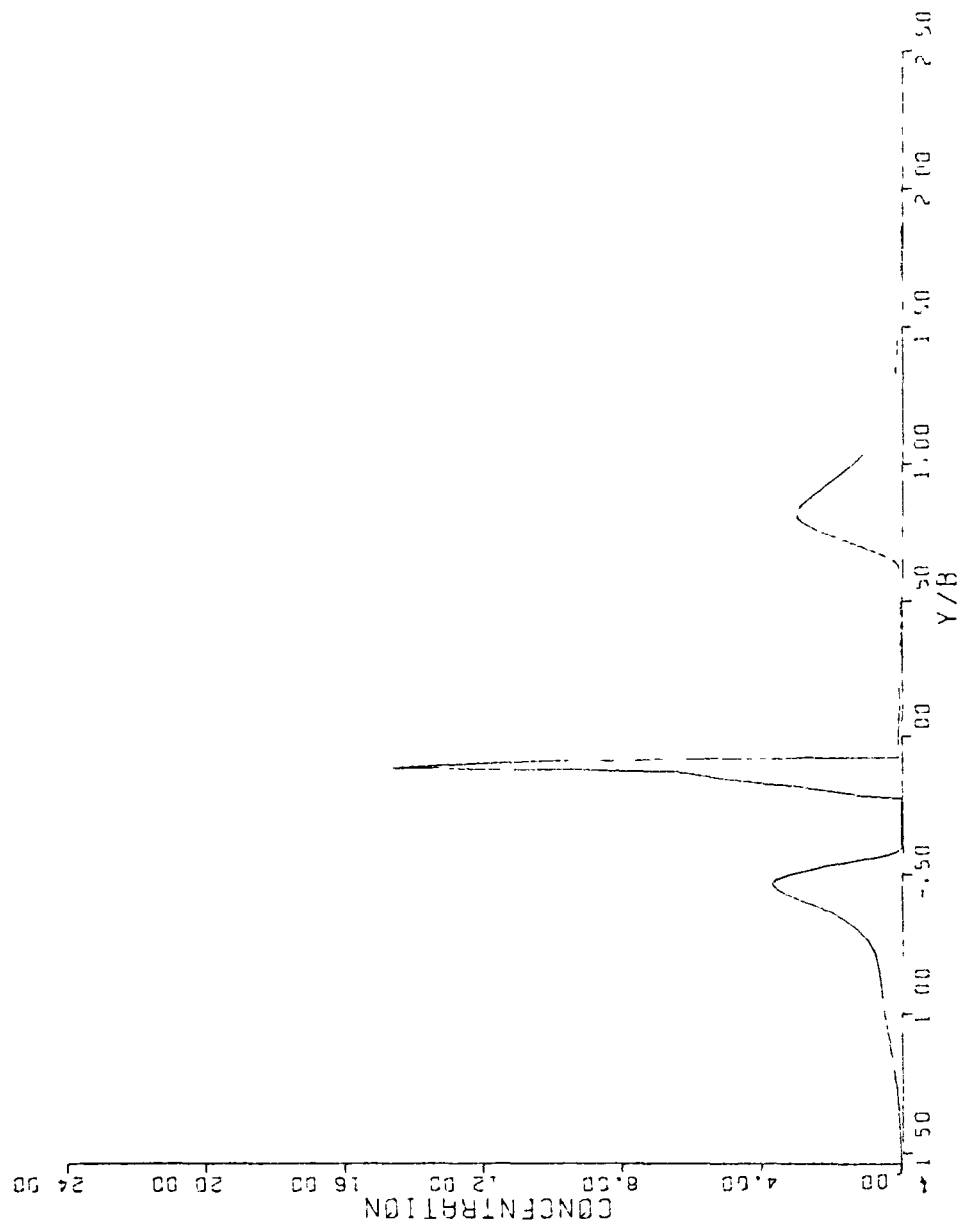
NOZZLE:	1			2			3	
YG	DIA	DIAG	YG	DIA	DIAG	YG	DIA	DIAG
-0.320151E+00	0.600000E+03	0.598080E+03	-0.876441E-01	0.600000E+03	0.597780E+03	0.544745E+00	0.600000E+03	0.598063E+03
-0.407045E+00	0.300000E+03	0.297088E+03	-0.515855E-01	0.300000E+03	0.296151E+03	0.708199E+00	0.300000E+03	0.297044E+03
-0.459665E+00	0.250000E+03	0.246594E+03	-0.539186E-01	0.275000E+03	0.270796E+03	0.781975E+00	0.250000E+03	0.246526E+03
-0.557848E+00	0.200000E+03	0.195780E+03	-0.593184E-01	0.250000E+03	0.245340E+03	0.896006E+00	0.200000E+03	0.195665E+03
-0.783947E+00	0.150000E+03	0.144136E+03	-0.842100E-01	0.200000E+03	0.193927E+03	0.109198E+01	0.150000E+03	0.144039E+03
-0.147024E+01	0.100000E+03	0.740140E+02	-0.139088E+00	0.150000E+03	0.140996E+03	0.152290E+01	0.100000E+03	0.900272E+02
-0.151925E+01	0.984375E+02	0.632314E+02	-0.172918E+00	0.125000E+03	0.112875E+03	0.209856E+01	0.770271E+02	0.613805E+02
0.000000E+01	0.000000E+01	0.000000E+01	0.321932E+00	0.100000E+03	0.110852E+02	0.000000E+01	0.000000E+01	0.000000E+01
0.000000E+01	0.000000E+01	0.000000E+01	0.325157E+00	0.725000E+02	0.451472E+02	0.000000E+01	0.000000E+01	0.000000E+01
0.000000E+01	0.000000E+01	0.000000E+01	0.704499E-01	0.690625E+02	0.157827E+02	0.000000E+01	0.000000E+01	0.000000E+01
0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01
0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01
0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01
0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01
0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01
0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01
0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01	0.000000E+01

FINAL DISTRIBUTION

YG	CONCENTRATION	YG	CONCENTRATION	YG	CONCENTRATION	YG	CONCENTRATION	YG	CONCENTRATION
-1.51	0.919440E-02	-1.50	0.103502E-01	-1.49	0.116264E-01	-1.48	0.130289E-01	-1.47	0.145629E-01
-1.46	0.162340E-01	-1.45	0.180510E-01	-1.44	0.200232E-01	-1.43	0.221601E-01	-1.42	0.244715E-01
-1.41	0.269678E-01	-1.40	0.296594E-01	-1.39	0.325572E-01	-1.38	0.356725E-01	-1.37	0.390167E-01
-1.36	0.426016E-01	-1.35	0.464390E-01	-1.34	0.505412E-01	-1.33	0.549204E-01	-1.32	0.595890E-01
-1.31	0.645596E-01	-1.30	0.698445E-01	-1.29	0.754561E-01	-1.28	0.814066E-01	-1.27	0.877078E-01
-1.26	0.943714E-01	-1.25	0.101408E+00	-1.24	0.108829E+00	-1.23	0.116644E+00	-1.22	0.124861E+00
-1.21	0.133489E+00	-1.20	0.142533E+00	-1.19	0.152000E+00	-1.18	0.161892E+00	-1.17	0.172212E+00
-1.16	0.182959E+00	-1.15	0.194131E+00	-1.14	0.205723E+00	-1.13	0.217726E+00	-1.12	0.230131E+00
-1.11	0.242943E+00	-1.10	0.256085E+00	-1.09	0.269595E+00	-1.08	0.283428E+00	-1.07	0.297556E+00
-1.06	0.311945E+00	-1.05	0.326556E+00	-1.04	0.341349E+00	-1.03	0.356277E+00	-1.02	0.371289E+00
-1.01	0.386331E+00	-1.00	0.401346E+00	-0.99	0.416273E+00	-0.98	0.431049E+00	-0.97	0.445610E+00
-0.96	0.459892E+00	-0.95	0.473831E+00	-0.94	0.487368E+00	-0.93	0.500447E+00	-0.92	0.513020E+00
-0.91	0.525048E+00	-0.90	0.536505E+00	-0.89	0.547382E+00	-0.88	0.557687E+00	-0.87	0.567455E+00
-0.86	0.576747E+00	-0.85	0.585659E+00	-0.84	0.594326E+00	-0.83	0.602928E+00	-0.82	0.611698E+00
-0.81	0.620931E+00	-0.80	0.630991E+00	-0.79	0.642328E+00	-0.78	0.655514E+00	-0.77	0.671325E+00
-0.76	0.690124E+00	-0.75	0.712217E+00	-0.74	0.737946E+00	-0.73	0.767698E+00	-0.72	0.801901E+00
-0.71	0.841038E+00	-0.70	0.885644E+00	-0.69	0.936312E+00	-0.68	0.993698E+00	-0.67	0.105852E+01
-0.66	0.113158E+01	-0.65	0.121374E+01	-0.64	0.130592E+01	-0.63	0.140916E+01	-0.62	0.152453E+01
-0.61	0.165320E+01	-0.60	0.179643E+01	-0.59	0.195549E+01	-0.58	0.213168E+01	-0.57	0.232615E+01
-0.56	0.253965E+01	-0.55	0.276672E+01	-0.54	0.299187E+01	-0.53	0.320669E+01	-0.52	0.340104E+01
-0.51	0.356281E+01	-0.50	0.367833E+01	-0.49	0.373321E+01	-0.48	0.371375E+01	-0.47	0.360894E+01
-0.46	0.341294E+01	-0.45	0.313619E+01	-0.44	0.279241E+01	-0.43	0.238913E+01	-0.42	0.194197E+01
-0.41	0.147452E+01	-0.40	0.100968E+01	-0.39	0.581835E+00	-0.38	0.260214E+00	-0.37	0.796776E-01
-0.36	0.136499E-01	-0.35	0.938853E-03	-0.17	0.168032E+01	-0.16	0.221909E+01	-0.15	0.265141E+01
-0.14	0.302279E+01	-0.13	0.430308E+01	-0.12	0.496400E+01	-0.11	0.561741E+01	-0.10	0.618396E+01
-0.09	0.657872E+01	-0.08	0.146893E+02	-0.07	0.128646E+02	-0.06	0.969981E+01	-0.05	0.625907E-01
-0.04	0.601423E-01	-0.03	0.577604E-01	-0.02	0.554479E-01	-0.01	0.532070E-01	0.00	0.510397E-01
0.01	0.489476E-01	0.02	0.469318E-01	0.03	0.449931E-01	0.04	0.431322E-01	0.05	0.413491E-01
0.06	0.396439E-01	0.07	0.380163E-01	0.08	0.365174E-01	0.09	0.350553E-01	0.10	0.336704E-01
0.11	0.323619E-01	0.12	0.311268E-01	0.13	0.299698E-01	0.14	0.288839E-01	0.15	0.278698E-01
0.16	0.269201E-01	0.17	0.260517E-01	0.18	0.252451E-01	0.19	0.245051E-01	0.20	0.238307E-01
0.21	0.232205E-01	0.22	0.226735E-01	0.23	0.221889E-01	0.24	0.217659E-01	0.25	0.214037E-01
0.26	0.211019E-01	0.27	0.208601E-01	0.28	0.206784E-01	0.29	0.205569E-01	0.30	0.204959E-01
0.31	0.204901E-01	0.32	0.205586E-01	0.61	0.398603E-03	0.62	0.188731E-02	0.63	0.691297E-02
0.64	0.204101E-01	0.65	0.502728E-01	0.66	0.106330E+00	0.67	0.197834E+00	0.68	0.330412E+00
0.69	0.503837E+00	0.70	0.711515E+00	0.71	0.941824E+00	0.72	0.118517E+01	0.73	0.143373E+01
0.74	0.167951E+01	0.75	0.191594E+01	0.76	0.213787E+01	0.77	0.234145E+01	0.78	0.252389E+01
0.79	0.268062E+01	0.80	0.280446E+01	0.81	0.289454E+01	0.82	0.295156E+01	0.83	0.297733E+01
0.84	0.297448E+01	0.85	0.294617E+01	0.86	0.289584E+01	0.87	0.282701E+01	0.88	0.274313E+01
0.89	0.264744E+01	0.90	0.254306E+01	0.91	0.243368E+01	0.92	0.232165E+01	0.93	0.220864E+01
0.94	0.209605E+01	0.95	0.198500E+01	0.96	0.187638E+01	0.97	0.177087E+01	0.98	0.166897E+01
0.99	0.157106E+01	1.00	0.147778E+01	1.01	0.138806E+01	1.02	0.130318E+01	1.03	0.122273E+01
1.04	0.114665E+01	1.05	0.107486E+01	1.06	0.100723E+01	1.07	0.943617E+00	1.08	0.883859E+00
1.09	0.827786E+00	1.10	0.775062E+00	1.11	0.725256E+00	1.12	0.678293E+00	1.13	0.634094E+00
1.14	0.592563E+00	1.15	0.543596E+00	1.16	0.517082E+00	1.17	0.482905E+00	1.18	0.450947E+00
1.19	0.421089E+00	1.20	0.393215E+00	1.21	0.367209E+00	1.22	0.342959E+00	1.23	0.320356E+00
1.24	0.299298E+00	1.25	0.279683E+00	1.26	0.261418E+00	1.27	0.244412E+00	1.28	0.228580E+00
1.29	0.213842E+00	1.30	0.200122E+00	1.31	0.187350E+00	1.32	0.175459E+00	1.33	0.164388E+00
1.34	0.154078E+00	1.35	0.144474E+00	1.36	0.135528E+00	1.37	0.127192E+00	1.38	0.119422E+00
1.39	0.112178E+00	1.40	0.105422E+00	1.41	0.991196E-01	1.42	0.932381E-01	1.43	0.877476E-01
1.44	0.826402E-01	1.45	0.778301E-01	1.46	0.733534E-01	1.47	0.691679E-01	1.48	0.652532E-01
1.49	0.615902E-01	1.50	0.581613E-01	1.51	0.549501E-01	1.52	0.519417E-01	1.53	0.491255E-01
1.54	0.464975E-01	1.55	0.440436E-01	1.56	0.417504E-01	1.57	0.396055E-01	1.58	0.375977E-01
1.59	0.357167E-01	1.60	0.339532E-01	1.61	0.322984E-01	1.62	0.307446E-01	1.63	0.292843E-01
1.64	0.279110E-01	1.65	0.266185E-01	1.66	0.254011E-01	1.67	0.242538E-01	1.68	0.231716E-01

1.73	0.174118E-01	1.76	0.163973E-01	1.79	0.157389E-01	1.82	0.151139E-01	1.85	0.145204E-01	1.88	0.139563E-01	1.91	0.134200E-01	1.94	0.129097E-01	1.97	0.124241E-01	2.00	0.119615E-01	2.03	0.115208E-01	2.06	0.111006E-01	2.09	0.106998E-01	2.12	0.103174E-01	2.15	0.099522E-02	2.18	0.096034E-02	2.21	0.092701E-02	2.24	0.895150E-02	2.27	0.875215E-02	2.30	0.855215E-02	2.33	0.835502E-02	2.36	0.817580E-02	2.39	0.800837E-02	2.42	0.784512E-02	2.45	0.768451E-02	2.48	0.752831E-02	2.51	0.737559E-02	2.54	0.722624E-02	2.57	0.708083E-02	2.60	0.693837E-02	2.63	0.679883E-02	2.66	0.666228E-02	2.69	0.652873E-02	2.72	0.639808E-02	2.75	0.627023E-02	2.78	0.614518E-02	2.81	0.602282E-02	2.84	0.590316E-02	2.87	0.578619E-02	2.90	0.567182E-02	2.93	0.555995E-02	2.96	0.545059E-02	2.99	0.534368E-02	3.02	0.523921E-02	3.05	0.513718E-02	3.08	0.503750E-02	3.11	0.494007E-02	3.14	0.484489E-02	3.17	0.475196E-02	3.20	0.466118E-02	3.23	0.457255E-02	3.26	0.448607E-02	3.29	0.440174E-02	3.32	0.431955E-02	3.35	0.423950E-02	3.38	0.416159E-02	3.41	0.408582E-02	3.44	0.401219E-02	3.47	0.394070E-02	3.50	0.387135E-02	3.53	0.380414E-02	3.56	0.373907E-02	3.59	0.367614E-02	3.62	0.361535E-02	3.65	0.355670E-02	3.68	0.349919E-02	3.71	0.344282E-02	3.74	0.338850E-02	3.77	0.333523E-02	3.80	0.328301E-02	3.83	0.323184E-02	3.86	0.318172E-02	3.89	0.313265E-02	3.92	0.308463E-02	3.95	0.303766E-02	3.98	0.299174E-02	4.01	0.294687E-02	4.04	0.290305E-02	4.07	0.285928E-02	4.10	0.281656E-02	4.13	0.277489E-02	4.16	0.273327E-02	4.19	0.269170E-02	4.22	0.265018E-02	4.25	0.260871E-02	4.28	0.256829E-02	4.31	0.252792E-02	4.34	0.248760E-02	4.37	0.244733E-02	4.40	0.240711E-02	4.43	0.236694E-02	4.46	0.232682E-02	4.49	0.228675E-02	4.52	0.224673E-02	4.55	0.220676E-02	4.58	0.216684E-02	4.61	0.212697E-02	4.64	0.208715E-02	4.67	0.204738E-02	4.70	0.200766E-02	4.73	0.196799E-02	4.76	0.192837E-02	4.79	0.188880E-02	4.82	0.184928E-02	4.85	0.180981E-02	4.88	0.177039E-02	4.91	0.173102E-02	4.94	0.169170E-02	4.97	0.165243E-02	5.00	0.161321E-02	5.03	0.157404E-02	5.06	0.153492E-02	5.09	0.149585E-02	5.12	0.145683E-02	5.15	0.141786E-02	5.18	0.137894E-02	5.21	0.133997E-02	5.24	0.130105E-02	5.27	0.126218E-02	5.30	0.122336E-02	5.33	0.118459E-02	5.36	0.114587E-02	5.39	0.110720E-02	5.42	0.106858E-02	5.45	0.102999E-02	5.48	0.099145E-02	5.51	0.095295E-02	5.54	0.091450E-02	5.57	0.087609E-02	5.60	0.083773E-02	5.63	0.079942E-02	5.66	0.076116E-02	5.69	0.072295E-02	5.72	0.068479E-02	5.75	0.064668E-02	5.78	0.060862E-02	5.81	0.057061E-02	5.84	0.053265E-02	5.87	0.049474E-02	5.90	0.045688E-02	5.93	0.041907E-02	5.96	0.038131E-02	5.99	0.034360E-02	6.02	0.030594E-02	6.05	0.026833E-02	6.08	0.023077E-02	6.11	0.019326E-02	6.14	0.015580E-02	6.17	0.011839E-02	6.20	0.008103E-02	6.23	0.004372E-02	6.26	0.000646E-02	6.29	0.000000E-02	6.32	0.000000E-02	6.35	0.000000E-02	6.38	0.000000E-02	6.41	0.000000E-02	6.44	0.000000E-02	6.47	0.000000E-02	6.50	0.000000E-02	6.53	0.000000E-02	6.56	0.000000E-02	6.59	0.000000E-02	6.62	0.000000E-02	6.65	0.000000E-02	6.68	0.000000E-02	6.71	0.000000E-02	6.74	0.000000E-02	6.77	0.000000E-02	6.80	0.000000E-02	6.83	0.000000E-02	6.86	0.000000E-02	6.89	0.000000E-02	6.92	0.000000E-02	6.95	0.000000E-02	6.98	0.000000E-02	7.01	0.000000E-02	7.04	0.000000E-02	7.07	0.000000E-02	7.10	0.000000E-02	7.13	0.000000E-02	7.16	0.000000E-02	7.19	0.000000E-02	7.22	0.000000E-02	7.25	0.000000E-02	7.28	0.000000E-02	7.31	0.000000E-02	7.34	0.000000E-02	7.37	0.000000E-02	7.40	0.000000E-02	7.43	0.000000E-02	7.46	0.000000E-02	7.49	0.000000E-02	7.52	0.000000E-02	7.55	0.000000E-02	7.58	0.000000E-02	7.61	0.000000E-02	7.64	0.000000E-02	7.67	0.000000E-02	7.70	0.000000E-02	7.73	0.000000E-02	7.76	0.000000E-02	7.79	0.000000E-02	7.82	0.000000E-02	7.85	0.000000E-02	7.88	0.000000E-02	7.91	0.000000E-02	7.94	0.000000E-02	7.97	0.000000E-02	8.00	0.000000E-02	8.03	0.000000E-02	8.06	0.000000E-02	8.09	0.000000E-02	8.12	0.000000E-02	8.15	0.000000E-02	8.18	0.000000E-02	8.21	0.000000E-02	8.24	0.000000E-02	8.27	0.000000E-02	8.30	0.000000E-02	8.33	0.000000E-02	8.36	0.000000E-02	8.39	0.000000E-02	8.42	0.000000E-02	8.45	0.000000E-02	8.48	0.000000E-02	8.51	0.000000E-02	8.54	0.000000E-02	8.57	0.000000E-02	8.60	0.000000E-02	8.63	0.000000E-02	8.66	0.000000E-02	8.69	0.000000E-02	8.72	0.000000E-02	8.75	0.000000E-02	8.78	0.000000E-02	8.81	0.000000E-02	8.84	0.000000E-02	8.87	0.000000E-02	8.90	0.000000E-02	8.93	0.000000E-02	8.96	0.000000E-02	8.99	0.000000E-02	9.02	0.000000E-02	9.05	0.000000E-02	9.08	0.000000E-02	9.11	0.000000E-02	9.14	0.000000E-02	9.17	0.000000E-02	9.20	0.000000E-02	9.23	0.000000E-02	9.26	0.000000E-02	9.29	0.000000E-02	9.32	0.000000E-02	9.35	0.000000E-02	9.38	0.000000E-02	9.41	0.000000E-02	9.44	0.000000E-02	9.47	0.000000E-02	9.50	0.000000E-02	9.53	0.000000E-02	9.56	0.000000E-02	9.59	0.000000E-02	9.62	0.000000E-02	9.65	0.000000E-02	9.68	0.000000E-02	9.71	0.000000E-02	9.74	0.000000E-02	9.77	0.000000E-02	9.80	0.000000E-02	9.83	0.000000E-02	9.86	0.000000E-02	9.89	0.000000E-02	9.92	0.000000E-02	9.95	0.000000E-02	9.98	0.000000E-02	10.01	0.000000E-02	10.04	0.000000E-02	10.07	0.000000E-02	10.10	0.000000E-02	10.13	0.000000E-02	10.16	0.000000E-02	10.19	0.000000E-02	10.22	0.000000E-02	10.25	0.000000E-02	10.28	0.000000E-02	10.31	0.000000E-02	10.34	0.000000E-02	10.37	0.000000E-02	10.40	0.000000E-02	10.43	0.000000E-02	10.46	0.000000E-02	10.49	0.000000E-02	10.52	0.000000E-02	10.55	0.000000E-02	10.58	0.000000E-02	10.61	0.000000E-02	10.64	0.000000E-02	10.67	0.000000E-02	10.70	0.000000E-02	10.73	0.000000E-02	10.76	0.000000E-02	10.79	0.000000E-02	10.82	0.000000E-02	10.85	0.000000E-02	10.88	0.000000E-02	10.91	0.000000E-02	10.94	0.000000E-02	10.97	0.000000E-02	11.00	0.000000E-02	11.03	0.000000E-02	11.06	0.000000E-02	11.09	0.000000E-02	11.12	0.000000E-02	11.15	0.000000E-02	11.18	0.000000E-02	11.21	0.000000E-02	11.24	0.000000E-02	11.27	0.000000E-02	11.30	0.000000E-02	11.33	0.000000E-02	11.36	0.000000E-02	11.39	0.000000E-02	11.42	0.000000E-02	11.45	0.000000E-02	11.48	0.000000E-02	11.51	0.000000E-02	11.54	0.000000E-02	11.57	0.000000E-02	11.60	0.000000E-02	11.63	0.000000E-02	11.66	0.000000E-02	11.69	0.000000E-02	11.72	0.000000E-02	11.75	0.000000E-02	11.78	0.000000E-02	11.81	0.000000E-02	11.84	0.000000E-02	11.87	0.000000E-02	11.90	0.000000E-02	11.93	0.000000E-02	11.96	0.000000E-02	11.99	0.000000E-02	12.02	0.000000E-02	12.05	0.000000E-02	12.08	0.000000E-02	12.11	0.000000E-02	12.14	0.000000E-02	12.17	0.000000E-02	12.20	0.000000E-02	12.23	0.000000E-02	12.26	0.000000E-02	12.29	0.000000E-02	12.32	0.000000E-02	12.35	0.000000E-02	12.38	0.000000E-02	12.41	0.000000E-02	12.44	0.000000E-02	12.47	0.000000E-02	12.50	0.000000E-02	12.53	0.000000E-02	12.56	0.000000E-02	12.59	0.000000E-02	12.62	0.000000E-02	12.65	0.000000E-02	12.68	0.000000E-02	12.71	0.000000E-02	12.74	0.000000E-02	12.77	0.000000E-02	12.80	0.000000E-02	12.83	0.000000E-02	12.86	0.000000E-02	12.89	0.000000E-02	12.92	0.000000E-02	12.95	0.000000E-02	12.98	0.000000E-02	13.01	0.000000E-02	13.04	0.000000E-02	13.07	0.000000E-02	13.10	0.000000E-02	13.13	0.000000E-02	13.16	0.000000E-02	13.19	0.000000E-02	13.22	0.000000E-02	13.25	0.000000E-02	13.28	0.000000E-02	13.31	0.000000E-02	13.34	0.000000E-02	13.37	0.000000E-02	13.40	0.000000E-02	13.43	0.000000E-02	13.46	0.000000E-02	13.49	0.000000E-02	13.52	0.000000E-02	13.55	0.000000E-02	13.58	0.000000E-02	13.61	0.000000E-02	13.64	0.000000E-02	13.67	0.000000E-02	13.70	0.000000E-02	13.73	0.000000E-02	13.76	0.000000E-02	13.79	0.000000E-02	13.82	0.000000E-02	13.85	0.000000E-02	13.88	0.000000E-02	13.91	0.000000E-02	13.94	0.000000E-02	13.97	0.000000E-02	14.00	0.000000E-02	14.03	0.000000E-02	14.06	0.000000E-02	14.09	0.000000E-02	14.12	0.000000E-02	14.15	0.000000E-02	14.18	0.000000E-02	14.21	0.000000E-02	14.24	0.000000E-02	14.27	0.000000E-02	14.30	0.000000E-02	14.33	0.000000E-02	14.36	0.000000E-02	14.39	0.000000E-02	14.42	0.000000E-02	14.45	0.000000E-02	14.48	0.000000E-02	14.51	0.000000E-02	14.54	0.000000E-02	14.57	0.000000E-02	14.60	0.000000E-02	14.63	0.000000E-02	14.66	0.000000E-02	14.69	0.000000E-02	14.72	0.000000E-02	14.75	0.000000E-02	14.78	0.000000E-02	14.81	0.000000E-02	14.84	0.000000E-02	14.87	0.000000E-02	14.90	0.000000E-02	14.93	0.000000E-02	14.96	0.000000E-02	14.99	0.000000E-02	15.02	0.000000E-02	15.05	0.000000E-02	15.08	0.000000E-02	15.11	0.000000E-02	15.14	0.000000E-02	15.17	0.000000E-02	15.20	0.000000E-02	15.23	0.000000E-02	15.26	0.000000E-02	15.29	0.000000E-02	15.32	0.000000E-02	15.35	0.000000E-02	15.38	0.000000E-02	15.41	0.000000E-02	15.44	0.000000E-02	15.47	0.000000E-02	15.50	0.000000E-02	15.53	0.000000E-02	15.56	0.000000E-02	15.59	0.000000E-02	15.62	0.000000E-02	15.65	0.000000E-02	15.68	0.000000E-02	15.71	0.000000E-02	15.74	0.0
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ASP IFST CASE 5



APPENDIX B

FORTRAN COMPUTER PROGRAM LISTING

C		10
C	ASP - AERIAL SPRAY PROGRAM	20
C		30
C	ASP WAS BEEN DESIGNED TO PREDICT THE GROUND DEPOSITION	40
C	AND PERCENT OF MATERIAL LOST DUE TO DRIFT FROM AN	50
C	AGRICULTURAL AIRCRAFT. THE PROGRAM IS WRITTEN TO INPUT	60
C	DATA IN SI UNITS AND INCLUDES THE FOLLOWING OPTIONS:	70
C	1) 2D OR 3D WAKE MODEL AND DROPLET DYNAMICS	80
C	2) A DROPLET EVAPORATION MODEL	90
C	3) A PROPELLER SLIPSTREAM MODEL	100
C	4) A CROSSWIND MODEL	110
C	5) A TUNNEL WALL MODEL	120
C	6) THE ABILITY FOR THE USER TO SUPPLY HIS OWN	130
C	FLOWFIELD MODEL.	140
C		150
C	DOCUMENTATION AVAILABLE IN THE FORM OF A NASA CONTRACTORS	160
C	REPORT	170
C		180
C	CODE WRITTEN AT THE AERONAUTICAL AND ASTRONAUTICAL ENGINEERING	190
C	RESEARCH LABORATORY, THE OHIO STATE UNIVERSITY, COLUMBUS,	200
C	OHIO 1981 AUTHOR MICHAEL B. BRAGG	210
C		220
C	LAST UPDATE 2/1/81	230
C		240
C		250
CDC	PROGRAM MAIN(INPUT,OUTPUT,TAPE8,TAPE5=INPUT,TAPE6=OUTPUT)	260
	DIMENSION YNOZ(100),SIZE(15),DIST(1001),DRIFT(100),NRUN(100)	270
	DIMENSION DIAMN(100),STDEV(100),Q(100),ZNOZ(100),DIAG(15,100)	280
	DIMENSION IRUN(50),TRAJ(15,100),SIZ(15,100)	290
	*,SLP(15),XLIST(5),OLIST(5)	300
	DIMENSION YY(13,13),SAVE(13,13),CSAVE(13,3),YMAX(13),MET(100)	310
	DIMENSION ERROR(13),DY(13),YY1(13),PW(13,13),IP(13),TITLE(13)	320
	COMMON /AREA1/ A,CL,B,U,G,ZY0,ZZ0,DA,VIS,VCH,D,DD	330
	COMMON /AREA2/ N2D,N3D,NEVAP,EPS,NPROP2,NCW,NTUN,NDIST,NPRINT	340
	COMMON /AREA3/ PSWB,PAE,PV,SCN,DV	350
	COMMON /AREA5/ ZPROP2,PK1,PK2,PK3,PK4,RP	360
	COMMON /AREA6/ JJ,IRUN,ICOUNT	370
	COMMON /AREA7/ NVAR	380
	COMMON /AREA10/ TRAJ,DIAG,SIZ	390
C		400
C	INPUT THE NECESSARY PARAMETERS ACCORDING TO THE OPTIONS	410
C	SELECTED AND OUTPUT HEADER AND TITLE	420
C		430
	800 CONTINUE	440
	REWIND 8	450
	READ(5,3,END=700) (TITLE(K),K=1,13)	460
	3 FORMAT(13A6)	470
	WRITE(6,5) (TITLE(K),K=1,13)	480
	5 FORMAT("1",T50,"OSU / AARL"/"0",T50,	490
	& "AERIAL SPRAY PROGRAM"/"0",T50,	500
	& "LAST UPDATE 2/1/81"/"0",13A6//)	510
	DO 1 K=1,100	520
	DRIFT(K)=0.	530
	NRUN(K)=0	540
	DO 1 J=1,15	550

	SIZ(J,K)=0.	560
	DIAG(J,K)=0.	570
1	TRAJ(J,K)=0.	580
	DO K=1,1001	590
2	DISI(K)=0.	600
	ICOUNT=1	610
	JJ=0	620
	READ(5,10)NDROPS	630
10	FORMAT(I5)	640
	DO I=1,NDROPS	650
	READ(5,20) N2D,N3D,NEVAP,NPROP2,NCW,NTUN,NDIST,	660
	& NPRINT,NPLOT,EPS	670
20	FORMAT(9I2,2X,E15.6)	680
	IF(NEVAP.EQ.1) READ(5,30) PA,TDB,TWB	690
30	FORMAT(3F10.6)	700
	IF(NPROP2.EQ.1) READ(5,50) ZPROP2,PK1,PK2,PK3,PK4,RP	710
50	FORMAT(6F10.4)	720
	READ(5,40) A,CL,R,U,G,ZY0,ZZ0	730
60	FORMAT(7F10.4)	740
	READ(5,70) DA,VIS,VCW,0	750
70	FORMAT(2F20.6,2F10.4)	760
	IF(NDIST.EQ.1) GO TO 90	770
	READ(5,90) DIA,DD,X,Y,Z,UD,VD,WD	780
80	FORMAT(8F10.5)	790
	GO TO 120	800
90	CONTINUE	810
	READ(5,95) NCOL,NROW,NDIAMN,NSTDEV,NQ,NZNOZ	820
95	FORMAT(6I5)	830
	READ(5,100)DN,X0,UD0,VD0,WD0,SWIDTH,DERR,DWIDTH	840
100	FORMAT(8F10.5)	850
	READ(5,110) (YNOZ(K),K=1,NCOL)	860
110	FORMAT(8F10.5)	870
	READ(5,110) (SIZE(K),K=1,NROW)	880
	IF(NDIAMN.EQ.1) READ(5,110) (DIAMN(K),K=1,NCOL)	890
	IF(NDIAMN.EQ.0) READ(5,110) DIAMN1	900
	IF(NSTDEV.EQ.1) READ(5,110) (STDEV(K),K=1,NCOL)	910
	IF(NSTDEV.EQ.0) READ(5,110) STDEV1	920
	IF(NQ.EQ.1) READ(5,110) (Q(K),K=1,NCOL)	930
	IF(NQ.EQ.0) READ(5,110) Q1	940
	IF(NZNOZ.EQ.1) READ(5,110) (ZNOZ(K),K=1,NCOL)	950
	IF(NZNOZ.EQ.0) READ(5,110) ZNOZ1	960
	DO 115 K=1,NCOL	970
	IF(NDIAMN.EQ.0) DIAMN(K)=DIAMN1	980
	IF(NQ.EQ.0) Q(K)=Q1	990
	IF(NSTDEV.EQ.0) STDEV(K)=STDEV1	1000
	IF(NZNOZ.EQ.0) ZNOZ(K)=ZNOZ1	1010
115	CONTINUE	1020
120	CONTINUE	1030
	C=3.141592653589793	1040
C		1050
C	CALCULATE EVAPORATION PARAMETERS	1060
C		1070
	IF(NEVAP.EQ.0) GO TO 130	1080
	TDBK=TDB+.27+.15	1090
	TDBH=TDB*(9./5.)+.37+.459.67	1100

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      TWBK=TWR*(9./5.)*32.+459.67
      PAE=PA*1.4504E-4
      PS=EXP(54.6329-12301.688/TDBR-5.16923*ALOG(TDBR))
      PSWB=EXP(54.6329-12301.688/TWR-5.16923*ALOG(TWR))
      PV=PSWB-((.2405*(PAE-PSWB))/(.62194*(1075.8965-.56983*
&      (TWR-491.69))))*(TDBR-TWR)
      RH=PV/PS
      DA=((PAE*144.)/(G*3.280839*53.34*TDBR))*515.4
      DV=(5.2PF-6*TDBK**1.88)*1.E-4
      VIS=1.795E-5*(TDBK/293.16)**1.5*((293.16+110.)/(TDBK+110.))
      SCN=(DV*DA)/VIS
130 CONTINUE
C
C      DETERMINE NVAR, THE NUMBER OF SIMULTANEOUS FIRST ORDER
C      DIFFERENTIAL EQUATIONS TO BE SOLVED
C
      NVAR=4
      IF(N3D.NE.0) NVAR=6
      IF(NEVAP.EQ.1) NVAR=NVAR+1
      IF(N2D.FQ.2.OR.N3D.EQ.2) GO TO 138
      IF(NTUN.EQ.1) GO TO 136
      IF(NCW.EQ.1.AND.NPROP2.EQ.1) GO TO 132
      IF(NLW.FQ.1) GO TO 134
      IF(NPROP2.EQ.1) NVAR=NVAR+1
      GO TO 138
132 NVAR=NVAR+6
      GO TO 138
134 NVAR=NVAR+4
      GO TO 138
136 NVAR=NVAR+2
      IF(NPROP2.EQ.1) NVAR=NVAR+1
      GO TO 138
138 CONTINUE
C
C      OUTPUT THE CONSTANTS THAT WERE INPUT
C
      WRITE(6,150) N2D,N3D,NEVAP,NPROP2,NCW,NTUN,NDIST,NPRINT,
&      NPLT,EPS
150 FORMAT("0","PROGRAM CONTROL",// " ",T5,"N2D=",I2,T25,"N3D=",I2,
&      T45,"NEVAP=",I2,T65,"NPROP2=",I2,T85,"NCW=",I2/" "
&      ,T5,"NTUN=",I2,T25,"NDIST=",I2,T45,"NPRINT=",I2,T65,
&      "NPLT=",I2,T85,"EPS=",E13.6/)
      WRITE(6,160) A,CL,B,U,G,ZY0,ZZ0,DA,VIS,VCW,D,DD
160 FORMAT("0","PARAMETER VALUES",// " ",T5,"A=",F10.4,T25,"CL=",
&      F10.4,T45,"B=",F10.4,T65,"U=",F10.4,T85,"G=",F10.4/" "
&      T5,"ZY0=",F10.4,T25,"ZZ0=",F10.4,T45,"DA=",E13.6,T65,
&      "VIS=",E13.6,T85,"VCW=",F10.4/" " ,T5,"D=",F10.4,
&      T25,"DD=",F10.4/)
      IF(NEVAP.EQ.1) WRITE(6,170) PA,TDB,TWB
170 FORMAT("0","EVAPORATION INPUTS",// " ",T5,"PA=",F10.2,T25,
&      "TDB=",F10.6,T45,"TWB=",F10.6/)
      IF(NPROP2.EQ.1) WRITE(6,190) ZPROP2,PK1,PK2,PK3,PK4,RP
190 FORMAT("0","NPROP2 INPUTS",// " ",T5,"ZPROP2=",F10.4,T25,
&      "PK1=",F10.4,T45,"PK2=",F10.4,T65,"PK3=",F10.4
&      ,T85,"PK4=",F10.4/" " ,T5,"RP=",F10.4/)

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C	CHANGE FROM RECTANGULAR TO ELLIPTICAL LOAD DISTRIBUTION	1660
C	BY ADJUSTING CL	1670
	CL=(4./C)*CL	1680
	IF(NDIST.EQ.1) GO TO 220	1690
	WRITE(6,195) X,Y,Z,UD,VD,WD,DIA	1700
195	FORMAT("0","INITIAL VALUES"/" " ,T5,"X=",F10.6,T25,	1710
&	"Y=",F10.6,T45,"Z=",F10.6,T65,"UD=",F10.6,T85,	1720
&	"VD=",F10.6/" " ,T5,"WD=",F10.6,T25,"DIA=",	1730
&	F10.4/)	1740
C		1750
C	CALL SUB2D OR SUB3D TO CALCULATE THE DROPLET TRAJECTORY	1760
C		1770
	IF(N2D.NE.0) CALL SUB2D(NVAR,Y,Z,VD,WD,T,DIA,YY,SAVE	1780
&	,CSAVE,YMAX,ERROR,DY,YY1,PW,IP)	1790
	IF(N3D.NE.0) CALL SUB3D(NVAR,X,Y,Z,UD,VD,WD,T,DIA,YY,SAVE	1800
&	,CSAVE,YMAX,ERROR,DY,YY1,PW,IP)	1810
	ICOUNT=ICOUNT+1	1820
	WRITE(6,200) X,Y,Z,UD,VD,WD,DIA,T	1830
200	FORMAT("0","FINAL VALUES"/" " ,T5,"X=",F10.6,T25,"Y=",	1840
&	F10.6,T45,"Z=",F10.6,T65,"UD=",F10.6,T85,"VD=",	1850
&	F10.6/" " ,T5,"WD=",F10.6,T25,"DIA=",F10.4,T45,	1860
&	"T=",F10.6/)	1870
210	CONTINUE	1880
	GO TO 320	1890
220	CONTINUE	1900
C		1910
C	THIS PART OF THE MAIN PROGRAM CONTROLS THE NDIST=1 CASE	1920
C		1930
	WRITE(6,225) NCOL,NROW,NDIAMN,NSTDEV,NQ,NZNOZ,DD,	1940
&	X0,UD0,VD0,WD0,SWIDTH,DERR,DWIDTH	1950
225	FORMAT("0","DISTRIBUTION INPUTS"/" " ,T5,"NCOL=",I2,	1960
&	T25,"NROW=",I2,T45,"NDIAMN=",I2,T65,"NSTDEV=",I2,	1970
&	T85,"NQ=",I2/" " ,T5,"NZNOZ=",I2,T25,"DD=",F10.5,	1980
&	T45,"X=",F10.5,T65,"UD=",F10.5,T85,"VD=",F10.5/	1990
&	" " ,T5,"WD=",F10.5,T25,"SWIDTH=",F10.5,T45,"DERR=",	2000
&	F10.5,T65,"DWIDTH=",F10.5)	2010
C		2020
C	PERFORM ALL THE TRAJECTORIES REQUIRED TO CALCULATE THE	2030
C	REQUIRED DISTRIBUTION	2040
C		2050
	DO 230 I=1,NCOL	2060
	DO 240 J=1,NROW	2070
	DIA=SIZE(J)	2080
	DIA1=DIA	2090
	K=K+1	2100
	T=0.	2110
	X=X0	2120
	UD=UD0	2130
	VD=VD0	2140
	WD=WD0	2150
	Z=ZNOZ(I)	2160
	Y=YNOZ(J)	2170
	IF(N2D.NF.0) CALL SUB2D(NVAR,Y,Z,VD,WD,T,DIA,YY,SAVE,	2180
&	CSAVE,YMAX,ERROR,DY,YY1,PW,IP)	2190
	IF(N3D.NF.0) CALL SUB3D(NVAR,X,Y,Z,UD,VD,WD,T,DIA,YY,SAVE,	2200

S	CSAVE,YMAX,ERROR,DY,YY1,PW,IP)	2210
	TRAJ(J,I)=Y	2220
	DIAG(J,I)=DIA	2230
	SIZ(J,I)=SIZE(J)	2240
	IF(Z.NE.0.) GO TO 241	2250
	NRUN(I)=NRUN(I)+1	2260
	IF(Z.EQ.0.AND.ABS(Y).GT.DWIDTH) GO TO 241	2270
240	CONTINUE	2280
C		2290
C	THIS SECTION UP TO STMT NO 250 CONTROLS THE CALCULATION	2300
C	OF ADDITIONAL TRAJECTORIES IN ORDER TO ESTIMATE THE DRIFT	2310
C	AND INSERTS PARTICLES IN REGIONS OF A LARGE GRADIENT	2320
C	UU1A/DYG	2330
C		2340
	KMISS=0	2350
	SIZMIS=0.	2360
241	CONTINUE	2370
	J=NRUN(I)+1	2380
	MET(I)=0	2390
	IF(J.EQ.1) DRIFT(I)=SIZ(J,I)	2400
	IF(J.EQ.1) GO TO 250	2410
	DSIZ1=ABS(DIAMN(I)-SIZ(J-1,I))	2420
	K250=0	2430
	GO TO 243	2440
234	CONTINUE	2450
	IF(Z.EQ.0.AND.ABS(Y).GT.DWIDTH) GO TO 250	2460
	IF((DSIZ1/STDEV(I)).LT.4.) GO TO 242	2470
	DRIFT(I)=0.	2480
	GO TO 250	2490
242	CONTINUE	2500
C		2510
C	DEFINE ACCEPTABLE ERROR IN TERMS OF STDEV	2520
C		2530
	ERR=.5*STDEV(I)	2540
	IF(USIZ1.LT.(2.*STDEV(I))) ERR=.30*STDEV(I)	2550
	IF(USIZ1.LT.STDEV(I)) ERR=.20*STDEV(I)	2560
	ERR=ERR*0ERR	2570
246	CONTINUE	2580
	IF(Z.NE.0.) KMISS=1	2590
	IF(Z.NE.0.) SIZMIS=SIZ(J,I)	2600
	IF(J.GE.3) GO TO 239	2610
	IF(J.EQ.2) SIZ(J,I)=SIZ(J-1,I)-(SIZ(J-1,I)-SIZMIS)/2.	2620
	GO TO 245	2630
239	CONTINUE	2640
C		2650
C	DETERMINE PARTICLE SIZE FOR DRIFT ESTIMATE	2660
C	AND CALCULATE TRAJECTORIES	2670
C		2680
	YGM1=TRAJ(J-2,I)	2690
	YG=TRAJ(J-1,I)	2700
	Y1=ABS(1./(YG-YN0Z(I)))	2710
	Y2=ABS(1./(YGM1-YN0Z(I)))	2720
	SLOPE=(SIZ(J-2,I)-SIZ(J-1,I))/(Y2-Y1)	2730
	SIZ(J,I)=SIZ(J-1,I)-SLOPE*Y1/2	2740
	IF((SLOPE*Y1/2.).LT.ERR) SIZ(J,I)=SIZ(J-1,I)-ERR	2750

	IF(SIZ(J,I).GT.SIZMIS) GO TO 245	2760
	SIZ(J,I)=SIZMIS+(SIZ(J-1,I)-SIZMIS)/2.	2770
245	CONTINUE	2780
	DIA=SIZ(J,I)	2790
	DIA1=DIA	2800
	T=0.	2810
	X=XU	2820
	UD=UD0	2830
	VD=VD0	2840
	WD=WD0	2850
	Z=ZNUZ(I)	2860
	Y=YNOZ(I)	2870
	IF(N2D.NF.0) CALL SUB2D(NVAR,Y,Z,VD,WD,T,DIA,YY,SAVE,	2880
	& CSAVE,YMAX,ERROR,DY,YY1,PW,IP)	2890
	IF(N3D.NE.0) CALL SUB3D(NVAR,X,Y,Z,UD,VD,WD,T,DIA,YY,SAVE,	2900
	& CSAVE,YMAX,ERROR,DY,YY1,PW,IP)	2910
C		2920
C	STORE TRAJECTORY INFORMATION, IF ERROR.LT.ERR GO TO 250	2930
C	OTHERWISE ITERATE	2940
C		2950
	TRAJ(J,I)=Y	2960
	DIA9(J,I)=DIA	2970
	IF(MET(I).EQ.1) GO TO 228	2980
	IF(L.NE.0.) GO TO 246	2990
	NRUN(I)=NRUN(I)+1	3000
	IF(ABS(Y).LT.DWIDTH) GO TO 330	3010
	GO TO 250	3020
330	CONTINUE	3030
	IF((SIZ(J,I)-SIZMIS).GT.ERR) GO TO 244	3040
	DRIFT(I)=SIZ(J,I)	3050
	K250=1	3060
	GO TO 243	3070
244	CONTINUE	3080
	J=J+1	3090
	IF(J.LE.15) GO TO 246	3100
	WRITE(6,247)	3110
247	FORMAT(" ", "**** MAX NUMBER OF RUNS EXCEEDED BEFORE DRIFT",	3120
	& " CALCULATION COMPLETED")	3130
	DRIFT(I)=SIZ(J,I)	3140
	GO TO 250	3150
243	CONTINUE	3160
C		3170
C	THIS SECTION HANDLES THE CASE WHERE DYG/DDIA CHANGES	3180
C	SIGN BETWEEN POINTS	3190
C		3200
	IF(J.EQ.2) GO TO 227	3210
	J2=J-2	3220
	DO 410 K=1,J2	3230
	SLP(K)=(TRAJ(K+1,I)-TRAJ(K,I))/(SIZ(K+1,I)-SIZ(K,I))	3240
410	CONTINUE	3250
	DO 221 K=2,J2	3260
	IF(SLP(K).GT.0.AND.SLP(K-1).LT.0.) GO TO 222	3270
	IF(SLP(K).LT.0.AND.SLP(K-1).GT.0.) GO TO 222	3280
221	CONTINUE	3290
	IF(K<50.EQ.1) GO TO 250	3300

	GO TO 234	3310
222	CONTINUE	3320
C		3330
C	INSERT PARTICLES WHERE DYG/DDIA CHANGES SIGN	3340
C		3350
	MET(I)=1	3360
	IF(K250.EQ.1) GO TO 250	3370
	IADD=0	3380
	DO 223 K2=2,J2	3390
	K=K2	3400
	IF(SLP(K).LT.0.AND.SLP(K-1).LT.0.) GO TO 223	3410
	IF(SLP(K).GT.0.AND.SLP(K-1).GT.0.) GO TO 223	3420
	IF(IADD.NE.0) K=K2+IADD	3430
	DELTA=ARS(SIZ(K,I)-DIAMN(I))/STDEV(I)	3440
	ER=1.0*STDEV(I)	3450
	IF(DELTA.LT.2.) ER=.75*STDEV(I)	3460
	IF(DELTA.LT.1.) ER=.50*STDEV(I)	3470
	ER=ER*DERR	3480
	IDELTA=ARS(SIZ(K+1,I)-SIZ(K,I))/ER	3490
	IADD=IADD+IDELTA	3500
	DELTA=(SIZ(K+1,I)-SIZ(K,I))/(IDELTA+1)	3510
	K1=K+1	3520
	J1=J-1	3530
	DO 224 K5=K1,J1	3540
	KK=K1-K5+J1	3550
	TRAJ(KK+IDELTA,I)=TRAJ(KK,I)	3560
	DIAG(KK+IDELTA,I)=DIAG(KK,I)	3570
	SIZ(KK+IDELTA,I)=SIZ(KK,I)	3580
224	CONTINUE	3590
	DO 226 KK=1,IDELTA	3600
	SIZ(K+KK,I)=SIZ(K,I)+KK*DELTA	3610
	DIA=SIZ(K+KK,I)	3620
	T=0.	3630
	X=XU	3640
	UD=UUD	3650
	VD=VDU	3660
	WD=WDU	3670
	Z=ZNUZ(T)	3680
	Y=YNUZ(T)	3690
	IF(N2D.NE.0) CALL SUB2D(NVAR,Y,Z,VD,WD,T,DIA,YY,SAVE,	3700
	CSAVE,YMAX,ERROR,DY,YY1,PW,IP)	3710
	IF(N3D.NE.0) CALL SUB3D(NVAR,X,Y,Z,UD,VD,WD,T,DIA,YY,SAVE,	3720
	CSAVE,YMAX,ERROR,DY,YY1,PW,IP)	3730
	TRAJ(KK+K,I)=Y	3740
	DIAG(KK+K,I)=DIA	3750
	NRUN(I)=NRUN(I)+1	3760
	J=J+1	3770
226	CONTINUE	3780
	NRQ=NRUN(I)	3790
223	CONTINUE	3800
227	CONTINUE	3810
C		3820
C	DETERMINE PARTICLE SIZE FOR DRIFT ESTIMATE AND	3830
C	CALCULATE THE TRAJECTORY	3840
C		3850

	DSIZ1=ABS(DIAMN(I)-SIZ(J-1,I))	3860
	IF((USI71/STDEV(I)).LT.4) GO TO 237	3870
	DRIFT(I)=0.	3880
	GO TO 250	3890
237	CONTINUE	3900
	ERR=.5*STDEV(I)	3910
	IF(DSIZ1.LT.(2.*STDEV(I))) ERR=.30*STDEV(I)	3920
	IF(USI71.LT.STDEV(I)) ERR=.20*STDEV(I)	3930
	ERR=ERR*0.99	3940
236	CONTINUE	3950
	IF(KMISS.EQ.0) SIZ(J,I)=SIZ(J-1,I)-1.1*ERR	3960
	IF(KMISS.EQ.1) SIZ(J,I)=SIZ(J-1,I)-(SIZ(J-1,I)-SIZMIS)/2.	3970
	GO TO 245	3980
228	CONTINUE	3990
C		4000
C	IF PARTICLE WITHIN ERR GO TO 250 OTHERWISE ITERATE	4010
C		4020
	IF(Z.EQ.0.) GO TO 229	4030
	KMISS=1	4040
	SIZMIS=SIZ(J,I)	4050
	GO TO 227	4060
229	CONTINUE	4070
	NRUN(I)=NRUN(I)+1	4080
	IF(ABS(Y).LT.DWIDTH) GO TO 340	4090
	DRIFT(I)=SIZ(J-1,I)-ABS((SIZ(J,I)-SIZ(J-1,I))/(Y-TRAJ(J-1,I)))*	4100
	& (DWIDTH-ABS(TRAJ(J-1,I)))	4110
	GO TO 250	4120
340	CONTINUE	4130
	IF(ABS(SIZ(J,I)-SIZ(J-1,I)).GT.ERR) GO TO 233	4140
	DRIFT(I)=SIZ(J,I)	4150
	GO TO 250	4160
233	CONTINUE	4170
	J=J+1	4180
	IF(J.LE.15) GO TO 236	4190
	WRITE(6,247)	4200
250	CONTINUE	4210
C		4220
C	COMBINE TRAJECTORY INFORMATION TO BUILD-UP THE DISTRIBUTION	4230
C	AND ESTIMATE THE DRIFT OUTPUT RESULTS	4240
C		4250
	CALL SWATH(NCOL,NROW,Q,DIAMN,STDEV,YNOZ,DWIDTH,NRUN,MET,	4260
	& SWIDTH,DIST,DRIFT)	4270
	WRITE(6,252)	4280
252	FORMAT("0","NOZZLE NO.",8X,"YNOZ",11X,"ZNOZ",8X,	4290
	& "DIAMN",10X,"STDEV",13X,"Q",12X,"DRIFT"/)	4300
	DO 255 K=1,NCOL	4310
	WRITE(6,253) K,YNOZ(K),ZNOZ(K),DIAMN(K),STDEV(K),Q(K),DRIFT(K)	4320
253	FORMAT(" ",4X,I3.6F15.4)	4330
255	CONTINUE	4340
	WRITE(6,256)	4350
256	FORMAT("1"//" SINGLE DROPLET LATERAL DISPLACEMENT"/)	4360
C		4370
C	OUTPUT TRAJECTORY INFORMATION TO PRINTER	4380
C		4390
	NP=NCOL/3	4400

NLEFT=NCNL-NP*3	4410
KP=-2	4420
IF(NP.EQ.0) GO TO 425	4430
DO 405 K1=1,NP	4440
KK=IFIX(K1/2.)*2	4450
IF(K1.EQ.KK) WRITE(6,415)	4460
415 FORMAT("1"," ")	4470
KP=K1*3-2	4480
KP1=KP+1	4490
KP2=KP+2	4500
IF(NEVAP.EQ.1) GO TO 450	4510
WRITE(6,420) KP,KP1,KP2	4520
420 FORMAT("0","NOZZLE:",T13,I3,T55,I3,T97,I3/" ",T7,"YG",T21,"DIA",	4530
& T49,"YG",T63,"DIA",T91,"YG",T105,"DIA")	4540
DO 430 K2=1,15	4550
WRITE(6,440) TRAJ(K2,KP),SIZ(K2,KP),TRAJ(K2,KP+1),SIZ(K2,KP+1),	4560
& TRAJ(K2,KP+2),SIZ(K2,KP+2)	4570
440 FORMAT(" ",2F14.6,14X,2E14.6,14X,2E14.6)	4580
430 CONTINUE	4590
GO TO 405	4600
450 CONTINUE	4610
WRITE(6,460) KP,KP1,KP2	4620
460 FORMAT("0","NOZZLE:",T20,I3,T62,I3,T104,I3/" ",T7,"YG",T21,"DIA",	4630
& T35,"DIAG",T49,"IG",T63,"DIA",T77,"DIAG",T91,"YG",T105,	4640
& "DIA",T119,"DIAG")	4650
DO 470 K2=1,15	4660
WRITE(6,480) TRAJ(K2,KP),SIZ(K2,KP),DIAG(K2,KP),TRAJ(K2,KP+1),	4670
& SIZ(K2,KP+1),DIAG(K2,KP+1),TRAJ(K2,KP+2),SIZ(K2,KP+2)	4680
& ,DIAG(K2,KP+2)	4690
480 FORMAT(" ",9E14.6)	4700
470 CONTINUE	4710
405 CONTINUE	4720
425 CONTINUE	4730
IF(NLEFT.EQ.0) GO TO 600	4740
KK=IFIX(K1/2.)*2	4750
KP3=KP+3	4760
KP4=KP+4	4770
IF(K1.NE.KK) WRITE(6,415)	4780
IF(NLEFT.EQ.1) GO TO 550	4790
IF(NEVAP.EQ.1) GO TO 510	4900
WRITE(6,490) KP3,KP4	4810
490 FORMAT("0","NOZZLE:",T13,I3,T55,I3/" ",T7,"YG",T21,"DIA",T49,"YG",	4820
& T63,"DIA")	4830
DO 500 K2=1,15	4840
WRITE(6,440) TRAJ(K2,KP+3),SIZ(K2,KP+3),TRAJ(K2,KP+4),SIZ(K2,KP+4)	4850
500 CONTINUE	4860
GO TO 600	4870
510 CONTINUE	4880
WRITE(6,520) KP3,KP4	4890
520 FORMAT("0","NOZZLE:",T20,I3,T62,I3/" ",T7,"YG",T21,"DIA",T35,	4900
& "DIAG",T49,"YG",T63,"DIA",T77,"DIAG")	4910
DO 530 K2=1,15	4920
WRITE(6,480) TRAJ(K2,KP+3),SIZ(K2,KP+3),DIAG(K2,KP+3),	4930
& TRAJ(K2,KP+4),SIZ(K2,KP+4),DIAG(K2,KP+4)	4940
530 CONTINUE	4950

GO TO 600	4960
550 CONTINUE	4970
IF (NEVAP.EQ.1) GO TO 580	4980
WRITE(6,560) KP3	4990
560 FORMAT("0","NOZZLE:",T13,I3/" ",T7,"YG",T21,"DIA")	5000
WRITE(6,570) (TRAJ(K2,KP+3),SIZ(K2,KP+3),K2=1,15)	5010
570 FORMAT(" ",2E14.6)	5020
GO TO 600	5030
580 CONTINUE	5040
WRITE(6,585) KP3	5050
585 FORMAT("0","NOZZLE:",T20,I3/" ",T7,"YG",T21,"DIA",T35,"DIAG")	5060
WRITE(6,590) (TRAJ(K2,KP+3),SIZ(K2,KP+3),DIAG(K2,KP+3),K2=1,15)	5070
590 FORMAT(" ",3F14.6)	5080
600 CONTINUE	5090
WRITE(6,260)	5100
260 FORMAT("1","FINAL DISTRIBUTION"/"0",5(3X," YG",6X,	5110
& "CONCENTRATION"))	5120
C	5130
C WRITE TO FILE 8 AND OUTPUT TO 6 THE FINAL DISTRIBUTION	5140
C	5150
KJ=0	5160
TOT=0.	5170
START=-SWIDTH	5180
NPTS=SWIDTH*200+1	5190
JJ=NPTS	5200
DO J=1,NPTS	5210
XPL=START+(J-1)*.01	5220
WRITE(8,302) XPL,DIST(J)	5230
302 FORMAT(2E13.6)	5240
TOT=TOT+DIST(J)	5250
IF(DIST(J).LT.1.E-4) GO TO 310	5260
IF(KJ.EQ.5) KJ=0	5270
KJ=KJ+1	5280
XLIST(KJ)=XPL	5290
DLIST(KJ)=DIST(J)	5300
IF(KJ.EQ.5) WRITE(6,300) (XLIST(K),DLIST(K),K=1,5)	5310
300 FORMAT(" ",5(3X,F6.2,3X,E13.6))	5320
310 CONTINUE	5330
IF(KJ.NE.5) WRITE(6,300) (XLIST(K),DLIST(K),K=1,KJ)	5340
C	5350
C DETERMINE TOTAL DRIFT	5360
C	5370
QF=0.	5380
DRIFTF=0.	5390
DO K=1,NCOL	5400
DRIFTF=DRIFTF+DRIFT(K)*Q(K)	5410
QF=QF+Q(K)	5420
261 CONTINUE	5430
DRIFTF=DRIFTF/QF	5440
WRITE(6,265) DRIFTF	5450
TOT=TOT/QF	5460
265 FORMAT("0","TOTAL DRIFT=",E13.6)	5470
WRITE(6,266) TOT	5480
266 FORMAT("0","TOTAL=",E14.6)	5490
320 CONTINUE	5500

	IF(NPLOT.EQ.1) CALL PLOT1(TITLE)	5510
	GO TO 800	5520
700	CONTINUE	5530
	STOP	5540
	END	5550
C		5560
C	SUBROUTINE SUB2D: CALLED BY THE MAIN PROGRAM, SUB2D CONTROLS	5570
C	THE TRAJECTORY CALCULATION FOR ALL 2D CASES. GIVEN THE	5580
C	INITIAL VALUES AND USER SELECTED OPTIONS, SUB2D CALCULATES	5590
C	THE TRAJECTORY AND RETURNS THE FINAL VALUES OF THE VARIABLES	5600
C		5610
	SUBROUTINE SUB2D(NVAR,Y,Z,VD,WD,T,DIA,YY,SAVE,CSAVE,	5620
	YMAX,ERROR,DY,YY1,PW,IP)	5630
	DIMENSION YY(13,NVAR),SAVE(13,NVAR),CSAVE(NVAR,3),YMAX(NVAR)	5640
	DIMENSION ERROR(NVAR),DY(NVAR),YY1(NVAR)	5650
	DIMENSION PW(NVAR,NVAR),IP(NVAR),IRUN(50)	5660
	COMMON /AREA1/ A,CL,B,U,G,ZY0,ZZ0,DA,VIS,VCW,D,DD	5670
	COMMON /AREA2/ N2D,N3D,NEVAP,EPS,NPROP2,NCW,NTUN,NDIST,NPRINT	5680
	COMMON /AREA5/ ZPROP2,PK1,PK2,PK3,PK4,RP	5690
	COMMON /AREA6/ JJ,IRUN,ICOUNT	5700
	COMMON /AREAR/ DIA1	5710
C		5720
C	INITIALIZE THE DEPENDENT VARIABLES AND STORE IN ARRAY	5730
C	YY(1,NVAR)	5740
C		5750
	DIA=DIA*1.E-6	5760
	DIA1=DIA	5770
	C=3.141592653589793	5780
	T=0.E0	5790
	YY(1,1)=Y	5800
	YY(1,2)=VD	5810
	YY(1,3)=7	5820
	YY(1,4)=WD	5830
	IF(NEVAP.EQ.1) YY(1,NVAR)=DIA	5840
	IF(N2D.EQ.2) GO TO 4	5850
	IF(NVAR.GE.10) GO TO 1	5860
	IF(NVAR.GE.8.AND.NPROP2.EQ.0) GO TO 2	5870
	IF(NTUN.EQ.1.AND.NPROP2.EQ.1) YY(1,7)=ZPROP2	5880
	IF(NTUN.EQ.1) GO TO 3	5890
	IF(NPROP2.EQ.1) YY(1,5)=ZPROP2	5900
	GO TO 4	5910
1	YY(1,10)=ZPROP2	5920
	YY(1,9)=0.	5930
2	YY(1,8)=7Z0	5940
	YY(1,7)=-ZY0	5950
3	YY(1,6)=7Z0	5960
	YY(1,5)=ZY0	5970
4	CONTINUE	5980
	IF(VU.EQ.0.AND.WD.EQ.0.) CALL INCON(YY,DIA,NVAR)	5990
	IF(YY(1,4).EQ.0.E0.AND.YY(1,2).EQ.0.E0)GO TO 20	6000
C		6010
C	INITIALIZE VALUES FOR DIFSUB	6020
C		6030
	H=1.E-4	6040
	MF=2	6050

	DO 5 I=1,NVAR	6060
	YMAX(I)=1.E0	6070
5	CONTINUE	6080
	MAXDER=6	6090
	HMIN=1.E-16	6100
	HMAX=1.E0	6110
	JSTART=0	6120
	LOOP=0	6130
	YPROP=0.	6140
C		6150
C	PERFORM THE STEP INTEGRATION BY SUCCESSIVE CALLS OF DIFSUB	6160
C		6170
C	DIFSUB IS A FORTUOI SUBROUTINE TO SOLVE A SYSTEM OF N SIMULTANE	6180
C	DIFFERENTIAL EQUATIONS	6190
C		6200
	DO 14 K=1,1000	6210
	DO 9 I=1,NVAR	6220
	YY1(I)=YY(1,I)	6230
9	CONTINUE	6240
	T1=	6250
C		6260
	CALL DIFSBM(NVAR,T,YY,SAVE,CSAVE,H,HMIN,HMAX,EPS,MF,YMAX,ERROR,	6270
	& KFLAG,JSTART,MAXDER,PW,IP)	6280
C		6290
C	MONITOR VARIABLES AFTER EACH TIME-STEP TO CONTROL OUTPUT	6300
C	AND STORAGE OF TRAJECTORY AND TERMINATE TRAJECTORY IF	6310
C	PARTICLE RECOMES ENTRAINED IN THE VORTEX OR HITS THE GROUND	6320
C		6330
	DIAPL=YY(1,NVAR)/1.E-6	6340
	IF(NEVAP.EQ.1) YY(1,NVAR)=YY(1,NVAR)/1.E-6	6350
	IF(NPRINT.EQ.1.AND.NOIST.EQ.0) WRITE (6,11) K,T,(YY(1,I),I=1,NVAR)	6360
11	FORMAT(" ",I4,4X,12F10.6)	6370
	IF(NEVAP.EQ.1) YY(1,NVAR)=YY(1,NVAR)*1.E-6	6380
	L=K+JJ	6390
	IF(YY(1,3).LT.0.E0)GO TO 16	6400
	IF(NOIST.EQ.1) GO TO 10	6410
	WRITE(8,22) T,YY(1,1),YY(1,3)	6420
22	FORMAT(3E13.6)	6430
10	CONTINUE	6440
	IF(NEVAP.EQ.1.AND.DIAPL.LT.15) GO TO 20	6450
	IF(NPROP2.NE.1) GO TO 25	6460
	IF(NCW.EQ.1) YPROP=YY(1,10)	6470
	IF(YY1(1).LE.YPROP.AND.YY(1,1).GT.YPROP) LOOP=LOOP+1	6480
	IF(LUOP.LT.2) GO TO 25	6490
	IF(NPRINT.EQ.1.AND.NOIST.EQ.0) WRITE(6,30)	6500
30	FORMAT("*** TRAJECTORY TERMINATED, PROPELLER SLIPSTREAM ",	6510
	& "ENTPAINMENT ***")	6520
	GO TO 20	6530
25	CONTINUE	6540
	IF(YY(1,3).GT.YY1(3))GO TO 12	6550
	GO TO 14	6560
12	IF(ABS(YY(1,1)).GT.1.E0)GO TO 13	6570
	GO TO 14	6580
13	IF(YY(1,1).GT.ZZ0.AND.ABS(YY(1,1)).LT.ABS(YY1(1))) GO TO 18	6590
14	CONTINUE	6600

	K=K-1	6610
	WRITE (6,15)	6620
15	FORMAT("0", "**** DO LOOP PARAMETER EXCEEDED IN SUB2D ****")	6630
	GO TO 20	6640
16	CONTINUE	6650
	IF(NDIST.EQ.0.AND.NPRINT.EQ.1) WRITE (6,17)	6660
17	FORMAT("0", "**** TRAJECTORY COMPLETED, Z LESS THAN ZERO ****")	6670
	CALL YZF1N(YY,YY1,NVAR,T,T1,L)	6680
	GO TO 20	6690
18	CONTINUE	6700
	IF(NDIST.EQ.0.AND.NPRINT.EQ.1) WRITE (6,19)	6710
19	FORMAT("0", "**** TRAJECTORY TERMINATED, VORTEX ENTRAINMENT ****")	6720
20	CONTINUE	6730
	JJ=JJ+K	6740
	IRUN(ICOUNT)=K	6750
	Y=YY(1,1)	6760
	VD=YY(1,2)	6770
	Z=YY(1,3)	6780
	WD=YY(1,4)	6790
	IF(NEVAP.EQ.1) DIA=YY(1,NVAR)/1.E-6	6800
	IF(NEVAP.EQ.0) DIA=DIA/1.E-6	6810
	RETURN	6820
	END	6830
C		6840
C	SUBROUTINE SUB3D: CALLED BY THE MAIN PROGRAM, SUB3D CONTROLS	6850
C	THE TRAJECTORY CALCULATION FOR ALL 3D CASES. GIVEN THE	6860
C	INITIAL VALUES AND USER SELECTED OPTIONS, SUB3D CALCULATES	6870
C	THE TRAJECTORY AND RETURNS THE FINAL VALUES OF THE VARIABLES	6880
C		6890
	SUBROUTINE SUB3D(NVAR,X,Y,Z,UD,VD,WD,T,DIA,YY,SAVE,CSAVE,	6900
	YMAX,ERROR,DY,YY1,PW,IP)	6910
	DIMENSION YY(13,NVAR),SAVE(13,NVAR),CSAVE(NVAR,3),YMAX(NVAR)	6920
	DIMENSION ERROR(NVAR),DY(NVAR),YY1(NVAR)	6930
	DIMENSION PW(NVAR,NVAR),IP(NVAR),IRUN(50)	6940
	COMMON /AREA1/ A,CL,B,U,G,ZY0,ZZ0,DA,VIS,VCW,D,DD	6950
	COMMON /AREA2/ N2D,N3D,NEVAP,FPS,NPROP2,NCW,NTUN,NDIST,NPRINT	6960
	COMMON /AREA5/ ZPROP2,PK1,PK2,PK3,PK4,RP	6970
	COMMON /AREA6/ JJ,IRUN,ICOUNT	6980
	COMMON /APEAR/ DIA1	6990
C		7000
C	INITIALIZE THE DEPENDENT VARIABLES AND STORE IN ARRAY	7010
C	YY(1,NVAR)	7020
C		7030
	DIA=DIA*1.E-6	7040
	DIA1=DIA	7050
	C=3.141592653589793	7060
	T=0.0	7070
	YY(1,1)=Y	7080
	YY(1,2)=VD	7090
	YY(1,3)=Z	7100
	YY(1,4)=WD	7110
	YY(1,5)=X	7120
	YY(1,6)=UD	7130
	IF(NEVAP.EQ.1) YY(1,NVAR)=DIA	7140
	IF(N3D.EQ.2) GO TO 4	7150

IF(NVAR.GE.12) GO TO 1	7160
IF(NVAR.GE.10) GO TO 2	7170
IF(NPROP2.EQ.1) YY(1,7)=ZPROP2	7180
GO TO 4	7190
1 YY(1,12)=ZPROP2	7200
YY(1,11)=0.	7210
2 YY(1,10)=ZZ0	7220
YY(1,9)=-ZY0	7230
3 YY(1,8)=ZZ0	7240
YY(1,7)=ZY0	7250
4 CONTINUE	7260
C	7270
C INITIALIZE VALUES FOR DIFSUB	7280
C	7290
H=1.E-4	7300
MF=2	7310
DO 5 I=1,NVAR	7320
YMAX(I)=1.E0	7330
5 CONTINUE	7340
MAXDER=6	7350
HMIN=1.E-16	7360
HMAX=1.E0	7370
JSTART=0	7380
LOOP=0	7390
YPROP=0.	7400
C	7410
C PERFORM THE STEP INTEGRATION BY SUCCESSIVE CALLS OF DIFSUB	7420
C	7430
C DIFSUB IS A FORTUOI SUBROUTINE TO SOLVE A SYSTEM OF N SIMULTANE	7440
C DIFFERENTIAL EQUATIONS	7450
C	7460
DO 14 K=1,1000	7470
DO 9 I=1,NVAR	7480
YY1(I)=YY(1,I)	7490
9 CONTINUE	7500
T1=T	7510
C	7520
CALL DIFSBM(NVAR,T,YY,SAVE,CSAVE,H,HMIN,HMAX,EPS,MF,YMAX,ERROR,	7530
8 KFLAG,JSTART,MAXDER,PW,IP)	7540
C	7550
C MONITOR VARIABLES AFTER EACH TIME-STEP TO CONTROL OUTPUT	7560
C AND STORAGE OF TRAJECTORY AND TERMINATE TRAJECTORY IF	7570
C PARTICLE BECOMES ENTHAINED IN THE VORTEX OR HITS THE GROUND	7580
C	7590
DIAPL=YY(1,NVAR)/1.E-6	7600
IF(NEVAP.EQ.1) YY(1,NVAR)=YY(1,NVAR)/1.E-6	7610
IF(NPRINT.EQ.1.AND.NDIST.EQ.0) WRITE (6,11) K,T,(YY(1,I),I=1,NVAR)	7620
11 FORMAT(" ",I4,4X,14F9.5)	7630
IF(NEVAP.EQ.1) YY(1,NVAR)=YY(1,NVAR)*1.E-6	7640
L=K+JJ	7650
IF(YY(1,3).LT.0.E0) GO TO 16	7660
IF(NDIST.EQ.1) GO TO 10	7670
WRITE(8,22) T,YY(1,1),YY(1,3)	7680
22 FORMAT(3F13.6)	7690
10 CONTINUE	7700

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      IF (NEVAP.EQ.1.AND.DIAPL.LT.15) GO TO 20
      IF (NPROP2.NF.1) GO TO 25
      IF (NCW.EQ.1) YPROP=YY(1,10)
      IF (YY1(1).LE.YPROP.AND.Y1(1,1).GT.YPROP) LOOP=LOOP+1
      IF (LOOP.LT.2) GO TO 25
      IF (NPRINT.EQ.1.AND.NDIST.EQ.0) WRITE(6,30)
30  FORMAT('*** TRAJECTORY TERMINATED, PROPELLER SLIPSTREAM ',
      &      'ENTRAINMENT ***')
      GO TO 20
25  CONTINUE
      IF (YY(1,3).GT.YY1(3)) GO TO 12
      GO TO 14
12  IF (ABS(YY(1,1)).GT.1.E0) GO TO 13
      GO TO 14
13  IF (YY(1,3).GT.ZZ0.AND.ABS(YY(1,1)).LT.ABS(YY1(1))) GO TO 18
14  CONTINUE
      K=K+1
      WRITE (6,15)
15  FORMAT('0','***** DO LOOP PARAMETER EXCEEDED IN SUB3D *****')
      GO TO 20
16  CONTINUE
      IF (NDIST.EQ.0.AND.NPRINT.EQ.1) WRITE (6,17)
17  FORMAT('0','***** TRAJECTORY COMPLETED, Z LESS THAN ZFRO *****')
      CALL YZF1N(YY,YY1,NVAR,T,T1,L)
      GO TO 20
18  CONTINUE
      IF (NDIST.EQ.0.AND.NPRINT.EQ.1) WRITE (6,19)
19  FORMAT('0','***** TRAJECTORY TERMINATED, VORTEX ENTRAINMENT *****')
20  CONTINUE
      JJ=JJ+K
      IRUN(ICOUNT)=K
      Y=YY(1,1)
      VD=YY(1,2)
      Z=YY(1,3)
      WD=YY(1,4)
      X=YY(1,5)
      UD=YY(1,6)
      IF (NEVAP.EQ.1) DIA=YY(1,NVAR)/1.E-6
      IF (NEVAP.EQ.0) DIA=DIA1/1.E-6
      RETURN
      END
C
C      PEDERV IS REQUIRED BY DIFSUB, BUT NOT NEEDED FOR THIS METHOD
C
      SUBROUTINE PEDERV(A,B,C,D)
      RETURN
      END
C
C      DIFFUN IS A SUBROUTINE CALLED BY DIFSUB TO EVALUATE THE RIGHT-
C      HAND SIDE OF THE SYSTEM OF DIFFERENTIAL EQUATIONS
C      KHS IS RETURNED TO DIFSUB IN ARRAY DY(NVAR)
C
      SUBROUTINE DIFFUN(NVAR,T,YY,DY)
      DIMENSION YY(13,NVAR),DY(NVAR)
      COMMON /AREA1/ A,CL,B,U,G,ZY0,ZZ0,DA,VIS,VCW,D,DD

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	COMMON /AREA2/ N2D,N3D,NEVAP,EPS,NPROP2,NCW,NTUN,NDIST,NPRINT	8260
	COMMON /AREA3/ PS4R,PA,pV,SCN,DV	8270
	COMMON /AREA5/ ZPROP2,PK1,PK2,PK3,PK4,RP	8280
	COMMON /AREA8/ DIA1	8290
	IF (NEVAP.EQ.0) DIA=DIA1	8300
	Y=YY(1,1)	8310
	VD=YY(1,2)	8320
	Z=YY(1,3)	8330
	WD=YY(1,4)	8340
	IF (N3D.EQ.1) X=YY(1,5)	8350
	IF (N3D.EQ.1) UD=YY(1,6)	8360
C		8370
C	CROSSWIND MODEL	8380
C		8390
	ZCW=Z	8400
	IF (ZCW.LT.0.) ZCW=0.	8410
	ZCW=ZCW*R	8420
	IF (NCW.EQ.0) VCW=0.	8430
		8440
C	BRANCH TO CORRECT SECTION OF CODE DEPENDING ON OPTIONS	8450
C	SELECTED	8460
C		8470
	IF (N2D.EQ.2.OR.N3D.EQ.2) GO TO 45	8480
	IF (N3D.EQ.1) GO TO 15	8490
	KK=0	8500
	IF (NVAR.EQ.11.OR.NVAR.EQ.10) GO TO 10	8510
	IF (NCW.EQ.1) GO TO 20	8520
	IF (NTUN.EQ.1) GO TO 30	8530
	IF (NPROP2.EQ.1) GO TO 35	8540
	IF (NVAR.EQ.5.OR.NVAR.EQ.4) GO TO 40	8550
	WRITE (6,9)	8560
9	FORMAT (' ', '***** NVAR INCORRECT *****')	8570
15	CONTINUE	8580
	KK=Z	8590
	IF (NVAR.GE.12) GO TO 10	8600
	IF (NVAR.GE.10) GO TO 20	8610
	IF (NVAR.GE.7.AND.NPROP2.EQ.1) GO TO 35	8620
	IF (NVAR.GE.6) GO TO 40	8630
	WRITE (6,9)	8640
10	CONTINUE	8650
		8660
C	BOTH VORTICES AND PROP POSITIONS ARE	8670
C	DETERMINED BY STEP INTEGRATION	8680
C	HERE NCW=1 AND NPROP2=1	8690
C		8700
	CALL VELTV(YY(1,5+KK),YY(1,6+KK),YY(1,7+KK),YY(1,8+KK),Y,Z,V1,w1)	8710
	CALL PROP2D(Y,Z,YY(1,9+KK),YY(1,10+KK),V2,w2)	8720
	VA=V1+V2+(VCW*ZCW**.25)/U	8730
	WA=W1+W2	8740
	IF (N3D.EQ.0) GO TO 17	8750
	CALL BOUND(X,Y,Z,UA,w3)	8760
	CALL PROPX(Y,Z,YY(1,11),YY(1,12),UAP)	8770
	UA=UA+UAP	8780
	WA=WA+W3	8790
17	CONTINUE	8800

	CALL VELTV(YY(1,5+KK),YY(1,6+KK),YY(1,7+KK),YY(1,8+KK),	8810
	5 YY(1,9+KK),YY(1,10+KK),VP1,WPI)	8820
	DY(7+KK)=VP1+(VCW*(YY(1,10+KK)*8)**.25)/U	8830
	DY(10+KK)=WP1	8840
	IF(YY(1,10+KK).LT.RP) DY(10+KK)=0.	8850
	CALL VELC(YY(1,5+KK),YY(1,6+KK),YY(1,7+KK),YY(1,8+KK),V1,W1)	8860
	DY(5+KK)=V1+(VCW*(YY(1,6+KK)*8)**.25)/U	8870
	DY(8+KK)=W1	8880
	CALL VELC(YY(1,7+KK),YY(1,8+KK),YY(1,5+KK),YY(1,6+KK),V1,W1)	8890
	DY(7+KK)=-V1+(VCW*(YY(1,8+KK)*8)**.25)/U	8900
	DY(8+KK)=-W1	8910
	GO TO 50	8920
20	CONTINUE	8930
C		8940
C	NCW=1: VORTEX POSITIONS DETERMINED BY STEP INTEGRATION,	8950
C	NO PROP	8960
C		8970
	CALL VELTV(YY(1,5+KK),YY(1,6+KK),YY(1,7+KK),YY(1,8+KK),Y,Z,V1,WA)	8980
	VA=V1+(VCW*ZCW**.25)/U	8990
	IF(N3D.FQ.0) GO TO 25	9000
	CALL BOUND(X,Y,Z,UA,W1)	9010
	WA=WA+W1	9020
25	CONTINUE	9030
	CALL VELC(YY(1,5+KK),YY(1,6+KK),YY(1,7+KK),YY(1,8+KK),V1,W1)	9040
	DY(5+KK)=V1+(VCW*(YY(1,6+KK)*8)**.25)/U	9050
	DY(8+KK)=W1	9060
	CALL VELC(YY(1,7+KK),YY(1,8+KK),YY(1,5+KK),YY(1,6+KK),V1,W1)	9070
	DY(7+KK)=-V1+(VCW*(YY(1,8+KK)*8)**.25)/U	9080
	DY(8+KK)=-W1	9090
	GO TO 50	9100
30	CONTINUE	9110
C		9120
C	NTUN=1: NPROP2=0 OR 1	9130
C		9140
	VP1=0.	9150
	WP1=0.	9160
	CALL TUNVEL(Y,Z,YY(1,5),YY(1,6),V1,W1)	9170
	IF(NPROP2.EQ.1) CALL PROP2D(Y,Z,0.,YY(1,7),VP1,WPI)	9180
	VA=V1+VP1+(VCW*ZCW**.25)/U	9190
	WA=W1+WP1	9200
	CALL VORTUN(YY(1,5),YY(1,6),DY(5),DY(6))	9210
	IF(NPROP2.EQ.0) GO TO 50	9220
	CALL TUNVEL(0.,YY(1,7),YY(1,5),YY(1,6),VDUM,DY(7))	9230
	GO TO 50	9240
35	CONTINUE	9250
C		9260
C	NPROP2=1: WITHOUT CROSSWIND OR TUNNEL	9270
C		9280
	CALL VEL2D(T,0.,YY(1,5+KK),V1,W1)	9290
	IF(YY(1,5+KK).LE.RP) W1=0.	9300
	DY(5+KK)=W1	9310
	CALL VEL2D(T,Y,Z,VA,WA)	9320
	CALL PROP2D(Y,Z,0.,YY(1,5+KK),VP1,WPI)	9330
	VA=VA+VP1	9340
	WA=WA+WPI	9350

	IF(N3D.FQ.0) GO TO 37	9360
	CALL BOUND(X,Y,Z,UA,w2)	9370
/	CALL PROPX(Y,Z,0.,YY(1,7),UAP)	9380
	UA=UA+UAP	9390
	WA=WA+w2	9400
37	CONTINUE	9410
	GO TO 50	9420
40	CONTINUE	9430
C		9440
C	N2D=1 OR N3D=1: NO OPTIONS EXCEPT NEVAP MAY EQUAL 1	9450
C		9460
	CALL VEL2D(T,Y,Z,VA,WA)	9470
	IF(N3D.FQ.0) GO TO 50	9480
	CALL BOUND(X,Y,Z,UA,w1)	9490
	WA=WA+w1	9500
	GO TO 50	9510
C		9520
C	N2D=2 OR N3D=2	9530
C	GET VELOCITY FROM USER SUPPLIED SUBROUTINE USERV	9540
C		9550
45	CONTINUE	9560
	CALL USERV(T,X,Y,Z,UA,VA,WA)	9570
50	CONTINUE	9580
C		9590
C	CALCULATE RHS OF DROPLET DYNAMICS EQUATIONS AND EVAPORATION	9600
C		9610
	IF(NEVAP.EQ.1) DIA=YY(1,NVAR)	9620
	R=U*((DA*DIA)*SQRT((VA-VD)**2+(WA-WD)**2))/VIS	9630
	IF(N3D.EQ.1) R=U*((DA*DIA)*SQRT((UA-UD)**2+(VA-VD)**2	9640
	+ (WA-WD)**2))/VIS	9650
	CDR=1.0+0.197*R**0.63+0.26E-3*R**1.38	9660
	S=18.*(R*DA)/(DIA*DD)	9670
	RU=(DA*DIA*U)/VIS	9680
	IF(NEVAP.EQ.0) GO TO 60	9690
	DY(NVAR)=-2.*(B/U)*(18.016/28.97)*(DV/DIA)*(DA/DD)*((PSWB-PV)	9700
	/(PA-PV))*(2.+6*SCN*(1./3.)*R**0.5)	9710
60	CONTINUE	9720
	DY(1)=YY(1,2)	9730
	DY(2)=((S*CDR)/RU)*(VA-VU)	9740
	DY(3)=YY(1,4)	9750
	DY(4)=((S*CDR)/RU)*(WA-WD)-(B*G)/U**2	9760
	IF(NEVAP.EQ.1) DY(2)=DY(2)-(3./DIA)*VD*DY(NVAR)	9770
	IF(NEVAP.EQ.1) DY(4)=DY(4)-(3./DIA)*WD*DY(NVAR)	9780
	IF(N3D.EQ.0) GO TO 70	9790
	DY(5)=YY(1,6)	9800
	DY(6)=((S*CDR)/RU)*(UA-UD)	9810
	IF(NEVAP.EQ.1) DY(6)=DY(6)-(3./DIA)*UD*DY(NVAR)	9820
70	RETURN	9830
	END	9840
C		9850
C	SUBROUTINE USERV.....USER SUPPLIED	9860
C		9870
C	WITH N2D=2 OR N3D=2 USERV IS CALLED BY DIFFUN FOR THE	9880
C	VELOCITY IN THE WAKE. THIS ALLOWS THE USER TO REPLACE	9890
C	THE CODE'S WAKE MODEL WITH ONE OF THEIR OWN. THE NEVAP	9900

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C      OPTION IS STILL AVAILABLE AS IS NOIST, NPRINT, AND NPL0T.          9910
C      THE PROPELLER AND TUNNEL MODELS ARE AVAILABLE TO THE              9920
C      USER BY INCLUDING THE PROPER INPUTS AND SUBROUTINE CALLS.         9930
C                                                                           9940
C      PASSED TO USERV IS THE NONDIMENSIONAL TIME, T, AND THE           9950
C      NONDIMENSIONAL POSITION X,Y,Z.  THE USER RETURNS THE             9960
C      CORRESPONDING NONDIMENSIONAL VELOCITIES UA,VA,WA.                9970
C                                                                           9980
C      SUBROUTINE USERV(T,X,Y,Z,UA,VA,WA)                                9990
C      RETURN                                                            10000
C      END                                                                10010
C                                                                           10020
C                                                                           10030
C      SUBROUTINE VEL2D: INPUT DIMENSIONLESS TIME AND POSITION(Y,Z)       10040
C      VEL2D RETURNS DIMENSIONLESS INDUCED VELOCITIES(UA,VA) FROM       10050
C      A VORTEX PAIR UNAFFECTED BY CROSSWIND                             10060
C                                                                           10070
C      SUBROUTINE VFL2D(T,Y,Z,VA,WA)                                     10080
C      COMMON /AREA1/ A,CL,B,U,G,ZY0,ZZ0,DA,VIS,VCW,D,DD                10090
C                                                                           10100
C      CALCULATE THE POSITION OF THE VORTICES                             10110
C                                                                           10120
C      C=3.141592654                                                    10130
C      C1=1/(ZY0**2)+1/(ZZ0**2)                                          10140
C      C2=-(C1*ZY0**2-2.0)/SQRT(C1*ZY0**2-1.0))                        10150
C      C3= -(C1*ZZ0**2-2.0)/SQRT(C1*ZZ0**2-1.0))                        10160
C      C4=C*(4.0*CL**2)/(CL**2)                                          10170
C      C5=C3*(4.0*CL**2)/(CL**2)                                         10180
C      IF(T.GT.C4) GO TO 10                                              10190
C      ZY=SQRT((((T*CL*C1)/(4.0*CL**2))**2)*C1-(C2*T*CL*C1**2)/(2.0*CL**2)+C1*C2**2+4.0*CL)-SQRT((((T*CL*C1)/(4.0*CL**2))**2)*C1-(C2*T*CL*C1**2)/(2.0*CL**2)+C1*C2**2+4.0*CL**2-4.0*CL**2*((T*CL*C1)/(4.0*CL**2))**2-(C2*T*CL*C1)/(2.0*CL**2)+C2**2+4.0)))/(2.0*CL**2)
C      GO TO 20
C 10 ZY=SQRT((((T*CL*C1)/(4.0*CL**2))**2)*C1-(C2*T*CL*C1**2)/(2.0*CL**2)+C1*C2**2+4.0*CL)-SQRT((((T*CL*C1)/(4.0*CL**2))**2)*C1-(C2*T*CL*C1**2)/(2.0*CL**2)+C1*C2**2+4.0*CL**2-4.0*CL**2*((T*CL*C1)/(4.0*CL**2))**2-(C2*T*CL*C1)/(2.0*CL**2)+C2**2+4.0)))/(2.0*CL**2)
C      GO TO 20
C 20 IF(T.LT.C5) GO TO 30
C      ZZ=SQRT((((T*CL*C1)/(4.0*CL**2))**2)*C1-(C3*T*CL*C1**2)/(2.0*CL**2)+C1*C3**2+4.0*CL)-SQRT((((T*CL*C1)/(4.0*CL**2))**2)*C1-(C3*T*CL*C1**2)/(2.0*CL**2)+C1*C3**2+4.0*CL**2-4.0*CL**2*((T*CL*C1)/(4.0*CL**2))**2-(C3*T*CL*C1)/(2.0*CL**2)+C3**2+4.0)))/(2.0*CL**2)
C      GO TO 40
C 30 ZZ=SQRT((((T*CL*C1)/(4.0*CL**2))**2)*C1-(C3*T*CL*C1**2)/(2.0*CL**2)+C1*C3**2+4.0*CL)-SQRT((((T*CL*C1)/(4.0*CL**2))**2)*C1-(C3*T*CL*C1**2)/(2.0*CL**2)+C1*C3**2+4.0*CL**2-4.0*CL**2*((T*CL*C1)/(4.0*CL**2))**2-(C3*T*CL*C1)/(2.0*CL**2)+C3**2+4.0)))/(2.0*CL**2)
C 40 CONTINUE
C      CALCULATE THE INDUCED VELOCITIES

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C      VA=(CL/(2.0*C*A))*((ZZ+Z)/((ZZ+Z)**2+(ZY-Y)**2)-(ZZ+Z)/((ZZ+Z)
&      **2+(ZY+Y)**2)-(ZZ-Z)/((ZZ-Z)**2+(ZY+Y)**2)+(ZZ-Z)/((ZZ-Z)**2
&      +(ZY-Y)**2))
      WA=(CL/(2.0*C*A))*((ZY-Y)/((ZZ+Z)**2+(ZY-Y)**2)+(ZY+Y)/((ZZ+Z)
&      **2+(ZY+Y)**2)-(ZY+Y)/((ZZ-Z)**2+(ZY+Y)**2)-(ZY-Y)/((ZZ-Z)**2
&      +(ZY-Y)**2))
      RETURN
      END
C
C      SUBROUTINE ROUND: CALCULATES THE INDUCED VELOCITIES (UA,WA)
C      DUE TO THE BOUND VORTEX AT A POINT (X,Y,Z)
C
      SUBROUTINE ROUND(X,Y,Z,UA,WA)
      COMMON /AREA1/ A,CL,B,U,G,ZY0,ZZ0,DA,VIS,VCW,D,DD
      RS=SQRT(X**2+(ZZ0-Z)**2)
      R6=SQRT(X**2+(ZZ0+Z)**2)
      UA=1.-(CL/(4.*3.141592654*A))*((1.-Y)/SQRT(R5**2+(1.-Y)**2)+
&      (1.+Y)/SQRT(R5**2+(1.+Y)**2))*((ZZ0-Z)/R5**2)+(1.-Y)/SQRT
&      (R6**2+(1.-Y)**2)+(1.+Y)/SQRT(R6**2+(1.+Y)**2))*((ZZ0+Z)
&      /R6**2))
      WA=(CL/(4.*3.141592654*A))*(-(1.-Y)/SQRT(R5**2+(1.-Y)**2)+
&      (1.+Y)/SQRT(R5**2+(1.+Y)**2))*(X/R5**2)+(1.-Y)/SQRT(R6**2
&      +(1.-Y)**2)+(1.+Y)/SQRT(R6**2+(1.+Y)**2))*(X/R6**2))
      RETURN
      END
C
C      SUBROUTINE PROPX: RETURNS THE X VELOCITY IN THE PROPELLER
C      SLIPSTREAM UAP DUE TO A PROP CENTERED AT (YP,ZP) AT A
C      POINT (Y,Z)
C
      SUBROUTINE PROPX(Y,Z,YP,ZP,UAP)
      COMMON /AREA1/ A,CL,B,U,G,ZY0,ZZ0,DA,VIS,VCW,D,DD
      COMMON /AREA5/ ZPROP2,PK1,PK2,PK3,PK4,RP
      R1=SQRT((Y-YP)**2+(Z-ZP)**2)
      IF(R1.LT.RP) GO TO 10
      UAP=U.
      RETURN
10  CONTINUE
      R2=PK1/RP
      UAP=(PK1*(1.-R2)+PK2*R2)*IN(R2*3.141592654)/U
      RETURN
      END
C
C      SUBROUTINE PROP2D: RETURNS DIMENSIONLESS PROPELLER INDUCED
C      VELOCITIES(V,W) AT POSITION (Y,Z) FOR A PROP AT (YP,ZP)
C      (ALL POSITIONS DIMENSIONLESS)
C
      SUBROUTINE PROP2D(Y,Z,YP,ZP,V,W)
      COMMON /AREA1/ A,CL,B,U,G,ZY0,ZZ0,DA,VIS,VCW,D,DD
      COMMON /AREA5/ ZPROP2,PK1,PK2,PK3,PK4,RP
      R1=SQRT((Y-YP)**2+(Z-ZP)**2)
      IF(R1.LT.RP) GO TO 10
      V=0.

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	W=0.	11010
	RETURN	11020
10	CONTINUE	11030
	R2=K1/R0	11040
	SV1=(PK3*(1.0-R2)+PK4*R2*SIN(R2*3.141592654))/U	11050
	V=((Z-Z0)/R1)*SV1	11060
	W=((Y-Y0)/R1)*SV1	11070
	RETURN	11080
	END	11090
C		11100
C		11110
C	SUBROUTINE VELTV: RETURNS INDUCED VELOCITIES (V,W) AT	11120
C	POSITION (Y,Z) FOR VORTICES LOCATED AT (ZYR,ZZR) AND	11130
C	(ZYL,ZZL) (ALL QUANTITIES DIMENSIONLESS)	11140
C		11150
	SUBROUTINE VELTV(ZYR,ZZR,ZYL,ZZL,Y,Z,V,W)	11160
	COMMON /AREA1/ A,CL,B,U,G,ZY0,ZZ0,DA,VIS,VCW,D,DD	11170
	COMPLEX GC,ZCR,ZCL,WZ,ZD	11180
	GAMMA=CL/(A*2.*3.141592654)	11190
	GC=CMPLX(0.,GAMMA)	11200
	ZD=CMPLX(Y,Z)	11210
	ZCR=CMPLX(ZYR,ZZR)	11220
	ZCL=CMPLX(ZYL,ZZL)	11230
	WZ=-GC*(1./(ZD-ZCR)-1./(ZD-ZCL)+1./(ZD-CONJG(ZCL)))	11240
	WZ=-1./(ZD-CONJG(ZCR))	11250
	V=REAL(WZ)	11260
	W=-AIMAG(WZ)	11270
	RETURN	11280
	END	11290
C		11300
C		11310
C	SUBROUTINE VELC: RETURNS INDUCED VELOCITIES (V,W) ON A	11320
C	VORTICE CORE AT (Y1,Z1) AND (Y2,Z2) (ALL QUANTITIES ARE	11330
C	DIMENSIONLESS)	11340
C		11350
	SUBROUTINE VELC(Y1,Z1,Y2,Z2,V,W)	11360
	COMMON /AREA1/ A,CL,B,U,G,ZY0,ZZ0,DA,VIS,VCW,D,DD	11370
	COMPLEX GC,ZC1,ZC2,WZ	11380
	GAMMA=CL/(2.*3.141592654*A)	11390
	GC=CMPLX(0.,GAMMA)	11400
	ZC2=CMPLX(Y2,Z2)	11410
	ZC1=CMPLX(Y1,Z1)	11420
	WZ=-GC*(-1./(ZC1-ZC2)+1./(ZC1-CONJG(ZC2))-1./(ZC1-CONJG(ZC1)))	11430
	V=REAL(WZ)	11440
	W=-AIMAG(WZ)	11450
	RETURN	11460
	END	11470
C		11480
C		11490
C	SUBROUTINE VORTUN: RETURNS INDUCED VELOCITIES (VV,WW)	11500
C	AT A POINT (ZY,ZZ) FOR THE TUNNEL CASE WHERE VORTICES	11510
C	ARE LOCATED AT (ZY,ZZ) AND (ZY,-ZZ) (ALL QUANTITIES ARE	11520
C	DIMENSIONLESS)	11530
C		11540
	SUBROUTINE VORTUN(ZY,ZZ,VV,WW)	11550

	COMMON /AREA1/ A,CL,B,U,G,ZY0,ZZ0,DA,VIS,VCW,D,DD	11560
	COMPLEX 71,W,CONST	11570
	C=3.141592654	11580
	DTUN=D/A	11590
	Z1=CMPLX(ZY,ZZ)	11600
	GAMMA=CL/(2.*C*A)	11610
	E=0.	11620
	CONST=CMPLX(E,GAMMA)	11630
	W=(0.,0.)	11640
C		11650
C	CALCULATING THE COMPLEX VELOCITY AT A VORTEX	11660
C		11670
	DO 10 I=1,21	11680
	N=I-11	11690
	IF(N.EQ.0) GO TO 10	11700
	W=W+CONST*(-(1./(N*DTUN))+(1./(2.*ZY+N*DTUN))-(1./(2.*Z1	11710
	& +N*DTUN))+(1./(Z1-CONJG(Z1)+N*DTUN)))	11720
10	CONTINUE	11730
	W=W+CONST*((1./(2.*ZY))-(1./(2.*Z1))+(1./(Z1-CONJG(Z1))))	11740
	VV=REAL(W)	11750
	WW=-AIMAG(W)	11760
	RETURN	11770
	END	11780
C		11790
C	TUNNEL SOLVES FOR THE INDUCED VELOCITY (VA,WA) AT A POINT IN TH	11800
C	TUNNEL (YD,ZD) AND A GIVEN VORTEX POSITION (ZY,+-ZZ).	11810
C	(ALL QUANTITIES ARE DIMENSIONLESS)	11820
C		11830
	SUBROUTINE TUNNEL(YD,ZD,ZY,ZZ,VA,WA)	11840
	COMMON /AREA1/ A,CL,B,U,G,ZY0,ZZ0,DA,VIS,VCW,D,DD	11850
	COMPLEX C,Z,CONST,W	11860
	CC=3.141592654	11870
	GAMMA=(CL*U*B)/A	11880
	CD=GAMMA/(2.*CC)	11890
	E=0.	11900
	CONST=CMPLX(E,CD)	11910
	F=CC/D	11920
	DYD=YD*B	11930
	DZD=ZD*B	11940
	DZY=ZY*B	11950
	DZZ=ZZ*B	11960
	C=CMPLX(DZY,DZZ)	11970
	Z=CMPLX(DYD,DZD)	11980
	W=-CONST*((CSIN(F*Z))**2-(CSIN(F*CONJG(C)))**2)/((CSIN(F*Z))**2	11990
	& -(CSIN(F*C))**2))*((2.*F*CSIN(F*Z)*CCOS(F*Z))*((CSIN(F*C))**2	12000
	& -(CSIN(F*CONJG(C)))**2))/((CSIN(F*Z))**2-(CSIN(F*CONJG(C)))**2	12010
	&)**2)	12020
	W=CONJG(W)/U	12030
	VA=REAL(W)	12040
	WA=AIMAG(W)	12050
	RETURN	12060
	END	12070
C		12080
C	SUBROUTINE SWATH: CALCULATES THE DISTRIBUTION FROM EACH NOZZLE	12090
C	AND SUMS THESE UP TO GENERATE A TOTAL DISTRIBUTION AND	12100

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C      ESTIMATES THE DRIFT. IF NPRINT=1 EACH NOZZLE'S DISTRIBUTION 12110
C      AND DRIFT IS OUTPUT 12120
C 12130
      SUBROUTINE SWATH(NCOL,NROW,Q,DIAMN,STDEV,YNOZ,DWIDTH,
&      NRUN,MET,WIDTH,DIST,DRIFT) 12140
      DIMENSION TRAJ(15,100),YNOZ(100),SIZE(15,100),NRUN(100),DIST(1) 12150
      DIMENSION DRIFT(100),YI(15),C(4,15),SIZ(15),SIZ1(15) 12160
      DIMENSION DIAG(15,100),S(15,5),MET(100),XLIST(4),SLIST(4) 12170
      DIMENSION DIAMN(100),STDEV(100),Q(100),C1(4,15),DLIST(4) 12180
      EXTERNAL CINT,BINT 12190
      COMMON /AREA2/ N2D,N3D,NEVAP,EPS,NPROP2,NCW,NTUN,NDIST,NPRINT 12200
      COMMON /AREA9/ SIZ,SIZ1,NR,C1,DIAMN,STDEVI 12210
      COMMON /AREA10/ TRAJ,DIAG,SIZE 12220
      DATA S/75*0./ 12230
      KJ=0 12240
      START=-WIDTH 12250
      NEWIDTH=100*WIDTH+1-START*100 12260
      DO 10 I=1,NCOL 12270
      NR=NRUN(I) 12280
      IF(NR.EQ.0) GO TO 200 12290
      IF(MET(I).EQ.1.OR.NR.LT.4) GO TO 100 12300
      IF MET(I)=0 THE I' TH NOZZLE'S DISTRIBUTION IS CALCULATED 12310
      USING A CUBIC SPLINE 12320
      IF(NPRINT.EQ.0) GO TO 46 12330
      WRITE (6,40) YNOZ(I) 12340
40  FORMAT('1',"YNOZ=","F5.3) 12350
      WRITE (6,45) 12360
45  FORMAT('0',"3(7X,"YG",11X,"DIA",5X,"CONCENTRATION") 12370
46  CONTINUE 12380
      DO 20 J=1,NR 12390
      K=NR-J+1 12400
      SIZ(J)=SIZE(K,I) 12410
      SIZ1(J)=DIAG(K,I) 12420
      YI(J)=1./(TRAJ(K,I)-YNOZ(I)) 12430
20  CONTINUE 12440
      IBC=0 12450
      AA=0. 12460
      BB=0. 12470
      IF(NEVAP.EQ.1) CALL CSPL(SIZ,SIZ1,NR,C1,S,IBC,AA,BB) 12480
      CALL CSPL(YI,SIZ,NR,C,S,IBC,AA,BB) 12490
      IF(ABS(TRAJ(NR,I)).LT.DWIDTH) GO TO 21 12500
      DELTAY=1./(DWIDTH-YNOZ(I)) 12510
      CALL CSFIND(YI,SIZ,NR,C,DELTAY,DRIFT(I),DUMI,DUMY) 12520
21  CONTINUE 12530
      CALCULATE THE DISTRIBUTION AND ADD TO DIST ARRAY, 12540
      OUTPUT TO 6 IF NPRINT=1 12550
      NPTS=(TRAJ(1,I)-START)*100+1 12560
      NEND=NEWIDTH 12570
      IF(TRAJ(NR,I).LT.TRAJ(1,I)) NEND=1 12580
      NSTART=NPTS 12590
      NSTOP=NEND 12600

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IF(NEND.NE.1) GO TO 25	12660
NSTART=NEND	12670
NSTOP=NPTS	12680
25 CONTINUE	12690
DIAOLD=10000.	12700
DO 30 KKK=NSTART,NSTOP	12710
KK=KKK	12720
IF(NEND.EQ.1) KK=NSTOP-(KKK-1)	12730
YG=START+(KK-1)*.01	12740
DELTAY=1./(YG-YN0Z(I))	12750
CALL CSFIND(YI,SIZ,NR,C,DELTAY,DIA,DDYG,DUMY)	12760
IF(DIA.GE.DIAOLD) WRITE(6,27) I,DIA	12770
27 FORMAT("0",,"*** CUBIC SPLINE PROBLEM NOZZLE NO.",I3," DIA=",	12780
& E13.6," ***")	12790
DIAOLD=DIA	12800
IF(DIA.LT.DRIFT(I)) GO TO 35	12810
IF(DIA.GT.SIZ(NR)) GO TO 30	12820
DVDD=(EXP(-.5*((DIA-DIAMN(I))/STDEV(I))**2))	12830
& /(STDEV(I)*SQRT(2.*3.141592654))	12840
IF(NEVAP.EQ.1) CALL CSFIND(SIZ,SIZ1,NR,C1,DIA,DIA1,DUMY,DUMMY)	12850
IF(NEVAP.EQ.1) DVDD=DVDD*(DIA1**3/DIA**3)	12860
DYG=-1./(YG-YN0Z(I))**2)	12870
DDYG=DDYG*DYG	12880
DIST1=ARS(Q(I)*DVDD*DDYG)	12890
DIST(KK)=DIST1+DIST(KK)	12900
IF(NPRINT.EQ.0) GO TO 30	12910
IF(DIST1.LT.(1.E-4/Q(I))) GO TO 30	12920
IF(KJ.EQ.3) KJ=0	12930
KJ=KJ+1	12940
XLIST(KJ)=YG	12950
DLIST(KJ)=DIST1	12960
SLIST(KJ)=DIA	12970
IF(KJ.EQ.3) WRITE(6,80) (XLIST(K1),SLIST(K1),DLIST(K1),K1=1,3)	12980
80 FORMAT(" ",3(E15.6,2E13.6))	12990
30 CONTINUE	13000
35 CONTINUE	13010
IF(KJ.NE.3.AND.NPRINT.EQ.1) WRITE(6,80) (XLIST(K1),SLIST(K1),	13020
& DLIST(K1),K1=1,KJ)	13030
GO TO 200	13040
100 CONTINUE	13050
C	13060
C IF MET(I)=1 THE I' TH NOZZLE'S DISTRIBUTION IS CALCULATED	13070
C USING A LINEAR FIT	13080
C	13090
KJ=0	13100
IF(NR.LT.2) GO TO 200	13110
IF(NPRINT.EQ.1) WRITE(6,105) YNOZ(I)	13120
105 FORMAT("1","YNOZ=",F5.3/"0",*(8X,"YG",7X,"CONCENTRATION")/)	13130
DO 110 J=1,NR	13140
K=NR-J+1	13150
SIZ(J)=SIZE(K,I)	13160
SIZ1(J)=DIAG(K,I)	13170
YI(J)=TRAJ(K,I)	13180
110 CONTINUE	13190
IF(NEVAP.EQ.0) GO TO 115	13200

IF(NR.GE.4) GO TO 112	13210
WRITE(6,113) I	13220
113 FORMAT("0","*** INSUFFICIENT DATA TO CALCULATE DISTRIBUTION ***")	13230
& " *** NOZZLE NO. ",I2," DISTRIBUTION SET = TO ZERO ***")	13240
GO TO 300	13250
112 CONTINUE	13260
IBC=0.	13270
AA=0.	13280
RB=0.	13290
CALL CSPL(SIZ,SIZ1,NR,C1,S,IBC,AA,88)	13300
115 CONTINUE	13310
C	13320
C	13330
C	13340
C	13350
YMIN=100	13360
YMAX=-100	13370
DO 140 K=1,NP	13380
IF(YI(K).LT.YMIN) YMIN=YI(K)	13390
IF(YI(K).GT.YMAX) YMAX=YI(K)	13400
120 CONTINUE	13410
ISTART=(YMIN-START)*100+1	13420
DO 130 K=ISTART,NWIDTH	13430
YG=(K-1)*.01+START	13440
IF(ABS(YG).GT.DWIDTH) GO TO 130	13450
DIST1=0.	13460
IF(YG.GT.YMAX) GO TO 199	13470
J1=NM-1	13480
DO 140 J=1,J1	13490
IF(YG.LT.YI(J).AND.YG.LT.YI(J+1)) GO TO 140	13500
IF(YG.GT.YI(J).AND.YG.GT.YI(J+1)) GO TO 140	13510
SLOPE=(SIZ(J+1)-SIZ(J))/(YI(J+1)-YI(J))	13520
DIA=SIZ(J)+SLOPE*(YG-YI(J))	13530
DVDU=(EXP(-.5*((DIA-DIAMN(I))/STDEV(I))**2))	13540
& /(STDEV(I)*SQRT(2.*3.141592654))	13550
IF(NEVAR.EQ.0) GO TO 135	13560
CALL CSFIND(SIZ,SIZ1,NR,C1,DIA,DIA1,DMY,DMY)	13570
DVDU=DVDD*(DIA1**3/DIA**3)	13580
135 CONTINUE	13590
DIST1=DIST1+Q(I)*ABS(SLOPE*DVDD)	13600
140 CONTINUE	13610
DIST(K)=DIST1+DIST(K)	13620
IF(NPRINT.EQ.0) GO TO 130	13630
IF(KJ.EQ.4) KJ=0	13640
KJ=KJ+1	13650
XLIST(KJ)=YG	13660
OLIST(KJ)=DIST1	13670
IF(KJ.EQ.4) WRITE(6,141) (XLIST(K1),OLIST(K1),K1=1,4)	13680
141 FORMAT(" ",4(E16.6,E14.6))	13690
130 CONTINUE	13700
199 CONTINUE	13710
IF(KJ.NE.4.AND.NPRINT.EQ.1) WRITE(6,141) (XLIST(K1),OLIST(K1),K1=1,KJ)	13720
&	13730
200 CONTINUE	13740
C	13750

C	DRIFT ESTIMATION	13760
C		13770
	DRIFT=DRIFT(I)	13780
	IF (DRIFT(I).EQ.0) GO TO 300	13790
	IF (DRIFT(I).NE.SIZE(1,I)) GO TO 205	13800
	DRIFT(I)=100.	13810
	GO TO 350	13820
205	CONTINUE	13830
	AA=0.	13840
	BB=(DRIFT(I)-DIAMN(I))/STDEV(I)	13850
	ISEG=ABS(BB*20)/2	13860
	ISEG=ISEG*2	13870
	CALL SIMP(AA,BB,ISEG,CINT,ANS)	13880
	ANS=ANS*SQRT(1./(2.*3.141592654))	13890
	DRIFT(I)=(.50+ANS)*100.	13900
300	CONTINUE	13910
	IF (NEVAP.EQ.0) GO TO 350	13920
C		13930
C	INCLUDE THE EFFECT OF EVAPORATION IN THE DRIFT	13940
C		13950
	IF (NR.GE.4) GO TO 320	13960
	WRITE(6,330) I	13970
330	FORMAT("0","***INSUFFICIENT DATA TO ESTIMATE DRIFT ***"/	13980
	& " *** NOZZLE NO. ",I2," DRIFT SET EQUAL TO ZERO ***")	13990
	DRIFT(I)=0.	14000
	GO TO 10	14010
320	CONTINUE	14020
	AA=DRIFT	14030
	BB=SIZE(NR)	14040
	ISEG=IFX((SIZE(NR)-SIZE(1))/20)*20	14050
	IF (ISEG.LT.2) ISEG=2	14060
	DIAMNI=DIAMN(I)	14070
	STDEVI=STDEV(I)	14080
	CALL SIMP(AA,BB,ISEG,CINT,ANS)	14090
	EVAP=1.-ANS	14100
	DRIFT(I)=EVAP*100.+DRIFT(I)	14110
350	CONTINUE	14120
	IF (NPRINT.EQ.1) WRITE (6,60) DRIFT(I)	14130
60	FORMAT("0","DRIFT=",E14.6)	14140
10	CONTINUE	14150
	RETURN	14160
	END	14170
C		14180
C	SIMP IS A SIMPSON'S RULE INTEGRATION ROUTINE	14190
C		14200
C	AA,BB ARE THE LIMITS OF INTEGRATION	14210
C	ISEG IS THE EVEN NUMBER OF SEGMENTS AA TO BB IS DIVIDED INTO	14220
C	CINT IS THE SUBROUTINE WHICH EVALUATES THE INTEGRAND	14230
C	ANS RETURNS THE ANSWER	14240
C		14250
	SUBROUTINE SIMP(AA,BB,ISEG,CINT,ANS)	14260
	DELTA=(BB-AA)/ISEG	14270
	ISEG1=ISEG-1	14280
	SUM=0.	14290
	DO 10 I=1,ISEG1	14300

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FAC=*.
K=I/2
IF((K*2).EQ.I) FAC=2.
X=AA+I*DELTA
CALL CINT(X,Y)
SUM=SUM+Y*FAC
10 CONTINUE
CALL CINT(AA,Y1)
CALL CINT(99,Y2)
SUM=SUM+Y1+Y2
ANS=(DELTA/3.)*SUM
RETURN
END

C
C      CINT EVALUATES INTEGRAND WHEN NEVAP=0
C
SUBROUTINE CINT(X,Y)
Y=EXP(-.50*X**2)
RETURN
END

C
C      BINT EVALUATES INTEGRAND WHEN NEVAP=1
C
SUBROUTINE BINT(X,Y)
DIMENSION SIZ(15),SIZ1(15),C1(4,15)
COMMON /AREA9/ SIZ,SIZ1,NR,C1,DIAMNI,STDEVI
CALL CSFIND(SIZ,SIZ1,NR,C1,X,DIA1,DUMY,DUMMY)
DIA=DIA
DVDD=(EXP(-.5*((DIA-DIAMNI)/STDEVI)**2))
      /(STDEVI*SQRT(2.*3.141592654))
Y=(DIA1/DIA)**3*DVDD
RETURN
END

C
C      INCON EVALUATES THE INITIAL VELOCITY OF THE DROPLET
C
C      TO CORRECT FOR THE EFFECT OF THE BOUND VORTEX THE DROPLET IS
C      GIVEN AN INITIAL VELOCITY EQUAL TO ITS TERMINAL VELOCITY AT
C      THAT POINT IN THE FLUX FIELD
C
SUBROUTINE INCON(YY,DIA,NVAR)
DIMENSION YY(13,NVAR)
COMMON /AREA1/ A,CL,B,U,G,ZY0,ZZ0,DA,VIS,VCW,D,DD
COMMON /AREA2/ N2D,N3D,NEVAP,EPS,NPROP2,NCW,NTUN,NDIST,NPRINT
C=3.141592654
ZY=ZY0
ZZ=ZZ0
Y=YY(1,1)
Z=YY(1,3)
IF(NTUN.EQ.1) GO TO 5
VA=(CL/(2.0*C*A))*((ZZ+Z)/((ZZ+Z)**2+(ZY-Y)**2)-(ZZ+Z)/((ZZ+Z)
& **2+(ZY+Y)**2)-(ZZ-Z)/((ZZ-Z)**2+(ZY+Y)**2)+(ZZ-Z)/((ZZ-Z)**2
& +(ZY-Y)**2))
WA=(CL/(2.0*C*A))*((ZY-Y)/((ZZ+Z)**2+(ZY-Y)**2)+(ZY+Y)/((ZZ+Z)
& **2+(ZY+Y)**2)-(ZY+Y)/((ZZ-Z)**2+(ZY+Y)**2)-(ZY-Y)/((ZZ-Z)**2

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5	+(ZY-Y)**2))	14860
5	CONTINUE	14870
	IF (NTUN.EQ.1) CALL TUNVEL(Y,Z,ZY,ZZ,VA,WA)	14880
	S=18.*(R*DA)/(DIA*DD)	14890
	RU=(DA*PIA*U)/VIS	14900
	VD=VA	14910
	YY(1,2)=VA	14920
	WDL=WA	14930
	WDR=WA	14940
	DO 10 J=1,100	14950
	WDR=WDR+WA	14960
	R=U*ABS(((DA*DIA)*(WA-WDR))/VIS)	14970
	CDR=1.0+0.197*R**0.63+0.26E-03*R**1.38	14980
	FR=((S*CDR)/RU)*(WA-WDR)-(B*G)/U**2	14990
	IF (FR.GT.0.F0) GO TO 40	15000
10	CONTINUE	15010
	WRITE (6,20)	15020
20	FORMAT("n","**** WDR EXCEEDED DO LOOP IN INCON ****")	15030
	WRITE (6,30)	15040
30	FORMAT("1","**** RIGHT ENDPOINT IN INCON INCORRECT ****")	15050
	YY(1,4)=0.E0	15060
	GO TO 90	15070
40	CONTINUE	15080
C		15090
C	SOLVE FOR THE VELOCITY BY THE METHOD OF BISECTION	15100
C		15110
	DO 50 I=1,500	15120
	WDM=(WDL+WDR)/2.E0	15130
	R=ABS(((DA*DIA)*(WA-WDM))/VIS)	15140
	R=R*U	15150
	CDR=1.0+0.197*R**0.63+0.26E-03*R**1.38	15160
	FM=((S*CDR)/RU)*(WA-WDM)-(B*G)/U**2	15170
	FF=ABS(FM)	15180
	IF (FF.LT.EPS) GO TO 70	15190
	IF (FM.GT.0.E0) WDR=WDM	15200
	IF (FM.LT.0.E0) WDL=WDM	15210
50	CONTINUE	15220
	WRITE (6,60)	15230
60	FORMAT("1","**** DO LOOP PARAMETER IN INCON EXCEEDED ****")	15240
	YY(1,4)=0.E0	15250
	GO TO 90	15260
70	YY(1,4)=WDM	15270
	IF (NUIST.EQ.0) WRITE (6,80) VD,WDM	15280
80	FORMAT("n","INCON VALUES",T25,"VD=",E13.6,T45,"WD=",E13.6//)	15290
90	RETURN	15300
	END	15310
C		15320
C	YZFIN EVALUATES THE DROPLET LOCATION AND VELOCITY	15330
C	WHEN IT INTERSECTS THE GROUND PLANE	15340
C	VALUES FOR THE NEXT TO LAST TIME-STEP ARE IN THE ARRAY	15350
C	YY1 AND T1. THE VALUES FOR THE LAST STEP YY AND T ARE	15360
C	REPLACED BY THE LINEARLY INTERPOLATED FINAL VALUES	15370
C	TO BE RETURNED	15380
C		15390
	SUBROUTINE YZFIN(YY,YY1,NVAR,T,T1,L)	15400

DIMENSION YY(13,NVAR),YY1(NVAR),IRUN(50)	15410
COMMON /AREA1/ A,CL,B,U,G,ZY0,ZZ0,DA,VIS,VCW,D,DD	15420
COMMON /AREA2/ N2D,N3D,NEVAP,EPS,NPROP2,NCW,NTUN,NDIST,NPRINT	15430
COMMON /AREA6/ JJ,IRUN,ICOUNT	15440
CONV=(0.E0-YY1(3))/(YY(1,3)-YY1(3))	15450
DO 10 I=1,NVAR	15460
YY(1,I)=(YY(1,I)-YY1(I))*CONV+YY1(I)	15470
10 CONTINUE	15480
YY(1,3)=0.	15490
T=((1-T1)*CONV)+T1	15500
IF(NDIST.EQ.1) GO TO 20	15510
WRITE(8,30) T,YY(1,1),YY(1,3)	15520
30 FORMAT(3E13,6)	15530
20 RETURN	15540
END	15550
C	15560
C SUBROUTINE CSPL	15570
C CSPL FITS A CUBIC SPLINE THROUGH THE M INPUT POINTS (X,Y).	15580
C THE EQUATION FOR THE CUBIC WHEN X IS GREATER THAN X(I) AND	15590
C LESS THAN X(I+1) IS:	15600
C $Y(X)=C(1,I)*(X(I+1)-X)**3+C(2,I)*(X-X(I))**3+C(3,I)*$	15610
C $(X(I+1)-X)+C(4,I)*(X-X(I))$	15620
C	15630
C PARAMETERS:	15640
C	15650
C X AN ARRAY INPUT AND DIMENSIONED X(M) CONTAINING THE	15660
C INDEPENDENT VARIABLE IN INCREASING ORDER	15670
C Y AN ARRAY INPUT AND DIMENSIONED Y(M) CONTAINING THE	15680
C DEPENDENT VARIABLE	15690
C M NUMBER OF (X,Y) INPUT	15700
C C AN ARRAY OUTPUT WITH DIMENSION C(4,M-1) CONTAINING	15710
C THE DESIRED COEFFICIENTS	15720
C S A WORK ARRAY DIMENSIONED S(M,5)	15730
C IBC =0 SECOND DERIVATIVE OF Y WRT X SET EQUAL TO A AT	15740
C X(1) AND B AT X(M)	15750
C =1 FIRST DERIVATIVE OF Y WRT X SET EQUAL TO A AT	15760
C X(1) AND B AT X(M)	15770
C A,B INPUT VALUE OF FIRST OR SECOND DERIVATIVE DEPENDING	15780
C ON IRC	15790
C	15800
C PROGRAMED BY M. BRAGG AARL/OSU MARCH,1980	15810
C	15820
C SUBROUTINE CSPL(X,Y,M,C,S,IBC,A,B)	15830
C DIMENSION X(1),Y(1),C(4,1),S(M,5)	15840
C DO 4 K2=1,5	15850
C DO 1 K1=1,M	15860
1 S(K1,K2)=0.	15870
2 CONTINUE	15880
IF(IBC.NE.0) GO TO 10	15890
S(1,4)=1.	15900
S(M,4)=1.	15910
S(1,4)=A	15920
S(M,4)=B	15930
GO TO 20	15940
10 CONTINUE	15950

S(1,2)=2.	15960
S(1,3)=1.	15970
S(1,4)=6.*((Y(2)-Y(1))/((X(2)-X(1))*2))-(6.*A/(X(2)-X(1)))	15980
S(M,1)=1.	15990
S(M,2)=2.	16000
S(M,4)=6.*((Y(M)-Y(M-1))/((X(M)-X(M-1))*2))+(6.*B/(X(M)-X(M-1)))	16010
20 CONTINUE	16020
M1=M-1	16030
DO 30 I=2,M1	16040
DX=X(I+1)-X(I)	16050
DX1=X(I)-X(I-1)	16060
S(I,1)=DX1/DX	16070
S(I,2)=(2.*DX+2.*DX1)/DX	16080
S(I,3)=1.	16090
S(I,4)=6.*(((Y(I+1)-Y(I))/DX*2)-((Y(I)-Y(I-1))/(DX*DX1)))	16100
30 CONTINUE	16110
C	16120
C USE THOMAS ALGORITHM TO SOLVE TRI-DIAGONAL SYSTEM	16130
C	16140
DO 40 I=2,M	16150
S(I,2)=S(I,2)-S(I,1)*S(I-1,3)/S(I-1,2)	16160
S(I,4)=S(I,4)-S(I,1)*S(I-1,4)/S(I-1,2)	16170
40 CONTINUE	16180
S(M,5)=S(M,4)/S(M,2)	16190
DO 50 I=2,M	16200
J=M-I+1	16210
S(J,5)=(S(J,4)-S(J,3)*S(J+1,5))/S(J,2)	16220
50 CONTINUE	16230
DO 60 I=1,M1	16240
DX=X(I+1)-X(I)	16250
C(1,I)=S(I,5)/(6.*DX)	16260
C(2,I)=S(I+1,5)/(6.*DX)	16270
C(3,I)=(Y(I)/DX)-((S(I,5)*DX)/6.)	16280
C(4,I)=(Y(I+1)/DX)-((S(I+1,5)*DX)/6.)	16290
60 CONTINUE	16300
RETURN	16310
END	16320
C	16330
C SUBROUTINE CSFIND	16340
C	16350
C SCFIND USES THE X,Y,M,C FROM A CALL TO CSPL AND RETURNS FOR	16360
C ANY INDEPENDENT VARIABLE XX; THE CUBIC SPLINE VALUE OF THE	16370
C DEPENDENT VARIABLE YY, THE FIRST DERIVATIVE YP, AND THE	16380
C SECOND DERIVATIVE Y2P.	16390
C	16400
SUBROUTINE CSFIND(X,Y,M,C,XX,YY,YP,Y2P)	16410
DIMENSION X(1),Y(1),C(4,1)	16420
M1=M-1	16430
IF(X(1).GT.X(M)) GO TO 20	16440
DO 10 I=1,M1	16450
IF(X(I).LE.XX.AND.X(I+1).GT.XX) GO TO 40	16460
10 CONTINUE	16470
IF(XX.LT.X(1)) I=1	16480
IF(XX.GE.X(M)) I=M1	16490
GO TO 40	16500

20	CONTINUE	16510
	DO 30 I=1,M1	16520
	IF(X(I).GE.XX.AND.X(I+1).LT.XX) GO TO 40	16530
30	CONTINUE	16540
	IF(XX.GT.X(1)) I=1	16550
	IF(XX.LT.X(M)) I=M1	16560
40	CONTINUE	16570
	YY=C(1,I)*(X(I+1)-XX)**3+C(2,I)*(XX-X(I))**3+C(3,I)*(X(I+1)-XX)	16580
	& +C(4,I)*(XX-X(I))	16590
	YP=-3.*C(1,I)*(X(I+1)-XX)**2+3.*C(2,I)*(XX-X(I))**2-C(3,I)	16600
	& +C(4,I)	16610
	Y2P=0.*C(1,I)*(X(I+1)-XX)+6.*C(2,I)*(XX-X(I))	16620
	RETURN	16630
	END	16640
C		16650
C	SUBROUTINE PLOT1	16660
C		16670
C	PLOT1 IS CALLED IF NPL0T=1 AND PLOTS PARTICLE TRAJECTORIES	16680
C	IF NDIST=0 AND THE FINAL DISTRIBUTION IF NDIST=1	16690
C		16700
	SUBROUTINE PLOT1(TITLE)	16710
	DIMENSION IRUN(50),YPL0T(1001),ZPL0T(1001),TITLE(13)	16720
	COMMON /AREA2/ N2D,N3D,NEVAP,EPS,NPROP2,NCW,NTUN,NDIST,NPRINT	16730
	COMMON /AREA6/ JJ,IRUN,ICOUNT	16740
	NT0T=JJ	16750
	REWIND 8	16760
	CALL PLOT1(120,4)	16770
	IF(NDIST.EQ.1) GO TO 200	16780
	DO 10 I=1,NT0T	16790
	READ(8,16) T,YPL0T(I),ZPL0T(I)	16800
16	FORMAT(3F13,6)	16810
10	CONTINUE	16820
	ZMAX=0.	16830
	DO 20 I=1,NT0T	16840
	IF(ZPL0T(I).GT.ZMAX) ZMAX=ZPL0T(I)	16850
20	CONTINUE	16860
	YPL0T(NT0T+1)=0.	16870
	NT0T1=NT0T+1	16880
	CALL SCALE(YPL0T,8.,NT0T1,FY,DY)	16890
	FZ=0.	16900
	DZ=0Y	16910
	IZ=ZMAX/DY+1.	16920
	IF(IZ.LE.6) GO TO 30	16930
	CALL SCALE(ZPL0T,6.,NT0T,FZ,DZ)	16940
	DY=0Y	16950
30	CONTINUE	16960
	AIZ=IZ	16970
	CALL PLOT(0.5,1.5,-3)	16980
	CALL AXIS(0.,0.,'2Y/B',-4,8.,0.,FY,DY)	16990
	CALL AXIS(0.,0.,'2Z/B',4,AIZ,90.,FZ,DZ)	17000
	CALL SYMBOL(1.0,6.5,.14,TITLE,0.,36)	17010
	K=1	17020
	DO 40 I=1,ICOUNT	17030
	II=IRUN(I)+K-1	17040
	DO 50 J=K,II	17050

JJ=J+K+1	17060
YPLUT(JJ)=YPLUT(J)	17070
ZPLUT(JJ)=ZPLUT(J)	17080
50 CONTINUE	17090
NLAST=IRUN(I)	17100
CALL LINE(YPLUT,FY,DY,ZPLUT,FZ,DZ,NLAST,0,1)	17110
K=K+IRUN(I)	17120
40 CONTINUE	17130
CALL PLOT(11.,-1.5,999)	17140
RETURN	17150
200 CONTINUE	17160
YMAX=-100.	17170
YMIN=100.	17180
DO 420 I=1,JJ	17190
READ(8,230) YPLUT(I),ZPLUT(I)	17200
230 FORMAT(2F13.6)	17210
IF(ZPLUT(I).GT.1.E-2.AND.YPLUT(I).LT.YMIN) YMIN=YPLUT(I)	17220
IF(ZPLUT(I).GT.1.E-2.AND.YPLUT(I).GT.YMAX) YMAX=YPLUT(I)	17230
220 CONTINUE	17240
CALL PLOT(0.,1.5,-3)	17250
K=0	17260
DO 410 I=1,JJ	17270
IF(YPLUT(I).LT.YMIN) GO TO 210	17280
IF(YPLUT(I).GT.YMAX) GO TO 210	17290
K=K+1	17300
YPLUT(K)=YPLUT(I)	17310
ZPLUT(K)=ZPLUT(I)	17320
210 CONTINUE	17330
CALL SCALE(YPLUT,8.,K,FX,DX)	17340
CALL SCALE(ZPLUT,6.,K,FD,DD)	17350
CALL AXIS(0.,0.,12Y/B',-4,8.,0.,FX,DX)	17360
CALL AXIS(0.,0.,1CONCENTRATION',13,6.,90.,FD,DD)	17370
CALL SYMROL(1.0,6.5.,14,TITLE,0.,36)	17380
CALL LINE(YPLUT,FX,DX,ZPLUT,FD,DD,K,0,1)	17390
CALL PLOT(11.,-1.5,999)	17400
RETURN	17410
END	17420

C* THIS SUBROUTINE INTEGRATES A SET OF N ORDINARY DIFFERENTIAL FIRST	*	10
C* ORDER EQUATIONS OVER ONE STEP OF LENGTH H AT EACH CALL. H CAN BE	*	20
C* SPECIFIED BY THE USER FOR EACH STEP, BUT IT MAY BE INCREASED OR	*	30
C* DECREASED BY DIFSUB WITHIN THE RANGE HMIN TO HMAX IN ORDER TO	*	40
C* ACHIEVE AS LARGE A STEP AS POSSIBLE WHILE NOT COMMITTING A SINGLE	*	50
C* STEP ERROR WHICH IS LARGER THAN EPS IN THE L-2 NORM, WHERE EACH	*	60
C* COMPONENT OF THE ERROR IS DIVIDED BY THE COMPONENTS OF YMAX.	*	70
C*	*	80
C* THE PROGRAM REQUIRES FOUR SUBROUTINES NAMED	*	90
C* DIFFUN(N,T,Y,DY) --USER-SUPPLIED--	*	100
C* DECOMP(N,M,PW,IP) --RENAMED TO NDI01Z IN THIS SOURCE--	*	110
C* SOLVE(N,M,PW,CSAVE(I,1),IP) --RENAMED TO NDI02Z--	*	120
C* PEDEKV(T,Y,PW,M) --USER-SUPPLIED--	*	130
C* THE FIRST, DIFFUN, EVALUATES THE DERIVATIVES OF THE DEPENDENT	*	140
C* VARIABLES STORED IN Y(I,I) FOR I = 1 TO N, AND STORES THE	*	150
C* DERIVATIVES IN THE ARRAY DY. THE NEXT TWO ARE CALLED ONLY IF THE	*	160
C* METHOD FLAG MF IS SET TO 1 OR 2 FOR STIFF METHODS. DECOMP IS A	*	170
C* STANDARD LU DECOMPOSITION WITH PIVOTING THAT DECOMPOSES THE MATRIX	*	180
C* PW, LEAVING THE PIVOTS IN THE INTEGER ARRAY IP. M IS THE DECLARED	*	190
C* SIZE OF PW. IP(N) IS SET TO 0 IF PW IS SINGULAR. SOLVE PERFORMS	*	200
C* BACK SUBSTITUTION ON THE CONTENTS OF CSAVE(I,1), LEAVING THE	*	210
C* RESULTS THERE.	*	220
C* PEDEKV IS USED ONLY IF MF IS 1, AND COMPUTES THE PARTIAL	*	230
C* DERIVATIVES OF THE DIFFERENTIAL EQUATIONS AS DESCRIBED UNDER THE	*	240
C* MF PARAMETER.	*	250
C*	*	260
C* THE TEMPORARY STORAGE SPACE IS PROVIDED BY THE CALLER IN THE	*	270
C* INTEGER ARRAY IP, THE ARRAY PW, AND THE	*	280
C* ARRAYS SAVE AND CSAVE. THE ARRAY PW IS USED ONLY TO HOLD	*	290
C* THE MATRIX OF THE SAME NAME, AND SAVE IS USED TO SAVE THE VALUES	*	300
C* OF Y IN CASE A STEP HAS TO BE REPEATED, BUT CSAVE IS USED TO HOLD	*	310
C* SEVERAL ARRAYS. THE REGIONS USED ARE	*	320
C* CSAVE(I,1) IS USED MAINLY TO HOLD THE CORRECTION TERMS IN THE	*	330
C* CORRECTOR LOOP, AND HOLDS THE DERIVATIVES DURING	*	340
C* JACOBIAN EVALUATIONS.	*	350
C* CSAVE(I,2) IS USED TO SAVE THE VALUES OF THE SUMS OF ALL OF THE	*	360
C* CORRECTION TERMS IN THE PREVIOUS STEP AFTER THEY	*	370
C* HAVE BEEN ACCUMULATED IN THE ARRAY ERROR IN THE	*	380
C* CURRENT STEP. THIS ENABLES THE BACKWARDS DIFFERENCE	*	390
C* OF ERROR TO BE FORMED. IT IS USED TO ESTIMATE THE	*	400
C* STEP SIZE FOR ONE ORDER HIGHER THAN CURRENT.	*	410
C* CSAVE(I,3) IS USED TO STORE THE DERIVATIVES WHEN THEY ARE	*	420
C* COMPUTED BY DIFFUN.	*	430
C*	*	440
C* THE PARAMETERS TO THE SUBROUTINE DIFSBM HAVE	*	450
C* THE FOLLOWING MEANINGS..	*	460
C*	*	470
C* N THE NUMBER OF FIRST ORDER DIFFERENTIAL EQUATIONS. N	*	480
C* MAY BE DECREASED ON LATER CALLS IF THE NUMBER OF	*	490
C* ACTIVE EQUATIONS REDUCES, BUT IT MUST NOT BE	*	500
C* INCREASED WITHOUT CALLING WITH JSTART = 0.	*	510
C* T THE INDEPENDENT VARIABLE.	*	520
C* Y A 13 BY N ARRAY CONTAINING THE DEPENDENT VARIABLES AND	*	530
C* THEIR SCALED DERIVATIVES. Y(J+1,I) CONTAINS	*	540
C* THE J-TH DERIVATIVE OF Y(I) SCALED BY	*	550

C*	H**J/FACTORIAL(J) WHERE H IS THE CURRENT	* 560
C*	STEP SIZE. ONLY Y(1:I) NEED BE PROVIDED BY	* 570
C*	THE CALLING PROGRAM ON THE FIRST ENTRY.	* 580
C*	IF IT IS DESIRED TO INTERPOLATE TO NON MESH POINTS	* 590
C*	THSE VALUES CAN BE USED. IF THE CURRENT STEP SIZE	* 600
C*	IS H AND THE VALUE AT T + E IS NEEDED, FORM	* 610
C*	S = E/H, AND THEN COMPUTE	* 620
C*	NQ	* 630
C*	Y(I)(T+E) = SUM Y(J+1:I)*S**J	* 640
C*	J=0	* 650
C*	SAVE A BLOCK OF AT LEAST 13*N FLOATING POINT LOCATIONS.	* 660
C*	CSAVE N*3 FLOATING POINT LOCATIONS USED BY THE SUBROUTINES.	* 670
C*	H THE STEP SIZE TO BE ATTEMPTED ON THE NEXT STEP.	* 680
C*	H MAY BE ADJUSTED UP OR DOWN BY THE PROGRAM	* 690
C*	IN ORDER TO ACHIEVE AN ECONOMICAL INTEGRATION.	* 700
C*	HOWEVER, IF THE H PROVIDED BY THE USER DOES	* 710
C*	NOT CAUSE A LARGER ERROR THAN REQUESTED, IT	* 720
C*	WILL BE USED. TO SAVE COMPUTER TIME, THE USER IS	* 730
C*	ADVISED TO USE A FAIRLY SMALL STEP FOR THE FIRST	* 740
C*	CALL. IT WILL BE AUTOMATICALLY INCREASED LATER.	* 750
C*	HMIN THE MINIMUM STEP SIZE THAT WILL BE USED FOR THE	* 760
C*	INTEGRATION. NOTE THAT ON STARTING THIS MUST BE	* 770
C*	MUCH SMALLER THAN THE AVERAGE H EXPECTED SINCE	* 780
C*	A FIRST ORDER METHOD IS USED INITIALLY.	* 790
C*	HMAX THE MAXIMUM SIZE TO WHICH THE STEP WILL BE INCREASED	* 800
C*	EPS THE ERROR TEST CONSTANT. SINGLE STEP ERROR ESTIMATES	* 810
C*	DIVIDED BY YMAX(I) MUST BE LESS THAN THIS	* 820
C*	IN THE EUCLIDEAN NORM. THE STEP AND/OR ORDER IS	* 830
C*	ADJUSTED TO ACHIEVE THIS.	* 840
C*	MF THE METHOD INDICATOR. THE FOLLOWING ARE ALLOWED..	* 850
C*	0 AN ADAMS PREDICTOR CORRECTOR IS USED.	* 860
C*	1 A MULTI-STEP METHOD SUITABLE FOR STIFF	* 870
C*	SYSTEMS IS USED. IT WILL ALSO WORK FOR	* 880
C*	NON STIFF SYSTEMS. HOWEVER THE USER	* 890
C*	MUST PROVIDE A SUBROUTINE PEDERV WHICH	* 900
C*	EVALUATES THE PARTIAL DERIVATIVES OF	* 910
C*	THE DIFFERENTIAL EQUATIONS WITH RESPECT	* 920
C*	TO THE Y'S. THIS IS DONE BY CALL	* 930
C*	PEDERV(T,Y,PW,M). PW IS AN N BY N ARRAY	* 940
C*	WHICH MUST BE SET TO THE PARTIAL OF	* 950
C*	THE I-TH EQUATION WITH RESPECT	* 960
C*	TO THE J DEPENDENT VARIABLE IN PW(I,J).	* 970
C*	PW IS ACTUALLY STORED IN AN M BY M	* 980
C*	ARRAY WHERE M IS THE VALUE OF N USED ON	* 990
C*	THE FIRST CALL TO THIS PROGRAM.	* 1000
C*	2 THE SAME AS CASE 1, EXCEPT THAT THIS	* 1010
C*	SUBROUTINE COMPUTES THE PARTIAL	* 1020
C*	DERIVATIVES BY NUMERICAL DIFFERENCING	* 1030
C*	OF THE DERIVATIVES. HENCE PEDERV IS	* 1040
C*	NOT CALLED.	* 1050
C*	YMAX AN ARRAY OF N LOCATIONS WHICH CONTAINS THE MAXIMUM	* 1060
C*	OF EACH Y SEEN SO FAR. IT SHOULD NORMALLY BE SET TO	* 1070
C*	1 IN EACH COMPONENT BEFORE THE FIRST ENTRY. (SEE THE	* 1080
C*	DESCRIPTION OF EPS.)	* 1090
C*	ERROR AN ARRAY OF N ELEMENTS WHICH CONTAINS THE ESTIMATED	* 1100

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C*          ONE STEP ERROR IN EACH COMPONENT. * 1110
C* KFLAG    A COMPLETION CODE WITH THE FOLLOWING MEANINGS.. * 1120
C*          +1 THE STEP WAS SUCCESSFUL. * 1130
C*          -1 THE STEP WAS TAKEN WITH H = HMIN, BUT THE * 1140
C*              REQUESTED ERROR WAS NOT ACHIEVED. * 1150
C*          -2 THE MAXIMUM ORDER SPECIFIED WAS FOUND TO * 1160
C*              BE TOO LARGE. * 1170
C*          -3 CORRECTOR CONVERGENCE COULD NOT BE * 1180
C*              ACHIEVED FOR H .GT. HMIN. * 1190
C*          -4 THE REQUESTED ERROR IS SMALLER THAN CAN * 1200
C*              BE HANDLED FOR THIS PROBLEM. * 1210
C* JSTART    AN INPUT INDICATOR WITH THE FOLLOWING MEANINGS.. * 1220
C*          -1 REPEAT THE LAST STEP WITH A NEW H * 1230
C*          0 PERFORM THE FIRST STEP. THE FIRST STEP * 1240
C*              MUST BE DONE WITH THIS VALUE OF JSTART * 1250
C*              SO THAT THE SUBROUTINE CAN INITIALIZE * 1260
C*              ITSELF. * 1270
C*          +1 TAKE A NEW STEP CONTINUING FROM THE LAST. * 1280
C*          JSTART IS SET TO NQ, THE CURRENT ORDER OF THE METHOD * 1290
C*          AT EXIT. NQ IS ALSO THE ORDER OF THE MAXIMUM * 1300
C*          DERIVATIVE AVAILABLE. * 1310
C* MAXDER    THE MAXIMUM DERIVATIVE THAT SHOULD BE USED IN THE * 1320
C*          METHOD. SINCE THE ORDER IS EQUAL TO THE HIGHEST * 1330
C*          DERIVATIVE USED, THIS RESTRICTS THE ORDER. IT MUST * 1340
C*          BE LESS THAN 13 FOR ADAMS AND 7 FOR STIFF METHODS. * 1350
C* PW        A BLOCK OF AT LEAST N**2 FLOATING POINT LOCATIONS. * 1360
C* IP        A BLOCK N INTEGERS. * 1370
C***** * 1380
C* SUBROUTINE DIFSRM (N,T,Y,SAVE,CSAVE,H,HMIN,HMAX,EPS,MF,YMAX,EKRROR, * 1390
C*   IKFLAG,JSTART,MAXDER,PW,IP) * 1400
C*   DIMENSION Y(13,1), YMAX(1), SAVE(13,1), ERROR(1), PW(1), A(13), PE * 1410
C*   1RTST(12,2,3), CSAVE(N*3), IP(1) * 1420
C***** * 1430
C* THE COEFFICIENTS IN PERTST ARE USED IN SELECTING THE STEP AND * 1440
C* ORDER, THEREFORE ONLY ABOUT ONE PERCENT ACCURACY IS NEEDED. * 1450
C***** * 1460
C*   DATA PERTST/2.,4.5,7.333,10.42,13.7,17.15,1.,1.,1.,1.,1.,2.,12. * 1470
C*   1.24.,37.89,53.33,70.08,87.97,100.9,126.7,147.3,168.8,191.4,3.,6.,9 * 1480
C*   2.167,12.4,15.98,1.,1.,1.,1.,1.,1.,1.,12.,24.,37.89,53.33,70.08,87. * 1490
C*   397.,100.9,126.7,147.3,168.9,191.4,211.,1.,1.,5.,1667.,04133.,00826 * 1500
C*   47.1.,1.,1.,1.,1.,1.,1.,2.,1.,3157.,07407.,0139.,0021818.,00029 * 1510
C*   5.,.000035.,.0000037.,.00000035/ * 1520
C*   DATA A(2)/-1.0/ * 1530
C*   IRET=1 * 1540
C*   KFLAG=1 * 1550
C*   IF (JSTART.LE.0) GO TO 50 * 1560
C***** * 1570
C* BEGIN BY SAVING INFORMATION FOR POSSIBLE RESTARTS AND CHANGING * 1580
C* H BY THE FACTOR R IF THE CALLER HAS CHANGED H. ALL VARIABLES * 1590
C* DEPENDENT ON H MUST ALSO BE CHANGED. * 1600
C* E IS A COMPARISON FOR ERRORS OF THE CURRENT ORDER NQ. EUP IS * 1610
C* TO TEST FOR INCREASING THE ORDER, ED*N FOR DECREASING THE ORDER. * 1620
C* HNEW IS THE STEP SIZE THAT WAS USED ON THE LAST CALL. * 1630
C***** * 1640
C*   10 DO 20 I=1,N * 1650

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DO 20 J=1,K	1660
20 SAVE(J,I)=Y(J,I)	1670
HOLDU=HNEW	1680
IF (M.EQ.HOLD) GO TO 40	1690
30 IRET=1	1700
GO TO 750	1710
40 NQOLD=NQ	1720
TOLDU=T	1730
IF (JSTART.GT.0) GO TO 300	1740
GO TO 80	1750
50 IF (JSTART.EQ.-1) GO TO 70	1760
C*****	1770
C* ON THE FIRST CALL, THE ORDER IS SET TO 1 AND THE INITIAL	1780
C* DERIVATIVES ARE CALCULATED.	1790
C*****	1800
RR=1.0	1810
NQ=1	1820
N3=N	1830
N4=N**2	1840
CALL DIFFUN (N,T,Y,CSAVE(1,3))	1850
DO 60 I=1,N	1860
60 Y(I,I)=CSAVE(I,3)*H	1870
HNEW=H	1880
K=2	1890
GO TO 10	1900
C*****	1910
C* REPEAT LAST STEP BY RESTORING SAVED INFORMATION.	1920
C*****	1930
70 IF (NQ.EQ.NQOLD) JSTART=1	1940
T=TOLD	1950
NQ=NQOLD	1960
K=NQ+1	1970
GO TO 30	1980
C*****	1990
C* SET THE COEFFICIENTS THAT DETERMINE THE ORDER AND THE METHOD	2000
C* TYPE. CHECK FOR EXCESSIVE ORDER. THE LAST TWO STATEMENTS OF	2010
C* THIS SECTION SET IWEVAL .GT.0 IF PW IS TO BE RE-EVALUATED	2020
C* BECAUSE OF THE ORDER CHANGE, AND THEN REPEAT THE INTEGRATION	2030
C* STEP IF IT HAS NOT YET BEEN DONE (IRET = 1) OR SKIP TO A FINAL	2040
C* SCALING BEFORE EXIT IF IT HAS BEEN COMPLETED (IRET = 2).	2050
C*****	2060
80 IF (MF.EQ.0) GO TO 90	2070
IF (NQ.GT.6) GO TO 100	2080
GO TO (230,240,250,260,270,280), NQ	2090
90 IF (NQ.GT.12) GO TO 100	2100
GO TO (110,120,130,140,150,160,170,180,190,200,210,220), NQ	2110
100 KFLAG=-2	2120
RETURN	2130
C*****	2140
C* THE FOLLOWING COEFFICIENTS SHOULD BE DEFINED TO THE MAXIMUM	2150
C* ACCURACY PERMITTED BY THE MACHINE. THEY ARE, IN THE ORDER USED..	2160
C*	2170
C* -1	2180
C* -1/2,-1/2	2190
C* -5/12,-3/4,-1/6	2200

C* -3/2,-11/12,-1/3,-1/24	*	2210
C* -251/720,-25/24,-35/72,-5/48,-1/120	*	2220
C* -95/288,-137/120,-5/8,-17/96,-1/40,-1/720	*	2230
C* -19087/60480,-49/40,-203/470,-49/192,-7/144,-7/1440,-1/5040	*	2240
C* -5257/17280,-363/280,-469/540,-967/2880,-7/90,-23/2160,-1/1260,		2250
C* -1/40320		2260
C* -107001/3628800,-761/560,-29531/30240,-267/640,-1069/9600,-3/160,		2270
C* -13/6720,-1/8960,-1/362880		2280
C* -4165506/14515200,-7129/5040,-6515/6048,-4523/9072,-19/128,		2290
C* -5013/1036800,-5/1344,-29/96768,-1/72576,-1/3628800		2300
C* -134211265/479001600,-7381/5040,-177133/151200,-84095/145152,		2310
C* -341693/1814400,-8591/207360,-7513/1209600,-121/193536,		2320
C* -11/272160,-11/7257600,-1/39916800		2330
C* -262747265/958003200,-83711/55440,-190553/151200,-341747/518400,		2340
C* -139381/604800,-2667907/47900160,-1903/201600,-10831/9676800,		2350
C* -11/120960,-1/207360,-1/6652800,-1/479001600		2360
C*	*	2370
C*	*	2380
C* -1	*	2390
C* -2/3,-1/3	*	2400
C* -12/25,-7/10,-1/5,-1/50	*	2410
C* -120/274,-225/274,-85/274,-15/274,-1/274	*	2420
C* -180/441,-58/63,-15/36,-25/252,-3/252,-1/1764	*	2430
C*****		2440
110 A(1)=-1.0		2450
GO TO 290		2460
120 A(1)=-.500000000000		2470
A(3)=-0.500000000		2480
GO TO 290		2490
130 A(1)=-0.4166666666666667		2500
A(3)=-0.750000000		2510
A(4)=-0.1666666666666667		2520
GO TO 290		2530
140 A(1)=-0.3750000000000		2540
A(3)=-0.9166666666666667		2550
A(4)=-0.3333333333333333		2560
A(5)=-0.04166666666666667		2570
GO TO 290		2580
150 A(1)=-0.3486111111111111		2590
A(3)=-1.0416666666666667		2600
A(4)=-0.4861111111111111		2610
A(5)=-0.1041666666666667		2620
A(6)=-0.008333333333333333		2630
GO TO 290		2640
160 A(1)=-0.3298611111111111		2650
A(3)=-1.1416666666666667		2660
A(4)=-0.625000000		2670
A(5)=-0.1770833333333333		2680
A(6)=-0.025000000		2690
A(7)=-0.001388888888888889		2700
GO TO 290		2710
170 A(1)=-0.3155919312169312		2720
A(3)=-1.225000000		2730
A(4)=-0.7518518518518519		2740
A(5)=-0.2552083333333333		2750

	A(6)=-0.0486111111111111	2760
	A(7)=-0.0048611111111111	2770
	A(8)=-0.0001984126984126984	2780
	GO TO 290	2790
180	A(1)=-0.3042245370370370	2800
	A(3)=-1.296428571428485	2810
	A(4)=-0.8685185185185185	2820
	A(5)=-0.3357638888888888	2830
	A(6)=-0.0777777777777777	2840
	A(7)=-0.01064814814814815	2850
	A(8)=-0.0007936507936507937	2860
	A(9)=-0.0000248015873015873	2870
	GO TO 290	2880
190	A(1)=-0.2948680004409171	2890
	A(3)=-1.35892857142857	2900
	A(4)=-0.976554232804233	2910
	A(5)=-0.41718750000	2920
	A(6)=-0.1113541666666667	2930
	A(7)=-0.0187500000	2940
	A(8)=-0.001934523809523809	2950
	A(9)=-0.000111607142857143	2960
	A(10)=-0.00000275573192239859	2970
	GO TO 290	2980
200	A(1)=-0.2869754464285714	2990
	A(3)=-1.41448412698413	3000
	A(4)=-1.07721560846561	3010
	A(5)=-0.498567019400353	3020
	A(6)=-0.148437500000	3030
	A(7)=-0.0290605709876543	3040
	A(8)=-0.0037202380952381	3050
	A(9)=-0.000299685846560847	3060
	A(10)=-0.0000137786596119929	3070
	A(11)=-0.00000027557319223986	3080
	GO TO 290	3090
210	A(1)=-0.280189564439367	3100
	A(3)=-1.46448412698413	3110
	A(4)=-1.17151455026455	3120
	A(5)=-0.579358190035273	3130
	A(6)=-0.188322861552028	3140
	A(7)=-0.0414303626543210	3150
	A(8)=-0.0062111447989418	3160
	A(9)=-0.000625206679894180	3170
	A(10)=-0.0000404174015285126	3180
	A(11)=-0.00000151565255731922	3190
	A(12)=-0.0000000250521083854417	3200
	GO TO 290	3210
220	A(1)=-0.274265540031599	3220
	A(3)=-1.50993867243867	3230
	A(4)=-1.26027116402116	3240
	A(5)=-0.659234182098765	3250
	A(6)=-0.230458002645503	3260
	A(7)=-0.0556972461052322	3270
	A(8)=-0.00943948412698413	3280
	A(9)=-0.00111927496693122	3290
	A(10)=-0.0000909391534391534	3300

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      A(11)=-0.00000492253086419753      3310
      A(12)=-0.000000150312650312650      3320
      A(13)=-0.00000000200767569878681    3330
      GO TO 290                              3340
230  A(1)=-1.000000000                      3350
      GO TO 290                              3360
240  A(1)=-0.6666666666666666666666667    3370
      A(3)=-0.3333333333333333333333333    3380
      GO TO 290                              3390
250  A(1)=-0.5454545454545454545454545    3400
      A(3)=A(1)                              3410
      A(4)=-0.09090909090909090909091      3420
      GO TO 290                              3430
260  A(1)=-0.4800000000                     3440
      A(3)=-0.7000000000                     3450
      A(4)=-0.2000000000                     3460
      A(5)=-0.0200000000                     3470
      GO TO 290                              3480
270  A(1)=-0.437956204379562              3490
      A(3)=-0.8211673832116788              3500
      A(4)=-0.3102189781021898              3510
      A(5)=-0.05474452554744526             3520
      A(6)=-0.0036496350364963504           3530
      GO TO 290                              3540
280  A(1)=-0.4081632653061225              3550
      A(3)=-0.9206349206349206              3560
      A(4)=-0.4166666666666666666666667    3570
      A(5)=-0.0992063492063492              3580
      A(6)=-0.0119047619047619              3590
      A(7)=-0.000566893424036282            3600
290  K=NU+1                                  3610
      IDOUB=K                                3620
      MTYP=(4-MF)/2                          3630
      ENQ4=.5/FLOAT(NQ+1)                    3640
      ENQ3=.5/FLOAT(NQ+2)                    3650
      ENQ1=.5/FLOAT(NQ)                      3660
      PEP SH=EPS                              3670
      EUP=(PERTST(NQ,MTYP,2)*PEP SH)**2      3680
      E=(PERTST(NQ,MTYP,1)*PEP SH)**2        3690
      EDWN=(PERTST(NQ,MTYP,3)*PEP SH)**2      3700
      IF (EDWN.EQ.0) GO TO 780                3710
      BND=(EPS*ENQ3)**2                      3720
      IWEVAL=MF                              3730
      GO TO (300,670), IRET                   3740
C*****                                     3750
C* THIS SECTION COMPUTES THE PREDICTED VALUES BY EFFECTIVELY * 3760
C* MULTIPLYING THE SAVED INFORMATION BY THE PASCAL TRIANGLE * 3770
C* MATRIX. * 3780
C*****                                     3790
300  T=T+M                                    3800
      DO 310 J=2,K                            3810
          DU 310 J1=J,K                      3820
              JC=K-J1+J-1                    3830
              DU 310 I=1,N                    3840
310  Y(J2,I)=Y(J2,I)+Y(J2+1,I)              3850

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C***** 3860
C* UP TO 2 CORRECTOR ITERATIONS ARE TAKEN. CONVERGENCE IS TESTED BY * 3870
C* REQUIRING THE L2 NORM OF CHANGES TO BE LESS THAN BND WHICH IS * 3880
C* DEPENDENT ON THE ERROR TEST CONSTANT. * 3890
C* THE SUM OF THE CORRECTIONS IS ACCUMULATED IN THE ARRAY * 3900
C* ERROR(I). IT IS EQUAL TO THE K-TH DERIVATIVE OF Y MULTIPLIED * 3910
C* BY H**K/(FACTORIAL(K-1)*A(K)). AND IS THEREFORE PROPORTIONAL * 3920
C* TO THE ACTUAL ERRORS TO THE LOWEST POWER OF H PRESENT. (H**K) * 3930
C***** 3940
DO 340 I=1,N
320 ERROR(I)=0.0
DO 430 L=1,2
CALL DIFFUN (N,T,Y,CSAVE(1,3))
C***** 3990
C* IF THERE HAS BEEN A CHANGE OF ORDER OR THERE HAS BEEN TROUBLE * 4000
C* WITH CONVERGENCE. PW IS RE-EVALUATED PRIOR TO STARTING THE * 4010
C* CORRECTOR ITERATION IN THE CASE OF STIFF METHODS. IWEVAL IS * 4020
C* THEN SET TO -1 AS AN INDICATOR THAT IT HAS BEEN DONE. * 4030
C***** 4040
IF (IWEVAL.LT.1) GO TO 390
IF (MF.EQ.2) GO TO 360
CALL DEDERV (T,Y,PW,N3)
R=A(1)*H
DO 330 I=1,N4
330 PW(I)=PW(I)*R
C***** 4110
C* ADD THE IDENTITY MATRIX TO THE JACOBIAN AND DECOMPOSE INTO LU = PW * 4120
C***** 4130
340 DO 350 I=1,N
350 PW(I*(N3+1)-N3)=1.0+PW(I*(N3+1)-N3)
IWEVAL=-1
CALL NDI01Z (N,N3,PW,IP)
IF (IP(N).NE.0) GO TO 390
GO TO 440
C***** 4200
C* EVALUATE THE JACOBIAN INTO PW BY NUMERICAL DIFFERENCING. R IS THE * 4210
C* CHANGE MADE TO THE ELEMENT OF Y. IT IS EPS RELATIVE TO Y WITH * 4220
C* A MINIMUM OF EPS**2. F STORES THE UNCHANGED VALUE OF Y. * 4230
C***** 4240
360 DO 380 J=1,N
F=Y(1,J)
R=EPS*AMAX1(EPS,ABS(F))
Y(1,J)=Y(1,J)+R
U=A(1)*H/R
CALL DIFFUN (N,T,Y,CSAVE(1,1))
DO 370 I=1,N
370 PW(I+(J-1)*N3)=(CSAVE(I,1)-CSAVE(I,3))*D
380 Y(1,J)=F
GO TO 340
DO 400 I=1,N
390 CSAVE(I,1)=Y(2,I)-CSAVE(I,3)*H
400 IF (MF.EQ.0) GO TO 410
CALL NDI027 (N,N3,PW,CSAVE(1,1),IP)
C***** 4380
C* CORRECT AND COMPARE DEL, THE L2 NORM OF CHANGE/YMAX, WITH BND. * 4390
4400

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C* ESTIMATE THE VALUE OF THE L2 NORM OF THE NEXT CORRECTION BY      * 4410
C* BR*2*DEL AND COMPARE WITH RND. IF EITHER IS LESS, THE CORRECTOR * 4420
C* IS SAID TO HAVE CONVERGED.                                     * 4430
C*****                                                             * 4440
410 DEL=0.0                                                         4450
    DO 420 I=1,N                                                    4460
        Y(1,I)=Y(1,I)+A(1)*CSAVE(I,1)                             4470
        Y(2,I)=Y(2,I)-CSAVE(I,1)                                  4480
        ERROR(I)=ERROR(I)+CSAVE(I,1)                             4490
        DEL=DEL+(CSAVE(I,1)/YMAX(I))*2                            4500
420 CONTINUE                                                         4510
    IF (L.GE.2) BR=4MAX1(.9*BR,DEL/DEL1)                           4520
    DEL1=DEL                                                         4530
    IF (AMIN1(DEL,BR*DEL*2.0).LE.BND) GO TO 480                    4540
430 CONTINUE                                                         4550
C*****                                                             4560
C* THE CORRECTOR ITERATION FAILED TO CONVERGE IN 2 TRIES. VARIOUS * 4570
C* POSSIBILITIES ARE CHECKED FOR. IF H IS ALREADY HMIN AND      * 4580
C* THIS IS EITHER ADAMS METHOD OR THE STIFF METHOD IN WHICH THE * 4590
C* MATRIX PW HAS ALREADY BEEN RE-EVALUATED, A NO CONVERGENCE EXIT * 4600
C* IS TAKEN. OTHERWISE THE MATRIX PW IS RE-EVALUATED AND/OR THE * 4610
C* STEP IS REDUCED TO TRY AND GET CONVERGENCE.                  * 4620
C*****                                                             4630
440 T=TULO                                                           4640
    IF (ABS(H).LE.HMIN*1.00001.AND.(IWEVAL-MTYP).LT.-1) GO TO 450 4650
    IF ((MF.EQ.0).OR.(IWEVAL.NE.0)) H=H*0.25                      4660
    IWEVAL=MF                                                         4670
    IRET1=2                                                           4680
    GO TO 750                                                         4690
450 KFLAG=-3                                                         4700
    NQ=N*0.5                                                         4710
460 DO 470 I=1,N                                                    4720
    DO 470 J=1,K                                                    4730
470 Y(J,I)=SAVE(J,I)                                                4740
    H=HOLD                                                            4750
    JSTMT=NQ                                                         4760
    RETURN                                                            4770
C*****                                                             4780
C* THE CORRECTOR CONVERGED AND CONTROL IS PASSED TO STATEMENT 520 * 4790
C* IF THE ERROR TEST IS O.K., AND TO 540 OTHERWISE.             * 4800
C* IF THE STEP IS O.K. IT IS ACCEPTED. IF IDOUB HAS BEEN REDUCED * 4810
C* TO ONE, A TEST IS MADE TO SEE IF THE STEP CAN BE INCREASED * 4820
C* AT THE CURRENT ORDER OR BY GOING TO ONE HIGHER OR ONE LOWER. * 4830
C* SUCH A CHANGE IS ONLY MADE IF THE STEP CAN BE INCREASED BY AT * 4840
C* LEAST 1.1. IF NO CHANGE IS POSSIBLE IDOUB IS SET TO 8 TO    * 4850
C* PREVENT FUTURE TESTING FOR 8 STEPS.                           * 4860
C* IF A CHANGE IS POSSIBLE, IT IS MADE AND IDOUB IS SET TO     * 4870
C* NQ + 1 TO PREVENT FURTHER TESTING FOR THAT NUMBER OF STEPS. * 4880
C* IF THE ERROR WAS TOO LARGE, THE OPTIMUM STEP SIZE FOR THIS OR * 4890
C* LOWER ORDER IS COMPUTED, AND THE STEP RETRIED. IF IT SHOULD * 4900
C* FAIL TWICE MORE IT IS AN INDICATION THAT THE DERIVATIVES THAT * 4910
C* HAVE ACCUMULATED IN THE Y ARRAY HAVE ERRORS OF THE WRONG ORDER * 4920
C* SO THE FIRST DERIVATIVES ARE RECOMPUTED AND THE ORDER IS SET * 4930
C* TO 1.                                                           * 4940
C*****                                                             4950

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```

480 D=0.0
      DO 490 I=1,N
490   D=D+(ERROR(I)/YMAX(I))**2
      IWEVAL=N
      IF (D.GT.E) GO TO 530
      IF (K.LT.3) GO TO 510
C*****
C* COMPLETE THE CORRECTION OF THE HIGHER ORDER DERIVATIVES AFTER A *
C* SUCCESSFUL STEP. *
C*****
      DO 500 J=3,K
      DO 500 I=1,N
500   Y(J,I)=Y(J,I)+A(J)*ERROR(I)
510   KFLAG=K+1
      HNEW=H
      IF (IDOUR.LE.1) GO TO 540
      IDOUB=IDOUR-1
      IF (IDOUR.GT.1) GO TO 590
      DO 520 I=1,N
520   CSAVE(I,2)=ERROR(I)
      GO TO 590
C*****
C* REDUCE THE FAILURE FLAG COUNT TO CHECK FOR MULTIPLE FAILURES. *
C* RESTORE T TO ITS ORIGINAL VALUE AND TRY AGAIN UNLESS THERE HAVE *
C* THREE FAILURES. IN THAT CASE THE DERIVATIVES ARE ASSUMED TO HAVE *
C* ACCUMULATED ERRORS SO A RESTART FROM THE CURRENT VALUES OF Y IS *
C* TRIED. THIS IS CONTINUED UNTIL SUCCESS OR H = HMIN. *
C*****
530   KFLAG=KFLAG-2
      IF (ABS(H).LF.(HMIN*1.00001)) GO TO 740
      T=TOLD
      IF (KFLAG.LF.-5) GO TO 710
C*****
C* PR1, PR2, AND PR3 WILL CONTAIN THE AMOUNTS BY WHICH THE STEP SIZE *
C* SHOULD BE DIVIDED AT ORDER ONE LOWER, AT THIS ORDER, AND AT ORDER *
C* ONE HIGHER RESPECTIVELY. *
C*****
540   PR2=(D/F)**ENQ2*1.2
      PR3=1.E+20
      IF ((NQ.GE.MAXDER).OR.(KFLAG.LE.-1)) GO TO 560
      D=0.0
      DO 550 I=1,N
550   D=D+((ERROR(I)-CSAVE(I,2))/YMAX(I))**2
      PR3=(D/F)**ENQ3*1.4
560   PR1=1.E+20
      IF (NQ.LE.1) GO TO 580
      D=0.0
      DO 570 I=1,N
570   D=D+(Y(K,I)/YMAX(I))**2
      PR1=(D/F)**ENQ1*1.3
580   CONTINUE
      IF (PR2.LE.PR3) GO TO 540
      IF (PR3.LT.PR1) GO TO 550
590   R=1.0/AMAX1(PR1,1.E-4)
      NEWQ=NQ-1

```

600 IDOUB=8	5510
IF ((KFLAG.EQ.1).AND.(R.LT.(1.1))) GO TO 690	5520
IF (NEWQ.LE.NQ) GO TO 620	5530
C*****	5540
C* COMPUTE ONE ADDITIONAL SCALED DERIVATIVE IF ORDER IS INCREASED. *	5550
C*****	5560
DO 610 I=1,N	5570
610 Y(NEWQ+1,I)=ERROR(I)*A(K)/FLOAT(K)	5580
620 K=NEWQ+1	5590
IF (KFLAG.EQ.1) GO TO 660	5600
H=H*R	5610
IREI=3	5620
GO TO 750	5630
630 IF (NEWQ.EQ.NQ) GO TO 300	5640
NQ=NEWQ	5650
GO TO 80	5660
640 IF (PR2.GT.PR1) GO TO 590	5670
NEWQ=NQ	5680
R=1.0/AMAX1(PR2,1.E-4)	5690
GO TO 500	5700
650 R=1.0/AMAX1(PR3,1.E-4)	5710
NEWQ=NQ+1	5720
GO TO 600	5730
660 IREI=2	5740
R=AMIN1(R,HMAX/ABS(H))	5750
H=H*R	5760
HNEW=H	5770
IF (NQ.EQ.NFWQ) GO TO 670	5780
NQ=NEWQ	5790
GO TO 80	5800
670 R1=1.0	5810
DO 680 J=2,K	5820
R1=R1*R	5830
DO 680 I=1,N	5840
680 Y(J,I)=Y(J,I)*R1	5850
IDOUB=K	5860
690 DO 700 I=1,N	5870
700 YMAX(I)=AMAX1(YMAX(I),ABS(Y(I,I)))	5880
JSTART=NQ	5890
RETURN	5900
710 IF (NQ.GT.1) GO TO 720	5910
IF (ABS(H).LE.2.*HMIN) GO TO 780	5920
GO TO 540	5930
720 R=H*HOLD	5940
DO 730 I=1,N	5950
Y(1,I)=SAVE(1,I)	5960
730 Y(2,I)=SAVE(2,I)*R	5970
NQ=1	5980
KFLAG=1	5990
GO TO 80	6000
740 KFLAG=-1	6010
HNEW=H	6020
JSTART=NQ	6030
RETURN	6040
C*****	6050

```

C* THIS SECTION SCALES ALL VARIABLES CONNECTED WITH H AND RETURNS * 6060
C* TO THE ENTERING SECTION. * 6070
C***** 6080
750 H=SIGN(AMAX1(HMIN,AMIN1(ABS(H),HMAX)),H) 6090
R1=1.0 6100
DO 760 J=2,K 6110
R=M/HOLD 6120
R1=R1*R 6130
DO 760 I=1,N 6140
760 Y(J,I)=SAVF(J,I)*R1 6150
DO 770 I=1,N 6160
770 Y(I,I)=SAVF(I,I) 6170
IDOUTB=K 6180
GO TO (40,300,630), IRET1 6190
780 KFLAG=-4 6200
GO TO 460 6210
END 6220
SUBROUTINE NDI01Z (N,NDIM,A,IP) 6230
C--MOLER'S "DFCOMP". 6240
C 6250
C MATRIX TRIANGULARIZATION BY GAUSSIAN ELIMINATION. 6260
C 6270
C INPUT... 6280
C N = ORDER OF MATRIX 6290
C NDIM = DECLARED DIMENSION OF ARRAY A. 6300
C A = MATRIX TO BE TRIANGULARIZED. (FOR STIFF METHODS, A IS SINGLE 6310
PRECISION; ALL OTHER VARIABLES ARE DOUBLE PRECISION.) 6320
C OUTPUT... 6330
C A(I,J), I.LE.J = UPPER TRIANGULAR FACTOR, U. 6340
C A(I,J), I.GT.J = MULTIPLIERS = LOWER TRIANGULAR FACTOR, I-L. 6350
C IP(K), K.LT.N = INDEX OF K-TH PIVOT ROW. 6360
C IP(N) = (-1)**(NUMBER OF INTERCHANGES) OR 0. 6370
C USE 'SOLVE' (NDI02Z) TO OBTAIN SOLUTION OF LINEAR SYSTEM. 6380
C DETERM(A) = IP(N)*A(1,1)*A(2,2)*...*A(N,N). 6390
C IF IP(N) = 0, A IS SINGULAR, SOLVE WILL DIVIDE BY ZERO. 6400
C 6410
C DIMENSION A(NDIM,NDIM), IP(NDIM) 6420
IP(N)=1 6430
DO 20 K=1,N 6440
IF (K.EQ.N) GO TO 50 6450
KPI=K+1 6460
M=K 6470
DO 10 I=KPI,N 6480
IF (ABS(A(I,K)).GT.ABS(A(M,K))) M=I 6490
10 CONTINUE 6500
IP(K)=M 6510
IF (M.NE.K) IP(N)=-IP(N) 6520
T=A(M,K) 6530
A(M,K)=A(K,K) 6540
A(K,K)=T 6550
IF (T.EQ.0) GO TO 50 6560
DO 20 I=KPI,N 6570
20 A(I,K)=-A(I,K)/T 6580
DO 40 J=KPI,N 6590
I=A(M,J) 6600

```

A(M,J)=A(K,J)	6610
A(K,J)=T	6620
IF (T.EQ.0.) GO TO 40	6630
DO 30 I=KP1,N	6640
30 A(I,J)=A(I,J)+A(I,K)*T	6650
40 CONTINUE	6660
50 IF (A(K,K).EQ.0.) IP(N)=0	6670
60 CONTINUE	6680
RETURN	6690
END	6700
SUBROUTINE NDI02Z (N,NDIM,A,B,IP)	6710
C--MGLER'S "SOLVE".	6720
C	6730
C SOLUTION OF LINEAR SYSTEM, $A \cdot X = B$.	6740
C	6750
C INPUT...	6760
C N = ORDER OF MATRIX.	6770
C NDIM = DECLARED DIMENSION OF ARRAY A.	6780
C A = TRIANGULARIZED MATRIX OBTAINED FROM 'DECOMP' (NDI01Z).	6790
C B = RIGHT HAND SIDE VECTOR.	6800
C IP = PIVOT VECTOR OBTAINED FROM 'DECOMP'.	6810
C OUTPUT...	6820
C B = SOLUTION VECTOR, X.	6830
DIMENSION A(NDIM,NDIM), B(NDIM), IP(NDIM)	6840
IF (N.EQ.1) GO TO 30	6850
NM1=N-1	6860
DO 10 K=1,NM1	6870
KP1=K+1	6880
M=IP(K)	6890
T=B(M)	6900
B(M)=B(K)	6910
B(K)=T	6920
DO 10 I=KP1,N	6930
10 B(I)=B(I)+A(I,K)*T	6940
DO 20 KR=1,NM1	6950
KM1=N-KR	6960
K=KM1+1	6970
B(K)=B(K)/A(K,K)	6980
T=-B(K)	6990
DO 20 I=1,KM1	7000
20 B(I)=B(I)+A(I,K)*T	7010
30 B(1)=B(1)/A(1,1)	7020
RETURN	7030
END	7040

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16 Abstract A computer program has been developed to predict the trajectory, ground deposition, and drift of liquid sprays injected into the wake of an aircraft in ground effect. The program uses a horseshoe vortex wake model and includes the effects of liquid droplet evaporation, crosswind, the propeller slipstream, ground effect, and tunnel walls on small scale models. This paper is intended as a user's guide and includes several sample cases demonstrating the various user options. A complete listing of the FORTRAN computer program is provided.					
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